

371

yes

MSC INTERNAL TECHNICAL NOTE
MPF - A GENERALIZED PROGRAM
FOR SOLVING NONLINEAR
TWO-POINT BOUNDARY VALUE PROBLEMS

By

J. M. Lewallen
and
S. D. Williams

*National Aeronautics
and Space Administration*

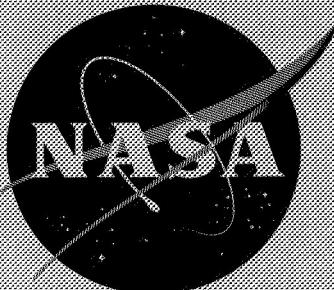
HOUSTON, TEXAS

FACILITY FORM 602

(ACCESSION NUMBER)
~~1870-76178~~
(PAGES)
~~186~~
TMX 65626
(NASA CR OR TMX OR AD NUMBER)

(TRAU)
~~none~~
(CODE)
(CATEGORY)

Manned Spacecraft Center



MSC-ED-R-68-68

MSC INTERNAL TECHNICAL NOTE
MPF - A GENERALIZED PROGRAM,
FOR SOLVING NONLINEAR
TWO-POINT BOUNDARY VALUE PROBLEMS

By

J. M. Lewallen
and
S. D. Williams

National Aeronautics and Space Administration
Manned Spacecraft Center
Houston, Texas
June 1968

Revision date: November 1968 80006

MSC-ED-R-68-68

MPF - A GENERALIZED PROGRAM
FOR SOLVING NONLINEAR
TWO-POINT BOUNDARY VALUE PROBLEMS

Prepared by: J M. Lewallen
J. M. Lewallen
Aerospace Technologist

Prepared by: S. D. Williams
S. D. Williams
Scientific Programming Specialist

Approved by: H. P. Decell, Jr.
H. P. Decell, Jr.
Chief, Theory and Analysis Office

Approved by: Eugene H. Brock
Eugene H. Brock
Chief, Computational and Analysis
Division

National Aeronautics and Space Administration
Manned Spacecraft Center
Houston, Texas

MPF - A GENERALIZED PROGRAM
FOR SOLVING NONLINEAR
TWO-POINT BOUNDARY VALUE PROBLEMS

Approved by:

J. A. Barnes

J. A. Barnes
Supervisor, Theory and Analysis Section
Lockheed Electronics Company

Approved by:

C R Wallace

C. R. Wallace
Manager, Scientific Analysis and
Programming Department
Lockheed Electronics Company

National Aeronautics and Space Administration
Manned Spacecraft Center
Houston, Texas

TABLE OF CONTENTS

	<u>Page</u>
SUMMARY	1
INTRODUCTION	2
THEORETICAL DEVELOPMENT	3
DESCRIPTION OF THE PROGRAM	10
MODIFICATIONS FOR DIFFERENT PROGRAM	26
DATA INPUT	32
EQUIPMENT	36
REFERENCES	37
APPENDICES	
A. EARTH-MARS TRANSFER EXAMPLE	A-1
1. FORMULATION	A-1
2. PROGRAM LISTING	A-5
3. FLOW CHART	A-68
4. SAMPLE INPUT	A-107
5. SAMPLE OUTPUT	B-109
B. BRACHISTOCHRONE EXAMPLE	B-1
1. FORMULATION	B-1
2. MODIFICATIONS FOR CUR SYSTEM LISTING	B-6
3. SAMPLE INPUT	B-8
4. SAMPLE OUTPUT	B-9

MPF - A GENERALIZED PROGRAM FOR SOLVING NONLINEAR
TWO-POINT BOUNDARY VALUE PROBLEMS

By

J. M. Lewallen¹
S. D. Williams²

SUMMARY

This report documents a numerical perturbation method that has had impressive success in solving the nonlinear two-point boundary value problem.

Many practical engineering problems are reduced mathematically to the nonlinear two-point boundary value problem. These problems arise quite naturally in areas of vibration, heat transfer, wave propagation, fluid mechanics, trajectory analysis and optimization theory. The method developed here requires the problem to be presented in terms of first-order, ordinary, differential equations with an adequate number of specified boundary conditions. The terminal conditions may be general functions of the problem variables. Several different convergence procedures are available for the user. With proper application, the chances are excellent that convergence may be achieved with only one computer run. If convergence has not been achieved, the information for a restart is provided automatically.

This routine is made available by the Theory and Analysis Office of the Computation and Analysis Division. The routine is generated in an effort to provide MSC engineers and contractor personnel with a proven method

¹ Manned Spacecraft Center, Houston, Texas
² Lockheed Electronics Company, Houston, Texas

for solving a popular class of mathematical problems. The purpose of this program is to reduce both the programming and computer time for the user, while maintaining the convenience of a general program. Office personnel will be available for consultation involving implementation of this program to the users problem. The authors would appreciate comments and suggestions that would improve the program.

INTRODUCTION

In analyzing scientific and engineering systems, it is often necessary to solve a nonlinear two-point boundary value problem. These problems arise naturally in the areas of vibration, heat transfer, wave propagation, fluid mechanics, trajectory analysis, and optimization theory. The equations that describe these dynamic systems are a set of first-order, nonlinear, ordinary differential equations (or may be put into this form). If n differential equations are involved, $n + 2$ boundary conditions must be specified. Assume that n initial values of the n dependent variables, the initial time, and the terminal time are given. In this case, straight forward numerical integration methods will yield a solution. If p ($p < n$) initial dependent variables are specified at the initial time and q ($p + q = n$) terminal dependent variables are specified at the terminal time, there is not a simple approach to the solution. Recourse is usually made to iterative methods.

Attempts to solve this split end-condition problem have been made for years, but until the advent of the high-speed digital computer most methods were considered impractical. In recent years, several efficient methods

have been implemented successfully. In 1963, Breakwell, Speyer and Bryson⁽¹⁾ proposed a method which considers trajectories perturbed from some reference path. These perturbed trajectories are generated when certain initial conditions are varied. After assuming the unknown initial conditions, each iteration produces corrections for the unknown initial conditions such that the terminal constraint error is reduced. Similar methods, considered here as perturbation methods, have been proposed by Jurovics and McIntyre⁽²⁾ and Jazwinski.⁽³⁾ In 1964, McGill and Kenneth⁽⁴⁾ proposed the generalized Newton-Raphson method. Although this method has some distinct advantages over the perturbation methods, a detailed consideration of the method will not be made here. Similar methods have been proposed by Merriam⁽⁵⁾, Sylvester and Meyer⁽⁶⁾ and Roberts and Shipman⁽⁷⁾. An extensive comparison of some of the perturbation and quasilinearization methods has been made by Tapley and Lewallen⁽⁸⁾ and several innovations to accelerate convergence rates have been proposed by Lewallen, Tapley and Williams⁽⁹⁾.

THEORETICAL DEVELOPMENT

The problem is formulated in terms of a nonlinear, ordinary, first-order vector differential equation

$$\dot{z} = F(z, t) \quad (1)$$

where z is an n -vector and t is the independent variable time. If the initial time is specified, $n + 1$ boundary conditions are required for solution.

Suppose that p ($p < n$) of the initial dependent variables are specified at the known initial time and that $q + 1$ ($p + q = n$) relations

$$h(z_f, t_f) = 0 \quad (2)$$

are specified at the unknown terminal time. Since p of the initial variables are specified, it becomes convenient to arrange and partition the n -vector of initial conditions as

$$z^T(t_0) = \begin{bmatrix} x^T(t_0) \\ \lambda^T(t_0) \end{bmatrix} \quad (3)$$

where the vector $x(t_0)$ is composed of the p known initial conditions and the vector $\lambda(t_0)$ is composed of the q unknown initial conditions.

If Equation (1) is expanded in a Taylor's series about some reference trajectory, say the i^{th} approximation to the correct solution, then

$$\dot{z}_{i+1} = \dot{z}_i + \left[\frac{\partial F}{\partial z} \right]_i \left[z_{i+1} - z_i \right] + \dots \quad (4)$$

If the notation $\delta z = z_{i+1} - z_i$ is used, and only linear terms in the expression are retained, the linear perturbation equation is seen to be

$$\dot{\delta z} = \left[\frac{\partial F}{\partial z} \right] \delta z = A \delta z \quad (5)$$

where the matrix $A = \left[\frac{\partial F}{\partial z} \right]$ is evaluated on the i^{th} trajectory.

In a similar manner, Equation (2) may be expanded about the terminal constraints associated with the i^{th} trajectory. If terms higher than the first in dz_f and dt_f are neglected, the result may be expressed as

$$dh = \left[\frac{\partial h}{\partial z} \right]_f dz_f + \left[\frac{\partial h}{\partial t} \right]_f dt_f \quad (6)$$

Considering the effect of a variation in the terminal time leads to

$$dz_f = \delta z_f + \dot{z}_f dt_f \quad (7)$$

If this condition is substituted into Equation (6), the result becomes

$$dh = \left[\frac{\partial h}{\partial z} \right]_f \delta z_f + \dot{h} dt_f \quad (8)$$

where $\left[\frac{\partial h}{\partial z} \right]$ and \dot{h} are evaluated at the terminal point for the i^{th} approximation. If the terminal state variations can be expressed in terms of the initial state variations (i.e., $\delta z_f = \Pi \delta z_o$), Equation (8) becomes

$$dh = \left[\frac{\partial h}{\partial z} \right]_f \Pi \delta z_o + \dot{h} dt_f \quad (9)$$

where Π is an $n \times n$ matrix to be developed.

The method of perturbation functions (MPF) uses the perturbation equation, Equation (5), to generate a matrix of perturbation solutions. This is accomplished by integrating Equation (5) forward n times with the starting conditions

$$\tilde{\Phi}(t_0, t_0) = \left[\delta z_1(t_0), \dots, \delta z_n(t_0) \right] = I \quad (10)$$

where I is the $n \times n$ unity matrix. The solution may be represented by

$$\delta z(t) = \tilde{\Phi}(t, t_0) \delta z(t_0) \quad (11)$$

where it is easily seen that $\Pi = \tilde{\Phi}$ and that Equation (9) becomes

$$dh = \left[\frac{\partial h}{\partial z} \right]_f \tilde{\Phi}(t_f, t_0) \delta z_0 + \dot{h} dt_f \quad (12)$$

The problem may be simplified and the computational efficiency increased by partitioning the vector δz_0 as shown in Equation (3). This requires partitioning the matrix $\tilde{\Phi}$ such that

$$dh = \left[\frac{\partial h}{\partial z} \right]_f \begin{bmatrix} \Psi & | & \Phi \end{bmatrix} \begin{bmatrix} \delta x_0 \\ \bar{\delta \lambda}_0 \end{bmatrix} + \dot{h} dt_f \quad (13)$$

where Ψ is an $n \times p$ matrix and Φ is an $n \times q$ matrix. Since the p initial values of x_0 are specified, $\delta x_0 = 0$, then Equation (13) reduces to

$$dh = \left[\frac{\partial h}{\partial z} \right]_f \Phi(t_f, t_0) \delta \lambda_0 + \dot{h} dt_f \quad (14)$$

where

$$\Phi(t_0, t_0) = \begin{bmatrix} 0 \\ \underline{\underline{P}_{q \times q}} \\ \underline{\underline{I}_{q \times q}} \end{bmatrix} \quad (15)$$

The linear algebraic equation, Equation (14), represents $q + 1$ scalar equations in the $q + 1$ unknowns $\delta\lambda_0$ and dt_f ; i.e.,

$$y = C^{-1}dh \quad (16)$$

where $y^T = [\delta\lambda_0^T; dt_f]$ and $C = \left\{ \left[\frac{\partial h}{\partial z} \right] \Phi(h) \right\}$. The computational efficiency is increased because the $\Phi(t_f, t_0)$ matrix requires only q vectors to be integrated rather than the n vectors required to generate $\Phi(t_f, t_0)$. The correction equation, Equation (16), requires the specification of a desired change in terminal error, dh . The philosophy associated with this correction is described in detail in Reference 9 and will be discussed briefly here.

The two major assumptions that have been made in this development are the linearization of both the differential equations and the terminal constraints. This linearization will not present severe limitations if successive trajectory iterations are near one another; i.e., the linearization assumptions are not stretched beyond the limits of validity. Hence, the Normal Correction Procedure of requesting that the change in terminal error, dh , for the next iteration equal the negative of the error, $-h$, on the last iteration may be too severe. In this case, divergence could occur.

To avoid this possibility, a Fractional Correction Procedure may be employed such that

$$dh = -ch \quad (17)$$

where $0 < c \leq 1.0$. Reference 9 also discusses the Minimum Norm Correction Procedure which allows each element of the h vector to have a different weight. Briefly, this procedure uses corrections obtained from either

$$y = -[C^T C + \beta I]^{-1} C^T h \quad (18)$$

or

$$y = -[C^T C + \alpha \text{diag}(C^T C)]^{-1} C^T h \quad (19)$$

depending on which procedure is to be used. The β and α are parameters which describe the relation between gradient dominated corrections and Newton-Raphson dominated corrections. The Stepped- β and $-\alpha$ procedures use Equation (18) or (19) until the terminal norm is less than some prespecified value, γ . When this occurs, the Normal Correction Procedure is used to achieve convergence. When the Variable $-\beta$ and $-\alpha$ procedures are being used, the parameters β and α are redefined on each iteration by

$$\beta = \beta_0 \left[\frac{\|h\|^i}{\|h\|_{\max}} \right]^p \quad (20)$$

and

$$\alpha = \alpha_0 \left[\frac{\|h\|^i}{\|h\|_{\max}} \right]^{\rho} \quad (21)$$

where β_0 , α_0 and ρ are chosen by the user.

The computation procedure for MPF is as follows:

- (1) Arrange the n differential equations, Equation (1), such that the p equations having the specified initial conditions occur first.
- (2) Integrate Equation (1) forward with the p known initial conditions and q assumed initial conditions. A terminal time, t_f , must be assumed to terminate the integration.
- (3) Simultaneously with (2) above, Equation (5) must be integrated q times with the initial conditions shown in Equation (15). This results in the matrix $\Phi(t_f, t_0)$. The matrix $[\partial F / \partial z]$ is formed from the trajectory generated in (2).
- (4) When the terminal time is reached, $[\partial h / \partial z]$ and h must be evaluated, and dh must be selected.
- (5) The $q + 1$ algebraic equations, Equation (14), are solved for the $q + 1$ corrections $\delta \lambda_0$ and dt_f . One of the correction procedures must be chosen at this time.

- (6) These corrections are applied to the assumed values of λ_o and t_f and the iterative procedure is repeated.
- (7) Iterations are continued until either the corrections become sufficiently small or the terminal constraints are adequately satisfied.

DESCRIPTION OF THE PROGRAM

In this section, reference is made frequently to the variables introduced in the theoretical development section. To avoid ambiguity, program variables will be capitalized as will be subroutine names. For further clarity, parenthesis will be used to distinguish the program variable associated with the variable introduced in the theory.

MAIN

MAIN is the main program. It initializes data storage, and acts as a driver for various subprograms.

The program first transfers control to the subroutine (F5) that reads in the plotting labels.

The next step is to initialize the clock routine (RESET). The program then transfers control to the general input routine (F6) which reads all the information necessary to run a case. The initial data are saved to

restart from the same trial point with a different correction, if necessary (see CONVRG).

The card punch routine (CRDPCH) is called and the initial conditions of the n dependent variables, the beginning time, the final time, and the maximum norm of the terminal constraints are punched on cards compatible for a continuing run, if desired.

The clock routine (TIME) is called and the computational time is calculated in seconds. The initialization routine (F7) is called to establish the initial conditions for the differential equations and the perturbation equations. The print routine (FPRNT) is called twice; once for the printing of the clock time, and once for the printing of the initial conditions, if desired. The plot storage routine (FPLEW) is called to plot the initial conditions.

The integration routine (INTEG) is called and the n first-order differential equations are integrated forward. Simultaneously, the perturbation equations are integrated forward with the proper starting conditions. The print routine (FPRNT) and the plot storage routine (FPLEW) are called and the terminal conditions are printed and plotted, if desired. The clock routine (TIME) is called and the elapsed time is calculated in seconds. The print routine is called to print the computational time required for the integration, if desired.

The convergence routine (CONVRG) and the plot routine (FPLOT) are called, and the desired graphs for the current iteration are plotted. If convergence has not been achieved, the next iteration is started. If convergence has been

achieved, the clock routine (TIME) is called and the total execution time is calculated in seconds. The print routine (FPRNT) is called to print the total time required for this case, if desired. The plot routine (FPLOT) is called and the desired graphs are generated. The routine will now attempt to execute another case.

Subroutine ABAM

This is a single step predictor-corrector technique used to solve a set of first-order differential equations. The method is based on an Adams-Basford predictor (fourth-order) and an Adams-Moulton corrector (fifth-order) as discussed by Hildebrand.⁽¹⁰⁾ Two iterations (METHOD = 2) on the corrector will usually provide sufficient accuracy for most problems. This routine requires four back points and their derivatives.

The routine integrates, on any given iteration, the differential equations in F1 once, and the differential equations in F2 q times. Storage is facilitated by use of an index register (I1B and I2B) which allows integration to proceed from point to point by changing the index registers rather than moving the data. This gives a rolling drum effect.

The mathematical relations used are

$$Y_{n+1} = y_n + \frac{h}{24} (55y'_n - 59y'_{n-1} + 37y'_{n-2} - 9y'_{n-3}) + \frac{251}{720} h^5 y^V(\epsilon_1)$$

for the predictor, and

$$Y_{n+1} = y_n + \frac{h}{720} (251Y'_{n+1} + 646y'_n - 264y'_{n-1} + 106y'_{n-2} - 19y'_{n-3}) - \frac{3}{160} h^6 y^{vi}(\varepsilon_2)$$

for the corrector, where h is the integration step size.

Subroutine CONVRG

This routine determines the norm of the terminal constraints and implements the different correction schemes according to the value of KEY.

If KEY = 1, the Normal Correction Procedure is used. The change in the terminal error dh (B) is taken as the negative of the error in the terminal constraints h (HNEW), the coefficient matrix C (AA) is evaluated and the correction y (HNEW) is determined. The correction is added to the values on the last iteration (see ITER), and if the norm of the terminal constraints $\|h\|$ (HNN) and the individual corrections are less than some prescribed epsilon (EPS) convergence has been achieved. If convergence has not been achieved, the iteration count (ICOUNT) is checked against the maximum iterations (KK) allowed to determine if the iterative process is to be continued or terminated.

If KEY = 2, the Fractional Correction Procedure is used. The procedure is similar to the case when KEY = 1 for determining a correction. However, the last two convergent solutions are preserved, and the fractional correction constant c (CI) is incremented by a small amount (DEL). If

divergence occurs, then the routine attempts to correct from the last convergent solution decrementing CI by DEL. When the correction vector is decremented to DEL and an improvement has not been made in the terminal error, the routine backs up to the second last convergent solution and tries again. If this does not achieve convergence, the case is aborted.

If KEY = 3, the Minimum Norm Correction Procedure is used. The routine first determines if either of the Stepped procedures are used ($IFL \leq 2$). In this case, the norm of the terminal constraints HNN is checked to determine if it is less than some prespecified value (FL2). If the norm is less than this value, control is switched to the Normal Correction Procedure. If the norm of the terminal constraints HNN is greater than FL2 or if the Variable procedures ($IFL > 2$) are used, the matrix C (AA) is formed (see F4). The matrix AA is printed in FPRNT, if desired, and modified in MLTPY depending on which procedure is used. The modified matrix is then printed in FPRNT, if desired. The corrections are determined in CORVEC and printed, if desired. The routine then determines if convergence has been achieved as in KEY = 1.

Subroutine CORVEC

This subroutine forms the multiplication of a matrix, A , times a vector, B , and stores the answer in a vector, C . Mathematically, $AB = C$.

Subroutine CRDPCH

The purpose of this routine is to punch on cards the values of the dependent variable, initial and terminal time, and the maximum norm of the terminal constraints

for each iteration in the same format required for input.

If IPCH = 0, control is returned to the MAIN program. If IPCH \neq 0, the N initial values of the dependent variables (DEPO) are punched under a 2D25.16 format. The initial time (TO) and the final time (TF) are punched with a 2D25.16 format. Finally, the maximum norm of the terminal constraints $\|h\|^{max}$ (HMAX) is punched under a D25.16 format.

Subroutine FPLEW

This routine stores the various parameters which are to be plotted on the high-speed drum. It uses the symbolic magnetic tape unit 3 to achieve this storage. The parameters which are stored, for the example shown in Appendix A, are the control angle versus time in days (STORA) and the norm of the terminal constraints versus the iteration count (STORB). For the convenience of the user, the control angle is calculated as a Y value, and time is calculated as an X value.

The program determines first, whether or not plotting is desired. If not ($PLT \leq 0$), a return is executed. However, if plotting is desired ($PLT > 0$), the routine has four options dependent upon the argument of I (I = 5,7,9,11).

For I = 5, the plotting point index is initialized for the first point ($K1 = 1$). If this is the first iteration (ICOUNT = 1), the index is set to store the data in the first pair of columns ($K2 = 1$). It is then determined if this is the PLT^{th} (L) iteration. The initial number of points for this curve is set ($INDEX(K2) = 1$), and the data tape is

rewound. The extreme limits for the abscissa (XMAX, XMIN) and the ordinate (YMAX, YMIN) are set. The initial time in days (X(1)) and the control angle in degrees (Y(1)) are calculated. Control is transferred to the data manipulation portion of the routine.

For I = 7, the point and curve indexes are incremented by one ($K_1 = K_1 + 1$ and INDEX(K_2) = INDEX(K_2) + 1), the final time is computed in days (X(1)), and the final control angle is computed in degrees (Y(1)). Control is transferred to the data manipulation portion of the routine.

For I = 9, it is determined if this is the $IPLT^{th}$ point ($J = 0$). For $J \neq 0$, return to the integration routine (INTEG). If this point is to be plotted, the point index is incremented by one ($K_1 = K_1 + 1$), and the time in days and the control angle in degrees are calculated. Control is transferred to the data manipulation portion of the program.

For I = 11, the iteration count (STORB(ICOUNT,1)) and the norm of the terminal constraints (STORB(ICOUNT,2)) are saved, and control is returned to the convergence routine (CONVRG).

In the data manipulation portion of the routine, the actual extrema of the data are determined (XMAX, XMIN, YMAX, YMIN) and the trajectory points (X, Y) are stored into a data array (STORA) which will be placed on the high-speed drum (or magnetic tape unit 3). It is now determined if the data array (STORA) is full ($K_1 = NS$) or if the last point has been calculated ($K = 3$). If neither of these conditions are met, control is returned to the calling routine (MAIN or INTEG).

If the data array is filled or this is the last point, the drum write counter (IWRT(K2)) is incremented, the maximum drum write count is evaluated (IWRT(1)), and the data is placed on the drum.

If this is the last point, the iteration count (IT2) is saved. If this is the first iteration (ICOUNT = 1), or if this is the PLTth ($L = 0$) iteration, the curve index (K2) is incremented by one. A maximum of seven curves is saved (NS2 = 7).

If this is not the last point and this is not the first iteration, the data array (STORA) is filled with the previous information calculated for that block on previous trajectories, and the drum is backed up to allow new information to update the drum. The last point calculated is stored in the first row of the array. The plotting point index is reinitialized (K1 = 1), and the number of points for this curve is incremented by one (INDEX(K2)). Control is returned to the calling routine.

Subroutine FPLOT

This routine plots the information in STORA and STORB. The routine determines first if plots are desired (PLT > 0). If plots are not desired, control is returned to the main program.

If plots are desired, the number of curves (K2) that may occur on one grid is calculated from the iteration count (IT2), the curve plot frequency (PLT), and the maximum numbers of curves (NS2). If JPLT is zero, only the curves for

the current iteration are plotted; otherwise, all K2 curves are generated.

The plotting boundaries are determined, and the associated scaling factors for each curve are calculated. The curve data are read from the drum, and are plotted via the QUIKML routine. If the data were scaled, the scale factors are printed on the grid. If JPLT is zero, control is returned to the main program; otherwise, a plot illustrating the convergence history of this case is desired.

The routine now prepares for the semi-logarithmic plot of the norm of the terminal constraints versus the iteration count (STORB). The plot registers are cleared and the film is advanced one frame (RESET). Next the nonlinear mode is established for 8-cycle log in Y and linear in X (MODE). The labels for the Y axis are established, the span for both the X and Y axis is determined (SET1, SET2, XKK), and the grid increments (FIT) are calculated in rasters for the X axis. The grid routine (GRIDGN) is called and the curve is plotted (PLOT1). The X and Y axis identifiers are printed (D and C). The labels associated with the X axis (LABELX) and the Y axis (PRINT) are plotted.

Finally, the film is advanced (FILMAV), the buffers are dumped (DUMPBUF), and the linear mode is reestablished (MODE).

Subroutine FPRNT

This routine prints the intermediate information required during each iteration. Two printing modes are permitted, the suppressed mode (SWCH = 0) or the full mode (SWCH = 1).

Dependent upon the value (I) received from various routines, different information is printed. The information to be printed is established by the value of JSWCH (JSWCH = I + SWCH) through the use of a computed GO TO statement.

For I = 1 and SWCH = 1, the time in seconds is printed prior to the main program calling the integration routine. Control is returned to the main program. If SWCH = 0, there is no print action.

For I = 3 and SWCH = 1, the time in seconds is printed after calling the integration routine. Control is returned to the main program. If SWCH = 0 there is no print action.

For I = 5 and SWCH = 1, the initial time (TO) and the matrix representing the initial conditions for the differential equations and the perturbation equations (DEP) are printed. Control is returned to the main program. If SWCH = 0, there is no print action.

For I = 7 and SWCH = 1, the final time (VIND) and the matrix representing the final conditions for the differential equations and the perturbation equations (DEP) are printed. If SWCH = 0, the last q differential equations and final time are printed with the iteration number (ICOUNT). Control is returned to the main program.

For I = 9, the IPROth step causes the current time (VIND) and the matrix representing the differential equations and the perturbation equations (DEP) to be printed. Control is returned to the integration routine (INTRK5). The same action is caused by SWCH being either 0 or 1.

For I = 11 and SWCH = 1, the terminal constraint vector h (H) and the norm of the terminal constraints (HNN) are printed. If SWCH = 0, only the norm of the terminal constraints is printed. Control is returned to the convergence routine (CONVRG).

For I = 13 and SWCH = 1, the fractional correction constant (C1) is printed. If SWCH = 0, there is no print action. Control is returned to the convergence routine (CONVRG).

For I = 15 and SWCH = 1, the corrections y (H) that have been computed will be printed. If SWCH = 0, there is no print action. Control is returned to the convergence routine (CONVRG).

For I = 17 and SWCH = 1, the fractional correction constant (C1), the attempt number (ISET), the iteration number (ICOUNT), and the corrections (H) will be printed. If SWCH = 0, there is no print action. Control is returned to the convergence routine (CONVRG).

For I = 19 and SWCH = 1, the matrix A (A) is printed. If SWCH = 0, there is no print action. Control is returned to the convergence routine (CONVRG).

For I = 21 and SWCH = 1, the matrix computed by the Minimum Norm Correction Procedure (see MLTPLY) is printed. If SWCH = 0, there is no print action. Control is returned to the convergence routine (CONVRG).

For I = 23, the same action occurs as in I = 19.

For I = 25 and SWCH = 1, the total time in seconds required to complete the case is printed. If SWCH = 0, there is no print action. Control is returned to the main program.

Subroutine F1

This routine contains the n first-order differential equations that describe the dynamical system. The equations as presented in Equation (1) are

$$\dot{z} = F(z, t)$$

where \dot{z} is represented in the program as P, z is represented as TY, and t is represented as T. The equations are arranged so that the equations having the p known initial conditions are first.

Subroutine F2

This routine contains the n perturbation equations. The equations as presented in Equation (5) are

$$\delta \dot{z} = A \delta z$$

where matrix A represents $[\partial F / \partial z]$ and is evaluated at each integration step.

In the first step, if the index JAY = 2, the procedure is initiated to evaluate the A matrix. Next, an index (NU) is checked to see if the midpoint for Runge-Kutta need be approximated. An index (J) is set depending on the calling routine or the stage of the Runge-Kutta routine. The coefficient matrix is set to zero, and the nonzero elements are

calculated. The product of the $[\partial F/\partial z]$ matrix (A) and δz (TY) is evaluated and stored into the derivative $\dot{\delta z}$ (P). The ordering of the equations in F2 is compatible with that in F1.

Subroutine F3

This routine evaluates the $q + 1$ terminal constraints h (H) from the terminal state $z(t_f)$ (DEP). This is achieved by satisfying a known terminal state which is input z_f (XF), i.e.,

$$h = z(t_f) - z_f$$

or by satisfying a functional relation dependent on the terminal values of the differential equations (DEP) and their derivatives (YPR), i.e.,

$$h = f(\dot{z}, z, t) \Big|_{t_f}$$

Subroutine F4

This routine evaluates the terminal error correction matrix $C = [(\partial h/\partial z)\Phi; \dot{h}]$ (A). The $(\partial h/\partial z)$ matrix (B) is evaluated at final time. The last column of A is set to \dot{h} evaluated at final time. Finally, the matrix multiplication of $(\partial h/\partial z)$ (B) times Φ (DEP) is performed and stored in the first q columns of A.

Subroutine F5

This routine reads in two cards. The first card contains the plotting label for the X axis and the second card contains the plotting label for the Y axis.

Subroutine F6

This is the initial data input routine. Essentially all the data needed to run a case is read in this routine. In addition, most of the data read is printed to verify that the data were input correctly. The only information not read in this routine is that concerned with the plotting labels (See F5).

Subroutine F7

This routine sets up the initialization necessary to perform the integration of the n differential equations and the q n -vectors of initial conditions for the perturbation equations. The routine stores the n initial conditions for the differential equations (DEP0) into working storage (DEP and S), then the initial conditions for the perturbation equations are stored in the right-most portion of the remaining q columns of DEP.

Subroutine INTEG

This routine integrates the n -vector of differential equations and the q n -vectors of perturbation equations. The routine sets the step size for the Runge-Kutta integration and initializes the index registers to give a rolling drum effect. The Runge-Kutta technique yields the first four points and their derivatives to be used by the Adams-Predictor-Corrector technique. The Adams technique integrates forward until the final time (T_2) is exceeded. Then the Runge-Kutta is called to integrate to the final time. Printing and plotting options are examined at each time step to see if this information is desired.

Subroutine INTRK5

This is a standard fourth-order Runge-Kutta integration technique (Reference 10). The storage is handled in the same manner as in ABAM. The mathematical formulation is

$$y_{n+1} = y_n + \frac{1}{6} (k_0 + 2k_1 + 2k_2 + k_3) + O(h^5)$$

where

$$k_0 = h F(x_n, y_n)$$

$$k_1 = h F(x_n + \frac{1}{2} h, y_n + \frac{1}{2} k_0)$$

$$k_2 = h F(x_n + \frac{1}{2} h, y_n + \frac{1}{2} k_1)$$

$$k_3 = h F(x_n + h, y_n + k_2)$$

and h is the step size.

Subroutine ITER

This routine adds the corrections y (H) to the assumed final time (TF) and the q unknown initial λ 's ($DEP0$). It is assumed that these are the last q elements of z ($DEP0$).

Subroutine MLTPLY

This routine sets up the matrix C (A) for the Minimum Norm Correction Procedure described in Reference 9.

The routine first evaluates $C^T C$ ($A^T A$) and stores this into the temporary storage (B). The routine then checks each main diagonal element of B to guarantee that it is

larger than some prescribed value (BETA). If the element is not larger than this number, it is set to the largest main diagonal element. For IFL = 1, the Stepped- α procedure is used, i.e.,

$$B = \left[(A^T A + K) + \alpha_{\text{diag}} (A^T A + K) \right]$$

For IFL = 2, the Stepped- β procedure is used, i.e.,

$$B = \left[(A^T A + K) + \beta I \right]$$

For IFL = 3, the Variable- α procedure is used, i.e.,

$$B = \left[(A^T A + K) + \alpha_0 \left(\frac{HNN}{HMAX} \right)^{\rho} \text{diagonal} (A^T A + K) \right]$$

For IFL = 4, the Variable- β procedure is used, i.e.,

$$B = \left[(A^T A + K) + \beta_0 \left(\frac{HNN}{HMAX} \right)^{\rho} I \right]$$

Finally, the routine transforms the error vector (C) into the modified error vector (D) and replaces the original matrix (A) by the newly computed matrix (B).

Subroutine MINVDP

See Reference 11.

MODIFICATIONS FOR DIFFERENT PROGRAM

This section is designed to assist the user in modifying the existing program to satisfy the users needs and requirements. To some extent, familiarity with the UNIVAC CUR system is useful and a basic knowledge of FORTRAN is assumed. All control cards are assumed to start in column 1 unless specified otherwise. An apostrophe in column 1 indicates a 7-8 multiple punch.

CONTROL CARD 1

\$JOB card

CONTROL CARD 2

'N HDG INPUT 10615
The message INPUT begins in column 13

CONTROL CARD 3

' ASG P=10615

CONTROL CARD 4

' XQT CUR

CONTROL CARDS 5-8 (all start in column 3)

TRW P
IN P
TRI P
TOC

CONTROL CARDS 9-10

'T FOR,* MAIN,MAIN,MAIN/B
-1,1

Insert the FORTRAN PARAMETER card equating N with the number of dependent variables, NQ with the q + 1 terminal constraints, KON with the number of special program constants required, and NS3 with the number of AXIS labels to be read in, i.e.,

PARAMETER N = 8, NQ = 4, KON = 4, NS3 = 2

After this, the parameter card and the desired variables on the card will be indicated by ?. The user supplies the correct value for N, NQ, KON, and NS3.

CONTROL CARDS 11-12

'T FOR,* ABAM,ABAM,ABAM/B
-2,2
PARAMETER N = ?, NQ = ?

CONTROL CARDS 13-14

'T FOR,* CONVRG,CONVRG,CONVRG/B
-2,2
PARAMETER N = ?, NQ = ?

CONTROL CARDS 15-16

'T FOR,* CRDPCH,CRDPCH,CRDPCH/B
-2,2
PARAMETER N = ?

CONTROL CARDS 17-18

'T FOR,* FPLEW,FPLEW,FPLEW/B
-3,5'
PARAMETER N = ?, NQ = ?, NS3 = ?, KON = 4
DOUBLE PRECISION - anything required

CONTROL CARD 19

```
-61,62
X(1) = information for the first X-axis
Y(1) = information for the first Y-axis
-----  
-----  
X(NS4) = information for the last X-axis
Y(NS4) = information for the last Y-axis
(Note: NS4 = NS3/2, and is defined in a
parameter statement.)
```

CONTROL CARDS 20-21

```
'T FOR,* FPLOT,FPLOT,FPLOT/B
-2,2
PARAMETER NS3 = ?
```

CONTROL CARDS 22-23

```
'T FOR,* FPRNT,FPRNT,FPRNT/B
-2,2
PARAMETER N = ?, NQ = ?
```

CONTROL CARDS 24-25

```
'T FOR,* F1,F1,F1/B
-2,2
PARAMETER N = ?, KON = ?
```

CONTROL CARD 26

```
-5,6
DOUBLE PRECISION local program constant names
EQUIVALENCE (CON(1), local constant), ---,(CON(KON),
local constant).
```

These two statements are only for those users that desire the use of an identifying label for their constants. They are not necessary for the proper use and execution of the program.

CONTROL CARD 27

-9,19

The set of N first-order differential equations P(I) which involves the dependent variables TY(I) with its corresponding coefficients.

CONTROL CARDS 28-29

'T FOR,* F2,F2,F2/B

-2,2

PARAMETER N = ?, KON = ?

CONTROL CARD 30

-5,7

Same comments under Control Card 26 apply here.

CONTROL CARD 31

-27,59

FORTRAN statements defining the N X N coefficient matrix of the perturbation equations. The matrix is A(I,J). The subscript I corresponds to the Ith equation and the J to the Jth coefficient in the equation. Only the nonzero coefficients need be evaluated.

CONTROL CARDS 32-33

'T FOR,* F3.F3,F3/B

-2,2

PARAMETER N = ?, NQ = ?, KON = ?

CONTROL CARD 34

-15,18

A set of FORTRAN statements evaluating the NQ elements of H(I) with the N dependent variables DEP(K,1), the N derivatives of the dependent variables YPR(J,I2B,1) and the NQ elements of the input terminal constraint XF(I).

CONTROL CARDS 35-36

'T FOR,* F4,F4,F4/B
-2,2
PARAMETER N = ?, NQ = ?, KON = ?

CONTROL CARD 37

-23,26

Place the nonzero elements of $[\partial h / \partial z]$ in the matrix B.

CONTROL CARD 38

-30,33

Place the NQ elements of \dot{h} in A(I,NQ).

CONTROL CARDS 39-40

'T FOR,* F5,F5,F5/B
-2,2
PARAMETER NS3 = ?

CONTROL CARDS 41-42

'T FOR,* F6,F6,F6/B
-2,2
PARAMETER N = ?, NQ = ?, KON = ?

CONTROL CARDS 43-44

'T FOR,* F7,F7,F7/B
-2,2
PARAMETER N = ?, NQ = ?

CONTROL CARDS 45-46

'T FOR,* INTEG,INTEG,INTEG/B
-2,2
PARAMETER N = ?

CONTROL CARDS 47-48

```
'T FOR,* INTRK5,INTRK5,INTRK5/B  
-2,2  
    PARAMETER N = ?, NQ = ?
```

CONTROL CARDS 49-50

```
'T FOR,* ITER,ITER,ITER/B  
-2,2  
    PARAMETER N = ?, NQ = ?
```

CONTROL CARDS 51-52

```
'T FOR,* MLTPLY,MLTPLY,MLTPLY/B  
-2,2  
    PARAMETER NQ = ?
```

CONTROL CARDS 53-55

```
' XQT CUR  
      TOC (this starts in column 3)  
' XQT MAIN/B
```

PROGRAM INPUT DATA

CONTROL CARD 56

```
' EOF
```

DATA INPUT

FORTRAN NOTATION	COLUMNS	FORMAT	DESCRIPTION
---------------------	---------	--------	-------------

PLOT LABELS

<u>Card 1</u>	$\text{BCD}(I,1), I=1,12$	1-72	12A6	Plot label for the X-axis.
---------------	---------------------------	------	------	----------------------------

Card 2

$\text{BCD}(I,2), I=1,12$	1-72	12A6	Plot label for the Y-axis.
---------------------------	------	------	----------------------------

DEPENDENT VARIABLES

Cards 3-6

$\text{DEPO}(I), I=1,N$	1-25 26-50	D25.16 D25.16	Initial values of the dependent variable. Input two to a card. (In this example $N=8$, see the PARAMETER card).
-------------------------	---------------	------------------	--

TERMINAL VALUES

Cards 7-8

$\text{XF}(I), I=1,NQ$	1-25 26-50	D25.16 D25.16	Desired terminal values. Input two to a card. (In this example $NQ=4$, see the PARAMETER card).
------------------------	---------------	------------------	--

TIME INTERVAL

Card 9

TO	1-25	D25.16	Initial value of beginning integration time.
TF	26-50	D25.16	Initial value of final integration time.

FORTRAN NOTATION	COLUMNS	FORMAT	DESCRIPTION
<u>ACCURACY CONTROL</u>			
Card 10			
STEP			
	1-25	D25.16	Integration step size.
EPS			
	26-50	D25.16	Accuracy required of the terminal values.
<u>RUN CONTROL</u>			
Card 11			
METHOD	1-5	I5	Number of iterations with the Adams-Moulton corrector (always ≥ 1).
KK	6-10	I5	Maximum number of iterations allowed (≤ 50).
IPLT	11-15	I5	Frequency of plotting points during the iteration. Require (TO-TF) < 475 (IPLT) STEP.
PLT	16-20	I5	Frequency of plotting iterations.
IPRO	21-25	I5	Frequency of printing points during each iteration.
KEY	26-30	I5	Correction procedure desired (KEY=1 is Normal Correction Procedure, KEY=2 is Fractional Correction Procedure, KEY=3 is Minimum Norm Correction Procedure).

FORTRAN NOTATION	COLUMNS	FORMAT	DESCRIPTION
SWCH	31-35	I5	Print Control Switch. (SWCH=0 is suppressed mode, SWCH=1 is full mode).
IPCH	36-40	I5	Punch Control Switch, (IPCH=0 no punching desired. IPCH=1, punch out the n dependent variables, the time interval, and HMAX on each iteration).
IHMAX	41-45	I5	Maximum Norm Control Switch. (IHMAX=0 do not read in HMAX, IHMAX≠0 read in HMAX).

CORRECTION SCHEME (not Used for KEY=1)

Card 12 (For KEY=2)

C	1-25	D25.16	Initial Fractional Correction Constant.
DEL	26-50	D25.16	Rate of change of Fractional Correction Constant.

ALTERNATE Card 12 (for KEY=3)

FL1	1-25	D25.16	α_0 if IFL = 1 or 3 β_0 if IFL = 2 or 4
FL2	26-50	D25.16	γ if IFL ≤ 2 ρ if IFL ≥ 3
IFL	55	I1	Denotes the Minimum Norm Correction Procedure to be used. (IFL=1 use the Stepped Alpha Procedure, IFL=2 use the Stepped Beta Procedure, IFL=3 use the Variable Alpha

FORTRAN NOTATION	COLUMNS	FORMAT	DESCRIPTION
			Procedure and IFL=4 use the Variable Beta Procedure).
<u>MAXIMUM NORM (To be used only when IHMAX \neq 0)</u>			
Card 13 HMAX	1-25	D25.16	The maximum norm of the terminal constraints. (This makes sense only when used in conjunction with the Variable Alpha or Variable Beta Correction Procedures).
<u>PROGRAM CONSTANTS</u>			
Cards 14-17 PLACE	1-24	4A6	The alphanumeric identification the user identifies with the input constant.
CON(I)	25-50	D25.16	The program constant. (I is the subscript that ranges from 1 to KON. In this program KON=4, see PARAMETER card).

The next case starts at Card 3.

EQUIPMENT

This program has been checked out on the UNIVAC 1108 computer. In addition to the systems tapes, data input (5), data output (6), punch (-3), and SC4060 (17) tapes, two additional tapes are required. They are:

1. Program PCF tape (assigned P).
2. A scratch tape (3). This is usually unassigned, but when the drum is not sufficiently large to contain the plot data, it must be assigned (assigned 3) to a magnetic tape unit.

All input and output are consistent with the current requirements for handling by the UNIVAC 1108 peripheral equipment. The software requirement is a standard FORTRAN V system.

REFERENCES

1. Breakwell, J. V., Speyer, J. L., and Bryson, A. E., *Optimization and Control of Nonlinear Systems Using the Second Variation*, SIAM Journal on Control, Vol. 1, No. 2, 1963.
2. Jurovics, S. A. and McIntyre, J. E., *The Adjoint Method and Its Application to Trajectory Optimization*, ARS Journal, Vol. 32, No. 9, 1962.
3. Jazwinski, A. H., *Optimal Trajectories and Linear Control of Nonlinear Systems*, AIAA Journal, Vol. 2, No. 8, 1964.
4. McGill, R. and Kenneth, P., *Solution of Variational Problems by Means of a Generalized Newton-Raphson Operator*, AIAA Journal, Vol. 2, No. 10, 1964.
5. Merriam, C. W., *An Algorithm for the Iterative Solution of a Class of Two-Point Boundary Value Problems*, SIAM Journal on Control, Vol. 2, No. 1, 1964.
6. Sylvester, R. J. and Meyer, F., *Two Point-Boundary Problems by Quasilinearization*, SIAM Journal, Vol. 13, No. 2, 1965.
7. Roberts, S. M. and Shipman, J. S., *Continuation in Shooting Methods for Two-Point Boundary Value Problems*, Journal of Mathematical Analysis and Applications, Vol. 18, No. 1, 1967.

8. Tapley, B. D. and Lewallen, J. M., *Comparison of Several Numerical Optimization Methods*, Journal of Optimization Theory and Applications, Vol. 1, No. 1, 1967.
9. Lewallen, J. M., Tapley, B. D. and Williams, S. D., *Iteration Procedures for Indirect Trajectory Optimization Methods*, Journal of Spacecraft and Rockets, Vol. 5, No. 3, 1968.
10. Hildebrand, F. B., *Introduction to Numerical Analysis*, McGraw-Hill Book Company, 1956.
11. Clayton, E. G., *Compact Methods for Inverting Matrices and Solving Simultaneous Equations by Use of Gauss-Jordon Elimination*, Unpublished thesis submitted to the graduate school of Texas A&M for partial fulfillment of the requirements for the degree of Master of Science.

Appendix A
EARTH-MARS TRANSFER EXAMPLE

1. FORMULATION

The minimum time Earth-Mars transfer problem may be stated as follows: Determine the control history $\psi(t)$ such that a spacecraft may transfer from initial conditions corresponding to that of Earth to conditions corresponding to that of Mars in minimum time.

The differential equations of motion are

$$\dot{x}_1 = \dot{u} = v^2/r - GM/r^2 + (T/m) \sin \psi$$

$$\dot{x}_2 = \dot{v} = -uv/r + (T/m) \cos \psi$$

$$\dot{x}_3 = \dot{r} = u$$

$$\dot{x}_4 = \dot{\theta} = v/r$$

where u , v , r , and θ are the radial velocity, tangential velocity, radial position and angular position, respectively. The control variable ψ is the thrust orientation to the local horizontal. The symbol GM is the gravitational constant of the sun, and $m = m_0 - \dot{m}t$ is the vehicle mass.

When the optimization process is applied, the control variable angle ψ is eliminated from the above equations and four additional equations are required to be satisfied (Euler-Lagrange equations).

The initial boundary conditions $\mu_3(t_0) = -1$ is used in place of one of the terminal transversality boundary conditions and upon reordering, the differential equations for F1 become

$$\dot{z}_1 = \dot{x}_1 = \dot{u} = v^2/r - GM/r^2 - (T/m) \mu_1 / \sqrt{\mu_1^2 + \mu_2^2}$$

$$\dot{z}_2 = \dot{x}_2 = \dot{v} = -uv/r - (T/m) \mu_2 / \sqrt{\mu_1^2 + \mu_2^2}$$

$$\dot{z}_3 = \dot{x}_3 = \dot{r} = u$$

$$\dot{z}_4 = \dot{x}_4 = \dot{\theta} = v/r$$

$$\dot{z}_5 = \dot{x}_5 = \dot{\mu}_3 = (v^2/r^2 - 2GM/r^3) \mu_1$$

$$- (uv/r^2) \mu_2 + (v/r^2) \mu_4$$

$$\dot{z}_6 = \dot{\lambda}_1 = \dot{\mu}_4 = 0$$

$$\dot{z}_7 = \dot{\lambda}_2 = \dot{\mu}_1 = (v/r) \mu_2 - \mu_3$$

$$\dot{z}_8 = \dot{\lambda}_3 = \dot{\mu}_2 = - (2v/r) \mu_1 + (u/r) \mu_2 - (1/r) \mu_4$$

where $t = t_0$ $t = t_f$ (unspecified)

$$u(t_0) = 0.0$$

$$u(t_f) = 0.0$$

$$v(t_0) = 1.0$$

$$v(t_f) = 0.8078$$

$$r(t_0) = 1.0$$

$$r(t_f) = 1.532$$

$$\theta(t_0) = 0.0$$

$$\mu_4(t_f) = 0.0$$

$$\mu_3(t_0) = -1.0$$

The perturbation equations for F2 are

$$\dot{\delta z_1} = (2v/r) \delta z_2 + (2GM/r^3 - v^2/r^2) \delta z_3$$

$$- \left[T\mu_2/m(\mu_1^2 + \mu_2^2)^{3/2} \right] [\mu_2 \delta z_7 - \mu_1 \delta z_8]$$

$$\dot{\delta z_2} = - (v/r) \delta z_1 - (u/r) \delta z_2 + (uv/r^2) \delta z_3$$

$$+ \left[T\mu_1/m(\mu_1^2 + \mu_2^2)^{3/2} \right] [\mu_2 \delta z_7 - \mu_1 \delta z_8]$$

$$\dot{\delta z_3} = \delta z_1$$

$$\dot{\delta z_4} = (1/r) \delta z_2 - (v/r^2) \delta z_3$$

$$\dot{\delta z_5} = - (v\mu_2/r^2) \delta z_1 + \left[(2v\mu_1 - u\mu_2 + \mu_4)/r^2 \right] \delta z_2$$

$$+ \left[(6GM\mu_1/r - 2v^2\mu_1 + 2uv\mu_2 - 2v\mu_4)/r^3 \right] \delta z_3$$

$$+ (v/r^2) \delta z_6 + (v^2/r^2 - 2GM/r^3) \delta z_7$$

$$- (uv/r^2) \delta z_8$$

$$\dot{\delta z_6} = 0$$

$$\dot{\delta z_7} = (\mu_2/r) \delta z_2 - (v\mu_2/r^2) \delta z_3 - \delta z_5 + (v/r) \delta z_8$$

$$\dot{\delta z_8} = (\mu_2/r) \delta z_1 - (2\mu_1/r) \delta z_2$$

$$+ \left[(2v\mu_1 - u\mu_2 + \mu_4)/r^2 \right] \delta z_3 - (1/r) \delta z_6$$

$$- (2v/r) \delta z_7 + (u/r) \delta z_8$$

The terminal constraints for F3 are

$$\begin{aligned} h_1 &= u(t_f) \\ h_2 &= v(t_f) - 0.8078 \\ h_3 &= r(t_f) - 1.532 \\ h_4 &= \mu_4(t_f) \end{aligned}$$

The partial derivative of the terminal constraints with respect to the dependent variables and the time rates of change of the terminal constraints for F4 are

$$\left[\frac{\partial h}{\partial z} \right] = \begin{bmatrix} 1 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 & 0 & 0 \end{bmatrix}$$

and

$$\dot{h} = \begin{bmatrix} \dot{u}(t_f) \\ \dot{v}(t_f) \\ \dot{r}(t_f) \\ \dot{\mu}_4(t_f) \end{bmatrix}$$

2. PROGRAM LISTING

```
PARAMETER N=6, NQ=4, K01=4, NS3=2  
PARAMETER NS=50, NS1=50, NS2=7, NS4=NS3/2  
COMMON/BC1/BACK(N),BACK2(N),HICK(NQ),HICK2(NQ),CIOLD,CIOLD2,TFB,  
1 TFB2,ICOUNT,KOUNT  
DOUBLE PRECISION BACK,BACK2,HICK,HICK2,CIOLD,CIOLD2,TFB,TFB2  
COMMON/TIMER/XJ,XMEX,XN  
COMMON/DIRV/DEP(N,NQ),YPR(N,4,NQ)  
DOUBLE PRECISION DEP,YPR  
COMMON/INITAL/DEPO(N),T0,TF,EPS  
DOUBLE PRECISION DEPO,T0,TF,EPS  
COMMON/PRNT/SWCH  
INTEGER SWCH  
COMMON/EVAL/VIND,DELT,INDI  
DOUBLE PRECISION VIND,DELT  
COMMON/MIDTR/J,JKK  
COMMON/NUM/STEP,KEY,KK,PLT,IPLT,IPRO  
INTEGER PLT  
DOUBLE PRECISION STEP  
COMMON/MATRIX/A(NQ,NQ),R(NQ),XF(NQ)  
DOUBLE PRECISION A,R,XF  
COMMON/LEWALN/HNN,HMAX,NTHRU  
DOUBLE PRECISION HNN,HMAX  
COMMON/NEWTON/CI,CRAT  
DOUBLE PRECISION CI,CRAT  
COMMON/TRL/B(NQ),ISET  
DOUBLE PRECISION B  
COMMON/CONST/CON(KON)
```

```

DOUBLE PRECISION CUN

COMMON/DIFEQ/P(N),TY(N),T

DOUBLE PRECISION P,TY,T

COMMON/MODE/S(N,3),JAT,NU,IONE,ITWO,IERRCT

DOUBLE PRECISION S

COMMON/COUNT/METHOD,ILUC,I1B,I2B,I3B,K2B

COMMON/FLAG/FL1,FL2,IFL

DOUBLE PRECISION FL1,FL2

COMMON/KPCH1/IPCH,IHMAX

COMMON/LOGIC/BOOL

LOGICAL BOOL

COMMON/STUFF/XMAX(NS4,NS2),YMAX(NS4,NS2),XMIN(NS4,NS2),IT2,UPLT,
1           YMIN(NS4,NS2),STORB(NS1,2),INDEX(NS2),IWRT(NS2)

COMMON/PLUTO/BCD(12,NS2)

CALL F5

5 CONTINUE

CALL RESET

      XMEX = 0.0

      NTHRU = 1

CALL F6

DO 405 J=1,N

      BACK(J)=DEPO(J)

405 CONTINUE

      TFB=1F

      KOUNT=0

      CIOLD=CI

      ICOUNT = 0

5      ICOUNT = ICOUNT + 1

      CALL CRDPCH

```

```
CALL TIME(J)
XJ = J
XJ = XJ/1000.0
XK = XJ
XMEX = XJ - XMEX

CALL F7
IT=1
CALL FPRNT(IT)
IT=5
CALL FPRNT(IT)
CALL FPLEW(IT)
CALL INTEG
IT=7
CALL FPRNT(IT)
CALL FPLEW(IT)

CALL TIME(J)
XJ = J
XJ = XJ/1000.0
XN = XJ - XK
XMEX=XJ

IT=3
CALL FPRNT(IT)
CALL CONVRG(KIK)
JPLT = 0
CALL FPLOT
GO TO (5,35),KIK
35      CALL TIME(J)
XJ = J
XJ = XJ/1000.0
```

IT=25
CALL FPRNT(IT)
JPLT = 1
CALL FPLOT
40 CONTINUE
GO TO 3
END

```

SUBROUTINE ABAM

PARAMETER N=8, NQ=4

COMMON/EVAL/VIND,DELT,INDI

DOUBLE PRECISION VIND,DELT

COMMON/COUNT/METHOD,ILOC,I1B,I2B,I3B,K2B

COMMON/DIFEQ/P(N),DEPVAR(N),XN

DOUBLE PRECISION P,DEPVAR,XN

COMMON/DIRV/DEP(N,NQ),YPR(N,4,NQ)

DOUBLE PRECISION DEP,YPR

COMMON/MODE/S(N,3),JAY,NU,IONE,ITWO,IERRCT

DOUBLE PRECISION S

DOUBLE PRECISION XIP(N)

DOUBLE PRECISION A(5),B(4)

DATA (B(J),J=1,4)/-.5/5D0,.154166666666666D1,-.245833333333333D1
1,.229166666666666D1/

DATA (A(J),J=1,5)/-.263888888888888D-1,.147222222222222D0,
1-0.366666666666667D0,.897222222222222D0,.348611111111111D0/

C
C      THIS ROUTINE INTEGRATES THE DIFFERENTIAL EQUATIONS USING AN
C
C      ADAMS-BASHPORD PREDICTOR WITH AN ADAMS-MOULTON CORRECTOR.
C

NS=NQ

      VIND = VIND + DELT

      XN=VIND

      DO 40 JAY = 1,NS

      ISAV = METHOD

```

```

DO 10 J = 1,N

    DEPVAR(J) = DEP(J,JAY) + DELT*(B(1)*YPR(J,I3B,JAY) +
1                                + B(2)*YPR(J,I2B,JAY) + B(3)*YPR(J,I1B,JAY) +
1                                + B(4)*YPR(J,ILOC,JAY))

10      XIP(J) = DEP(J,JAY) + DELT*(A(1)*YPR(J,I3B,JAY) +
1                                + A(2)*YPR(J,I2B,JAY) + A(3)*YPR(J,I1B,JAY) +
1                                + A(4)*YPR(J,ILOC,JAY))

IF(JAY.NE.1) GO TO 12

CALL F1

GO TO 14

12 CONTINUE

CALL F2

14 CONTINUE

15      DO 20 J = 1,N

        DEP(J,JAY) = DELT*A(5)*P(J) + XIP(J)

20      DEPVAR(J) = DEP(J,JAY)

IF(JAY.NE.1) GO TO 22

CALL F1

GO TO 24

22 CONTINUE

CALL F2

24 CONTINUE

IF(ISAV.EQ.1) GO TO 25

ISAV = ISAV - 1

GO TO 15

25      DO 30 J = 1,N

        YPR(J,I3B,JAY) = P(J)

IF(JAY.NE.1) GO TO 40

DO 35 J = 1,N

```

35 S(J+ITWO) = DEPVAR(J)
40 CONTINUE
 ISAV = I3B
 I3B = I2B
 I2B = I1B
 I1B = ILOC
 ILOC = ISAV
 ISAV = IONE
 IONE = ITWO
 ITWO = ISAV
 RETURN
 END

SUBROUTINE CONVRG(KIK)

PARAMETER N=8, NQ=4

COMMON/MATRIX/AA(NQ,NQ),HNEW(NQ),XF(NQ)

DOUBLE PRECISION AA,HNEW,XF

COMMON/DIRV/DEP(N,NQ),YPR(N,4,NQ)

DOUBLE PRECISION DEP,YPR

COMMON/LEWALN/HNN,HMAX,NTHRU

DOUBLE PRECISION HNN,HMAX

COMMON/TRL/B(NQ),ISET

DOUBLE PRECISION B

COMMON/INITAL/DEPO(N),T0,TF,EPS

DOUBLE PRECISION DEPO,T0,TF,EPS

COMMON/NUM/STEP,KEY,KK,PLT,IPLT,IPRO

INTEGER PLT

DOUBLE PRECISION STEP

COMMON/FLAG/FL1,FL2,IFL

DOUBLE PRECISION FL1,FL2

COMMON/BC1/BACK(N),BACK2(N),HICK(NQ),HICK2(NQ),CIOLD,CIOLD2,TFB,

1 TFB2,ICOUNT,KOUNT

DOUBLE PRECISION BACK,BACK2,HICK,HICK2,CIOLD,CIOLD2,TFB,TFB2

COMMON/NEWTON/CI,CRAT

DOUBLE PRECISION CI,CRAT

COMMON/KPCH1/IPCH,IHMAX

DOUBLE PRECISION HNO,HN02,XN,X(NQ)

DIMENSION J2(NQ),I2(NQ)

LOGICAL BOOL

KIK=1

C
C DETERMINE THE TERMINAL NORM
C
CALL F3
C
C CALCULATE THE NORM OF THE TERMINAL CONSTRAINTS
C
HNN=0.D0
DO 15 J=1,NQ
15 HNN = HNN + HNEW(J)*HNEW(J)
HNN=DSQRT(HNN)
ITT=11
CALL FPRNT(ITT)
CALL FPLEW(ITT)
C
C EVALUATE THE MAXIMUM NORM OF THE TERMINAL CONSTRAINTS
C
IF(IHMAX.NE.0)GO TO 50
IF(NTHRU.EQ.1) HMAX=HNN
50 CONTINUE
HMAX=UMAX1(HMAX,HNN)
GO TO(100,200,300),KEY
100 CONTINUE
C
C NORMAL CORRECTION PROCEDURE
C
C EVALUATE THE MATRIX AND SET UP THE DISSATISFACTION VECTOR
C
CALL F4

DO 130 J=1,NQ

B(J)=HNEW(J)

130 CONTINUE

C

C EVALUATE THE INVERSE AND ADD ON THE CORRECTIONS

C

150 CONTINUE

CALL MINVUP(AA,NQ,K,X,J2,12)

IF(K.EQ.1) GO TO 1500

CALL CURVEC(AA,B,NQ,HNEW)

ITT=15

CALL FPRNT(ITT)

CALL ITER

C

C DETERMINE IF CONVERGENCE HAS BEEN ACHIEVED

C

160 BOOL = .TRUE.

DO 180 J=1,NQ

XN=DABS(HNEW(J))

IF(XN.GT.EPS) BOOL = .FALSE.

180 CONTINUE

IF(HNN.LE.EPS .AND. BOOL) GO TO 70

NTHRU=2

GO TO 5

200 CONTINUE

C

C FRACTIONAL PROCEDURE

C

IF(NTHRU.EQ.1) GO TO 220

```

IF(HNN.GE.HNU) GO TO 250
DO 210 J=1,N
    BACK2(J)=BACK(J)
    BACK(J)=DEPO(J)
210 CONTINUE
    TFB2=TFB
    TFB=TF
C
C      DETERMINE THE FRACTIONAL CORRECTION
C
C      CI=CI+CRAT
        IF(CI.GT.1.00) CI=1.00
C
C      EVALUATE THE MATRIX AND SET UP THE DISSATISFACTION VECTOR
C
220 CALL F4
    ISET=1
    ISET1=0
    IF(NTHRU.EQ.1) ISET1=1
    DO 230 J=1,NQ
        B(J)=-CI*HNEW(J)
230 CONTINUE
    ITT=19
    CALL FPRNT(ITT)
C
C      EVALUATE THE INVERSE AND ADD ON THE CORRECTIONS
C
    CALL MINVDP(AA,NQ,K,X,U2,I2)
    IF(K.EQ.1) GO TO 1500

```

```

CALL CORVEC(AA,B,NQ,HNEW)

ITT=17

CALL FPRNT(ITT)

CALL ITER

KOUNT=KOUNT+1

DO 240 J=1,NQ

    HICK2(J)=HICK(J)

    HICK(J)=HNEW(J)

240 CONTINUE

CIOLD2=CIOLD

CIOLD=CI

HNO2=HNO

HNO=HNN

GO TO 160

250 IF(CI.EQ.CRAT) GO TO 280

255 ISET=ISET+1

ICOUNT=ICOUNT-1

KOUNT=KOUNT+1

DO 260 I=1,N

    DEPU(I)=BACK(I)

260 CONTINUE

IF(ISET.LE.2) CI = CIOLD

TF=TFB

CI=CI-CRAT

IF(CI.LT.CRAT) CI=CRAT

DO 270 J=1,NQ

    HNEW(J)=CI*HICK(J)/CIOLD

270 CONTINUE

ITT=17

```

```

CALL FPRNT(ITT)
CALL ITER
GO TO 5
280 IF(ISET1.EQ.1) GO TO 1000
IF(ICOUNT.LT.2) GO TO 1000
ISET1=1
DO 285 I=1,N
    BACK(I)=BACK2(I)
285 CONTINUE
DO 290 J=1,N
    HICK(J)= HICK2(J)
290 CONTINUE
C1OLD=C1OLD2
TFB=TFB2
HNO=HN02
ISET=0
ICOUNT=ICOUNT-1
GO TO 255
300 CONTINUE
310 IF(IFL.GT.2) GO TO 320
IF(HNN.LT.FL2) KEY=1
IF(KEY.EQ.1) GO TO 100
C
C      FORM THE MATRIX AND SET UP THE DISSATISFACTION VECTOR
C
320 CALL F4
ITT=19
CALL FPRNT(ITT)
CALL MLTPLY

```

```

111=21
CALL FPRNT(111)
HNO=HNN
CALL MINVDP(AA,NQ,K,X,W2,I2)
IF(K.EQ.1) GO TO 1500
CALL CORVEC(AA,B,NQ,HNEW)
CALL ITER
111=15
CALL FPRNT(111)
GO TO 160
70 WRITE(6,71)
71 FORMAT(1HU,19X,66H***** CONVERGENCE HAS BEEN ACHIEVED
        1***** )
        GO TO 35
1000 WRITE(6,1010)
1010 FORMAT(1HU,19X,66H***** CONVERGENCE HAS NOT BEEN ACHIEVE
        1D ***** )
        GO TO 35
1500 CONTINUE
        WRITE(6,1510)
1510 FORMAT(1HU,41X,22HTHE MATRIX IS SINGULAR )
        GO TO 1000
5 CONTINUE
IF(TF.LT. 0.00) GO TO 1000
IF(ICOUNT.EQ.KK) GO TO 1000
IF(KOUNT.EQ.KK) GO TO 1000
RETURN
35 CONTINUE
KIK=2

```

RETURN

END

```
SUBROUTINE CURVEC(A,B,NQ,C)
DOUBLE PRECISION A(NQ,NQ),B(NQ),C(NQ)

C
C      THIS ROUTINE TAKES THE MATRIX A MULTIPLIES IT TIMES THE VECTOR B
C
C      AND STORES THE RESULT IN THE VECTOR C.

C
DO 10 I=1,NQ
      C(I)=0.D0
      DO 5 J=1,NQ
            C(I)=C(I)+A(I,J)*B(J)
5       CONTINUE
10    CONTINUE
      RETURN
END
```

```
SUBROUTINE CRDPCH

PARAMETER N=8

PARAMETER IUNIT=-3

COMMON/INITAL/DEPO(N),TO,TF,EPS

DOUBLE PRECISION DEPO,TO,TF,EPS

COMMON/KPCH1/IPCH,IHMAX

COMMON/LEWALN/HNN,HMAX,NTHRU

DOUBLE PRECISION HNN,HMAX

100 FORMAT(2D25.16)

IF(IPCH.EQ.0)RETURN

C

C      IF IPCH = 0 RETURN. FOR IPCH NE 0 THE INITIAL VALUES OF THE
C
C      DEPENDENT VARIABLES, BEGINNING AND FINAL TIME, AND THE MAXIMUM
C
C      NORM OF THE TERMINAL CONSTRAINTS IS PUNCHED OUT ON CARDS.
C

WRITE(IUNIT,100)DEPO

WRITE(IUNIT,100)TO,TF

WRITE(IUNIT,100)HMAX

RETURN

END
```

```

SUBROUTINE FPLEW(I)

PARAMETER N=8, NQ=4, NS3=2, KON=4

DOUBLE PRECISION TIME,DEG

DATA TIME/.5813225602/,DEG/.5729577951308232U2/

PARAMETER NS=50, NS1=50, NS2=7, NS4=NS3/2

COMMON/CONST/CON(KON)

DOUBLE PRECISION CON

COMMON/STUFF/XMAX(NS4,NS2),YMAX(NS4,NS2),XMIN(NS4,NS2),IT2,IPLT,
1           YM1N(NS4,NS2),STORB(NS1,2),INDEX(NS2),IWRT(NS2)

C
C   INFORMATION IS STORED COLUMN-WISE
C

COMMON/INITAL/DEPO(N),IO,TF,EPS

DOUBLE PRECISION DEPO,IO,TF,EPS

COMMON/DIRV/DEP(N,NQ),TPR(N+4,NQ)

DOUBLE PRECISION DEP,TPR

COMMON/LEWALN/HNN,HMAX,NTHRU

DOUBLE PRECISION HNN,HMAX

COMMON/EVAL/VIND,DELT,INDI

DOUBLE PRECISION VIND,DELT

COMMON/MIDTR/J,JKK

COMMON/BC1/BACK(N),BACK2(N),HICK(NQ),HICK2(NQ),CIOLD,CIOLD2,TFB,
1           TFB2,ICOUNT,KOUNT

DOUBLE PRECISION BACK,BACK2,HICK,HICK2,CIOLD,CIOLD2,TFB,TFB2

COMMON/NUM/STEP,KEY,KK,PLT,IPLT,IPRO

DOUBLE PRECISION STEP

INTEGER PLT

```

```

COMMON/COUNT/METHOD,KLOC,K1B,K2B,K3B,I2B
DIMENSION X(NS4),Y(NS4)
DATA A1/.1E37/
DIMENSION STORA(NS,NS3,NS2)

C
C      IPLT IS THE FREQUENCY OF PLOTTING EACH POINT PER CURVE
C
C      PLT  IS THE FREQUENCY OF PLOTTING EACH ITERATION IF 0 OR LESS
C            NO PLOTTING IS DESIRED
C
C
IF(PLT.LE.0)RETURN
K=I-4
GO TO (1,100,2,100,3,100,4),K
1 K1=1
IF(ICOUNT.EQ.1) K2=1
L=MOD(ICOUNT,PLT)
INDEX(K2) = 1
IWRT(K2) = 0
REWIND 3
IF(ICOUNT.EQ.1) GO TO 15
READ(3) STORA
REWIND 3
15 CONTINUE
DO 20 IR=1,NS4
      XMAX(IR,K2) = -A1
      YMAX(IR,K2) = -A1
      XMIN(IR,K2) = A1
      YM1N(IR,K2) = A1
20 CONTINUE

```

```

GO TO 5

2 K1=K1+1
    INDEX(K2) = INDEX(K2) + 1

5 CONTINUE
    X(1) = TIME*VIND
    Y(1) = DEG*DATAN2(DEP(7,1),DEP(8,1))
    GO TO 100

6 CONTINUE
    IT2=ICOUNT
    IF(ICOUNT.EQ.1.OR. L.EQ.0) K2=K2+1
    IF(K2.GT.NS2) K2=NS2
    RETURN

4 STORB(ICOUNT,1)=ICOUNT
    STORB(ICOUNT,2)=HNN
    RETURN

3 IF(J.EQ.0) GO TO 2
    RETURN

100 CONTINUE
DO 120 LL=1,NS4
    LJ = LL + LL
    LK = LJ - 1
    STORA(K1,LK,K2) = X(LL)
    STORA(K1,LJ,K2) = Y(LL)
    XMAX(LL,K2) = AMAX1(XMAX(LL,K2),X(LL))
    YMAX(LL,K2) = AMAX1(YMAX(LL,K2),Y(LL))
    XMIN(LL,K2) = AMIN1(XMIN(LL,K2),X(LL))
    YMINT(LL,K2) = AMIN1(YMIN(LL,K2),Y(LL))

120 CONTINUE
IF(K1.LT.NS .AND. K.NE.3) RETURN

```

```
    IWRT(K2) = IWRT(K2) + 1
    IWRT(1) = MAXU(IWRT(1),IWRT(K2))
    WRITE(3)STORA
    IF(K.EQ.3) GO TO 6
    IF(ICOUNT.EQ.1) GO TO 130
    READ(3) STORA
    BACKSPACE 3
130 CONTINUE
    DO 140 LL=1,NS4
        LJ = LL + LL
        LK = LJ - 1
        STORA(1,LK,K2) = X(LL)
        STORA(1,LJ,K2) = Y(LL)
140 CONTINUE
    K1 = 1
    INDEX(K2) = INDEX(K2) + 1
    RETURN
    END
```

```

SUBROUTINE FPLOT

PARAMETER NS3=2

PARAMETER NS=50, NS1=50, NS2=7, NS4=NS3/2

COMMON/STUFF/XMAX(NS4,NS2),YMAX(NS4,NS2),XMIN(NS4,NS2),IT2,IPLT,
I           YMIN(NS4,NS2),STORB(NS1,2),INUEX(NS2),IWRT(NS2)

COMMON/LEWALN/HNN,HMAX,NTHRU

DOUBLE PRECISION HNN,HMAX

COMMON/NUM/STEP,KEY,KK,PLT,IPLT,IPRO

DOUBLE PRECISION STEP

INTEGER PLT

COMMON/PLUTO/BCD(12,NS3)

DIMENSION A(32),B(24),C(6),D(3),X(NS),Y(NS),ISYMB(25),G(2),W0(2)

DIMENSION STORA(NS,NS3,NS2)

DATA FIT2/112.5/

DATA (A(I),I=1,24)/1.E+14,1.E+13,1.E+12,1.E+11,1.E+10,1.E+09,
1           1.E+08,1.E+07,1.E+06,1.E+05,1.E+04,1.E+03,
2           1.E+02,1.E+01,1.E+00,1.E-01,1.E-02,1.E-03,
3           1.E-04,1.E-05,1.E-06,1.E-07,1.E-08,1.E-09/

C
C

DATA (B(I),I=1,32)/6H1.E+14, 6H1.E+13, 6H1.E+12, 6H1.E+11,
1           6H1.E+10, 6H1.E+09, 6H1.E+08, 6H1.E+07,
2           6H1.E+06, 6H1.E+05, 6H1.E+04, 6H1.E+03,
3           6H1.E+02, 6H1.E+01, 6H1.E+00, 6H1.E-01,
4           6H1.E-02, 6H1.E-03, 6H1.E-04, 6H1.E-05,
5           6H1.E-06, 6H1.E-07, 6H1.E-08, 6H1.E-09,
6           6H1.E-10, 6H1.E-11, 6H1.E-12, 6H1.E-13,

```

7

6H1.E-14, 6H1.E-15, 6H1.E-16, 6H1.E-17/

C

C

DATA (C(I),I=1,06)/6HNORM 0, 6HF THE , 6HTERMIN, 6HAL CON,
1 6HSRAIN, 6HTS /

C

C

DATA (D(I),I=1,03)/6HITERAT, 6HIUN CO, 6HUNT /

C

C

DATA E1/1.0E+06/,E2/1.0E-01/, (G(I),I=1,2)/6HSCALE ,6HFACTOR/

C

C

DATA (ISYMB(I),I=1,25)/1H1,1H2,1H3,1H4,1H5,1H6,1H7,1H8,1H9,1HA,1HI
1,1HC,1HD,1HE,1HF,1HG,1HI,1HJ,1HK,1HL,1HM,1HN,1HO,1HP,1HQ/

C

C

C

C DETERMINE IF PLOTTING IS DESIRED.

C

IF(PLT.LE.0)RETURN

C

C DETERMINE THE MAXIMUM NUMBER OF CURVES TO BE PLOTTED (K2).

C

LD = IT2 + 2 * (PLT - 1)

LD = LD / PLT

K2 = MIN0(LD,NS2)

KT = 1

IF(JPLT.EQ.0) KT = K2

```

IWRT1 = IWRT(KT)

C
C      IF JPLT IS ZERO -- ONLY THE CURVE FOR THE CURRENT ITERATION IS
C      DESIRED, OTHERWISE PLOT ALL K2 CURVES.
C

DO 200 I=1,NS3+2

      I1 = (I+1)/2

      LP = -1

      REWIND 3

C
C      DETERMINE THE PLOTTING BOUNDARIES.
C

      XR = XMAX(I1,KT)
      XL = XMIN(I1,KT)
      YT = YMAX(I1,KT)
      YB = YMIN(I1,KT)
      IF(KT.EQ.K2) GO TO 115
      DO 110 K=2,K2

          XR = AMAX1(XR,XMAX(I1,K))
          XL = AMIN1(XL,XMIN(I1,K))
          YT = AMAX1(YT,YMAX(I1,K))
          YB = AMIN1(YB,YMIN(I1,K))

110      CONTINUE
115      CONTINUE

C
C      DETERMINE THE SCALE FACTORS FOR X AND Y SUCH THAT X,Y LT 1,E+06.
C

      SX = 1.0
      SY = 1.0

```

```

XS = AMAX1(ABS(XR),ABS(XL))
YS = AMAX1(ABS(YT),ABS(YB))

120      CONTINUE
          IF(XS.LT.E1) GO TO 125
          SX = SX*E2
          XS = XS*E2
          GO TO 120

125      CONTINUE
          IF(YS.LT.E1) GO TO 130
          SY = SY*E2
          YS = YS*E2
          GO TO 125

130      CONTINUE
C
C      SCALE THE PLOT BOUNDARIES
C
          XR = XR*SX
          XL = XL*SX
          YT = YT*SY
          YB = YB*SY
          DO 150 J=1,1WKRT1
              JP = NS*(J-1)
C
C      READ THE DATA TAPE
C
              READ(3) S1ORA
              DO 140 K=KT,K2
C
C      SCALE THE K-TH CURVE FOR THIS GRID

```

C

NP = INDEX(K) - JP

IF(NP.LT.1) GO TO 140

NP = MIN0(NP,NS)

ISYM = ISYMB(K)

DO 135 L=1,NP

X(L) = STORA(L,I,K)*SX

Y(L) = STORA(L,1+I,K)*ST

135

CONTINUE

C

C PLOT ALL DATA ON THE SAME GRID AND IF MORE DATA MUST BE READ FROM
C TAPE 3 ** PLOT IT ALSO

C

CALL QUIKML(LP,XL,XR,YB,YT,ISYM,

1 BCD(1,I),BCD(1,1+I),NP,X,Y)

LP = 0

140

CONTINUE

150

CONTINUE

C

C IF THE VARIABLES WERE SCALED PRINT THE SCALE FACTORS ON THE GRID

C

IF(SX.GT..5) GO TO 160

CALL BINDEC(SX,NC,W0(1),W0(2))

CALL PRINT(474, 5,8,0,12,G)

CALL PRINT(578, 5,8,0,NC,W0)

160

CONTINUE

IF(SY.GT..5) GO TO 170

CALL BINDEC(SY,NC,W0(1),W0(2))

CALL PRINT(474,20,8,0,12,G)

```
        CALL PRINT(570,20,8,0,NC,W0)
170      CONTINUE
C
C      PLOT THE REMAINING GRIDS
C
200  CONTINUE
C
C      DETERMINE IF THIS IS THE END OF A JOB
C
REWIND 3
IF(JPLT.EQ.0) RETURN
C
C      PREPARE FOR SIMI-LOG GRIDS
C
C
CALL RSET(0)
CALL MODE(0,1,8,8)
K1=0
DO 5 I=1,24
IF(HMAX.GE.A(I)) GO TO 10
K1=I
5 CONTINUE
10 K2=K1
IF(K2.EQ.0)K2=1
SET2=A(K2)
SET1=SET2*1.E-08
XKK=FLOAT(KK)
IF(XKK.LT.10.) XKK = 10.
```

```
FIT=900./XKK  
FIX=50.  
IT1=-IT2  
CALL GRIDGN(100,1000,050,0950,FIT,FIT,5,5)  
CALL PLOT1(1,1,0,,XKK,SET1,SET2,STORB(1,1),STORB(1,2),IT1,1,1HX)  
CALL PRINT(486,980,8,0,15,D)  
CALL PRINT(20,244,0,16,32,C)  
JK=KK+1  
DO 20 JI=1,JK  
  I=JI-1  
  CALL LABELX(I,1,0)  
20 CONTINUE  
K1=K2-1  
DO 30 I=1,9  
  IFIX=FIX  
  CALL PRINT(50,IFIX,8,0,6,B(I+K1))  
  FIX=FIX+FIT2  
30 CONTINUE  
CALL FILMAV(0)  
CALL DMPBUF  
CALL MODE(0,0,8,8)  
RETURN  
END
```

```
SUBROUTINE FPRNT(I)

PARAMETER N=8 , NQ=4

COMMON/COUNT/METHOD,KLOC,K1B,K2B,K3B,I2B

COMMON/TIMER/XJ,XMEX,XN

COMMON/BC1/BACK(N,2),BACK2(NQ,2),BACK3(4),ICOUNT,KOUNT

DOUBLE PRECISION BACK,BACK2,BACK3

COMMON/DIRV/DEP(N,NQ),YPR(N,4,NQ)

DOUBLE PRECISION DEP,YPR

COMMON/INITAL/DEPO(N),T0,TF,EPS

DOUBLE PRECISION DEPO,T0,TF,EPS

COMMON/PRNT/SWCH

INTEGER SWCH

COMMON/EVAL/VIND,DELT,INDI

DOUBLE PRECISION VIND,DELT

COMMON/MIDTR/J,JKK

COMMON/NUM/STEP,KEY,KK,PLT,IPLT,IPRO

INTEGER PLT

DOUBLE PRECISION STEP

COMMON/MATRIX/A(NQ,NQ),H(NQ),XF(NQ)

DOUBLE PRECISION A,H,XF

COMMON/LEWALN/HNN,HMAX,NTHRU

DOUBLE PRECISION HNN,HMAX

COMMON/NEWTON/CI,CRAT

DOUBLE PRECISION CI,CRAT

COMMON/TRL/B(NQ ),ISET

DOUBLE PRECISION B

JSWCH=I+SWCH
```

```

GO TO(90,100,90,120,90,130,160,140,150,150,175,170,90,180,90,190,
      X      90,200,90,210,90,220, 90,190,90,110),JSWCH

90 RETURN

100 WRITE(6,101)ICOUNT,XJ,XMEX

101 FORMAT(1H1,37X,14HBEGINNING THE ,I3,13H TH ITERATION ,/,21X,
          1           51HTIME AT THE COMMENCEMENT OF FORWARD INTEGRATION IS ,
          2           F12.4,/,38X,1/MELAPSED TIME WAS  ,F12.4)

GO TO 90

110 WRITE(6,111)XJ

111 FORMAT(1H0,37X,18HTHE FINAL TIME IS ,F12.4)

GO TO 90

120 WRITE(6,121)XJ,XN

121 FORMAT(1H0,23X,45HTIME AT COMPLETION OF FORWARD INTEGRATION IS ,
          1           F12.4,/,38X,1/MELAPSED TIME WAS  ,F12.4)

GO TO 90

130 WRITE(6,131) TO

131 FORMAT(1H0,22X,34HTHE DEPENDENT VARIABLES AT TIME = ,D25.16)

135 DO 139 K2=1,NQ,2

      K3=K2+1

      IF(K3.GT.NQ) GO TO 137

      WRITE(6,136)(K1,K2,DEP(K1,K2),K1,K3,DEP(K1,K3),K1=1,N)

136 FORMAT( //,(20X,I2,1X,I2,1X,D25.16,4X,I2,1X,I2,1X,D25.16))

      GO TO 139

137      WRITE(6,138)(K1,K2,DEP(K1,K2),K1=1,N)

138 FORMAT( //,(37X,I2,1X,I2,2X,D25.16))

139 CONTINUE

GO TO 90

140 WRITE(6,131)VIND

GO TO 135

```

```

150 IF(J.NE.0) GO TO 90
      WRITE(6,151)JKK,VIND
151 FORMAT(1H0,24X,14HPRINI COUNT = ,I5,5X,7HTIME = ,D25.16)
      GO TO 135
160 NQ1=NQ-1
      NP=N-NQ1
      WRITE(6,161)ICOUNT,NQ1
161 FORMAT(1H0,44X,I2,13H TH ITERATION,/,,25X,35HDISPLAYING FINAL TIME
      1AND THE LAST ,I2,20H DEPENDENT VARIABLES )
      WRITE(6,162)TF,(K1,DEPU(NP+K1),K1=1,NQ1)
162 FORMAT(1H0,37X,5HTF = ,D25.16,/,(21X,I2,2X,D25.16,6X,I2,2X,D25.16)
      1)
      GO TO 90
170 WRITE(6,171)(K,H(K),K=1,NQ)
171 FORMAT(1H0,37X,30HTHE TERMINAL CONSTRAINT VECTOR ,/,,(21X,I2,2X,
      1          D25.16,6X,I2,2X,D25.16))
175 WRITE(6,176)HNN
176 FORMAT(/,22X,37HTHE NORM OF THE TERMINAL CONSTRAINTS ,D25.16)
      GO TO 90
180 WRITE(6,181)CI
181 FORMAT(/,23X,35HTHE FRACTIONAL CORRECTION CONSTANT ,D25.16)
      GO TO 90
190 WRITE(6,191)ICOUNT,(K,H(K),K=1,NQ)
191 FORMAT(/,33X,23HTHE CORRECTIONS AT THE ,I2,13H TH ITERATION ,/,(21X,I2,2X,D25.16,6X,I2,2X,D25.16))
      GO TO 90
200 WRITE(6,181)CI
      WRITE(6,201)ISET,ICOUNT
201 FORMAT(1H0,22X,12HTHIS IS THE ,I2,32H TH ATTEMPT TO CORRECT FROM T

```

```

1HE ,I2,13H IH ITERATION )

GO TO 190

210 CONTINUE

WRITE(6,225)

DO 219 K2=1,NQ,2

K3=K2+1

IF(K3.GT.NQ) GO TO 217

WRITE(6,136)(K1,K2,A(K1,K2),K1,K3,A(K1,K3),K1=1,NQ)

GO TO 219

217      WRITE(6,138)(K1,K2,A(K1,K2),K1=1,NQ)

219 CONTINUE

GO TO 90

220 CONTINUE

WRITE(6,230)

DO 222 K2=1,NQ,2

K3=K2+1

IF(K3.GT.NQ) GO TO 221

WRITE(6,136)(K1,K2,A(K1,K2),K1,K3,A(K1,K3),K1=1,NQ)

GO TO 222

221      WRITE(6,138)(K1,K2,A(K1,K2),K1=1,NQ)

222 CONTINUE

225 FORMAT(1HU,47X,12HTHE A MATRIX )

230 FORMAT(1HU,46X,14HTHE A*A MATRIX )

GO TO 90

END

```

```

SUBROUTINE F1

PARAMETER N=8, KON=4

COMMON/CONST/CON(KON)

DOUBLE PRECISION CON

DOUBLE PRECISION WOGD,WOED,TH,GM

EQUIVALENCE (CON(1),GM),(CON(2),WOGD),(CON(3),WOED),(CON(4),TH)

COMMON/DIFEQ/P(N),TY(N),T

DOUBLE PRECISION P,TY,T

DOUBLE PRECISION XNN ,DSQRT

XNN=(WOGD-WOED*T)*DSQRT(TY(7)**2+TY(8)**2)

P(1)=TY(2)**2/TY(3)-GM/ TY(3)**2-TH*TY(7)/XNN

P(2)=-TY(1)*TY(2)/TY(3)-TH*TY(8)/XNN

P(3)=TY(1)

P(4)=TY(2)/TY(3)

P(5)=(((TY(2)**2-2.00*GM/TY(3))*TY(7))-(TY(1)*TY(2)*TY(8))+(TY(2)*
1 TY(6)))/(TY(3)**2)

P(6)=0.00

P(7)=TY(2)*TY(8)/TY(3)-TY(5)

P(8)=(-2.00*TY(2)*TY(7)+TY(1)*TY(8)-TY(6))/TY(3)

RETURN

END

```

```

SUBROUTINE F2

PARAMETER N=8, KON=4

COMMON/CONST/CON(KON)

DOUBLE PRECISION CON

DOUBLE PRECISION WOOGD,WOED,TH,GM

EQUIVALENCE (CON(1),GM),(CON(2),WOOGD),(CON(3),WOED),(CON(4),TH)

DOUBLE PRECISION R,R2,R3,DSQRT

COMMON/DIFEQ/P(N),TY(N),T

DOUBLE PRECISION P,TY,T

COMMON/MQDE/S(N,3),JAY,NU,IONE,ITWO,IERRCT

DOUBLE PRECISION S

DOUBLE PRECISION A(N,N)

IF(JAY.NE.2) GO TO 30

IF(NU.NE.1) GO TO 15

DO 10 J=1,N

      S(J,3)=.5DD*(S(J,IONE)+S(J,ITWO))

10 CONTINUE

15 CONTINUE

J=3

IF(NU.EQ.1)J=IONE

IF(NU.EQ.4)J=ITWO

DO 25 I=1,N

DO 20 K=1,N

A(I,K)=0.D0

20 CONTINUE

25 CONTINUE

R2=S(7,J)**2+S(8,J)**2

```

```

R=(W0GD-W0ED*T)*DSQRT(R2)

R3=TH/(R*R2)

R=S(3,J)

A(1,2)=(2.D0*S(2,J))/R

A(1,3)=(2.D0*GM/R-S(2,J)**2)/R**2

A(1,7)=-R3*S(8,J)**2

A(1,8)=R3*S(7,J)*S(8,J)

A(2,1)=-S(2,J)/R

A(2,2)=-S(1,J)/R

A(2,3)=(S(1,J)*S(2,J))/R**2

A(2,7)=R3*S(7,J)*S(8,J)

A(2,8)=-R3*S(7,J)**2

A(3,1)=1.D0

A(4,2)=1.D0/R

A(4,3)=-S(2,J)/R**2

A(5,1)=-(S(2,J)*S(8,J))/R**2

A(5,2)=(2.D0*S(2,J)*S(7,J)-S(1,J)*S(8,J)+S(6,J))/R**2

A(5,3)=((6.D0*GM/R-2.D0*S(2,J)**2)*S(7,J)

1      +2.D0*S(2,J)*(S(1,J)*S(8,J)-S(6,J)))/R**3

A(5,6)=S(2,J)/R**2

A(5,7)=(S(2,J)**2-2.D0*GM/R)/R**2

A(5,8)=-(S(1,J)*S(2,J))/R**2

A(7,2)=S(8,J)/R

A(7,3)=-(S(2,J)*S(8,J))/R**2

A(7,5)=-1.D0

A(7,8)=S(2,J)/R

A(8,1)=S(8,J)/R

A(8,2)=-(2.D0*S(7,J))/R

A(8,3)=(2.D0*S(2,J)*S(7,J)-S(1,J)*S(8,J)+S(6,J))/R**2

```

```
A(8,6)=-1.DU/R  
A(8,7)=-(2.D0*S(2,J))/R  
A(8,8)=S(1,J)/R  
30 CONTINUE  
DO 40 I=1,N  
P(I)=0.D0  
DO 35 J=1,N  
P(I)=P(I)+A(I,J)*TY(J)  
35 CONTINUE  
40 CONTINUE  
RETURN  
END
```

SUBROUTINE F3

PARAMETER N=8, NQ=4, KUN=4

PARAMETER NQ1=NQ-1

COMMON/CONST/CON(KUN)

DOUBLE PRECISION CON

COMMON/MATRIX/A(NQ,NQ),H(NQ),XF(NQ)

COMMON/DIRV/DEP(N,NQ),YPR(N,4,NQ)

DOUBLE PRECISION A,H,XF,DEP,YPR

COMMON/COUNT/METHOD,KLUC,K1B,K2B,K3B,I2B

C

C THIS ROUTINE DETERMINES THE DISSATISFACTION IN THE TERMINAL

C

C VALUES OF THE DEPENDENT VARIABLE.

C

DO 10 J=1,NQ1

H(J)=DEP(J,1)-XF(J)

10 CONTINUE

H(NQ)=DEP(6,1)-XF(NQ)

RETURN

END

SUBROUTINE F4

PARAMETER N=8, NQ=4, KUN=4

PARAMETER NQ1=NQ-1

COMMON/CONST/CON(KUN)

DOUBLE PRECISION CON

COMMON/MATRIX/A(NQ,NQ),H(NQ),XF(NQ)

COMMON/DIRV/DEP(N,NQ),YPR(N,4,NQ)

DOUBLE PRECISION A,H,XF,DEP,YPR

COMMON/COUNT/METHOD,KLOC,K1B,K2B,K3B,I2B

DOUBLE PRECISION B(NQ,N)

C

C YPR(J,I2B,1) CONTAINS Z-DOT, THE J CORRESPONDS TO THE I-TH H-DOT.

C

C NOW PLACE THE NON-ZERO ELEMENTS OF PARTIAL OF H WITH RESPECT TO Z

C

C IN B.

C

DO 20 J=1,N

DO 10 I=1,NQ

B(I,J)=0.00

10 CONTINUE

20 CONTINUE

B(1,1)=1.00

B(2,2)=1.00

B(3,3)=1.00

B(4,4)=1.00

C

C PLACE THE NQ ELEMENTS OF H=DOT IN A(I,NQ).
C
A(1,NQ)=YPR(1,I2B,1)
A(2,NQ)=YPR(2,I2B,1)
A(3,NQ)=YPR(3,I2B,1)
A(4,NQ)=YPR(6,I2B,1)
DO 60 I=1,NQ
DO 50 J=1,NQ1
A(I,J)=0.00
L=J+1
DO 40 K=1,N
A(I,J)=A(I,J)+B(I,K)*DEP(K,L)
40 CONTINUE
50 CONTINUE
60 CONTINUE
RETURN
END

SUBROUTINE F5
PARAMETER NS3=2
COMMON/PLOTO/BCD(12,NS3)
2 FORMAT(12A6)
C
C READ IN THE LABELS FOR THE X AND Y AXES.
C
READ(5,2)BCD
RETURN
END

```

SUBROUTINE F6

PARAMETER N=8, NQ=4, K0N=4

COMMON/CONST/CON(K0N)

DOUBLE PRECISION CON

DIMENSION PLACE(4)

COMMON/MATRIX/A(NQ,NQ),H(NQ),XF(NQ)

DOUBLE PRECISION A,H,XF

COMMON/INITIAL/DEPO(N),T0,TF,EPS

DOUBLE PRECISION DEPO,T0,TF,EPS

COMMON/NUM/STEP,KEY,KK,PLT,IPLT,IPRO

DOUBLE PRECISION STEP

COMMON/COUNT/METHOD,I1B,I2B,I3B

COMMON/NEWTON/C,DEL

DOUBLE PRECISION C,DEL

COMMON/LEWALN/HNN,HMAX,NTHRU

DOUBLE PRECISION HNN,HMAX

COMMON/KPCH1/IPCH,IHMAX

COMMON/FLAG/FL1,FL2,IFL

DOUBLE PRECISION FL1,FL2

COMMON/PRNT/SWCH

INTEGER SWCH,PLT

10 FORMAT(1H1,36X,32HMETHOD OF PERTURBATION FUNCTIONS ,//,
        1           37X,32HTWO-POINT BOUNDARY VALUE PROBLEM ,//)

20 FORMAT(1HU,32X,40HINITIAL VALUE OF THE DEPENDENT VARIABLES ,//)

25 FORMAT(20X,13,3X,D25.10,4X,13,3X,D25.16)

30 FORMAT(1HU,32X,40HDESIRED VALUES OF THE TERMINAL VARIABLES ,//)

40 FORMAT(1HU,39X,25HTHE INITIAL TIME INTERVAL ,/,20X,6HFROM ,/

```

1 D25.16,I0H / TO ,D25.16)

50 FORMAT(1H0,25X,29HTHE INTEGRATION STEP SIZE IS ,D25.16)

60 FORMAT(1H0,19X,61HTHE NUMBER OF ITERATIONS WITH THE ADAMS-MUULTON
1CORRECTOR IS ,I5)

80 FORMAT(1H0,27X,44HTHE MAXIMUM NUMBER OF ITERATIONS ALLOWED IS ,I5)

90 FORMAT(1H0,19X,41HTHE ACCURACY REQUIRED FOR TERMINATION IS ,
1 D25.16)

100 FORMAT(1H0,22X,5HEVERY,I5,33HTH POINT WILL BE PLOTTED FOR EACH ,
1 I5,I2HITH ITERATION)

110 FORMAT(1H0,26X,5HEVERY,I5,42HTH POINT WILL BE PRINTED ON EACH ITER
1 ATION)

120 FORMAT(1H0,39X,23HPRINT CONTROL SWITCH = ,I2)

125 FORMAT(1H0,39X,23HPUNCH CONTROL SWITCH = ,I2)

130 FORMAT(1H0,26X,I2,50H SPECIAL INPUT CONSTANTS AND THEIR IDENTIFICA
1 TIONS //)

135 FORMAT(23X,4A6,11X,D25.16)

140 FORMAT(1H0,38X,27HNORMAL CORRECTION PROCEDURE)

150 FORMAT(1H0,37X,31HFRACTIONAL CORRECTION PROCEDURE //,
1 20X,4HC = ,D25.16,6X,6HDEL = ,D25.16)

160 FORMAT(1H0,36X,33HMINIMUM NORM CORRECTION PRCDURE)

161 FORMAT(1H0,40X,23HSTEPPED ALPHA PROCEDURE //,20X,7HALPHA = ,D25.16,
1 9H GAMMA = ,D25.16)

162 FORMAT(1H0,41X,22HSTEPPED BETA PROCEDURE //,20X,7HBETA = ,D25.16,
1 9H GAMMA = ,D25.16)

163 FORMAT(1H0,40X,24HVARIABLE ALPHA PROCEDURE //,20X,7HALPHA = ,
1 D25.16,6X,3HP = ,D25.16)

164 FORMAT(1H0,40X,23HVARIABLE BETA PROCEDURE //,20X,7HBETA = ,D25.16,
1 5X,4HP = ,D25.16)

165 FORMAT(1H1)

```
170 FORMAT(1HU,23X,34HTHE MAXIMUM NORM IS ASSUMED TO BE ,D25.16)
210 FORMAT(2D25.16)
220 FORMAT(14I5)
230 FORMAT(4A6,1X,D25.16)
240 FORMAT(2D25.16,4X,I1)

C
C      THE FORMAT STATEMENTS UNDER 200 ARE FOR OUTPUT
C
C      THE FORMAT STATEMENTS OVER 200 ARE FOR INPUT
C

      WRITE(6,10)
      WRITE(6,20)
      READ(5,210)DEPO
      WRITE(6,25)(J,DEPO(J),J=1,N)
      WRITE(6,30)
      READ(5,210)XF
      WRITE(6,25)(J,XF(J),J=1,NQ)
      READ(5,210)T0,TF
      WRITE(6,40)T0,TF
      READ(5,210)STEP,EPS
      WRITE(6,50)STEP
      READ(5,220)METHOD,KK,1PLT,PLT,IPRO,KEY,SWCH,1PCH,IHMAX
      IF(SWCH.NE.0) SWCH=1
      WRITE(6,80)KK
      WRITE(6,90)EPS
      WRITE(6,60)METHOD
      WRITE(6,100)1PLT,PLT
      WRITE(6,110)IPRO
      WRITE(6,120)SWCH
```

WRITE(6,125)IPCH
GO TO(310,320,330),KEY

310 WRITE(6,140)
GO TO 340

320 READ(5,210)C,DEL
WRITE(6,150)C,DEL
GO TO 340

330 WRITE(6,160)
READ(5,240)FL1,FL2,IFL
GO TO(331,332,333,334),IFL

331 WRITE(6,161)FL1,FL2
GO TO 340

332 WRITE(6,162)FL1,FL2
GO TO 340

333 WRITE(6,163)FL1,FL2
GO TO 340

334 WRITE(6,164)FL1,FL2

340 CONTINUE
IF(IHMAX.EQ.0) GO TO 345
READ(5,210)HMAX
WRITE(6,170)HMAX

345 CONTINUE
NOK=KUN
IF(NOK.LE.0)GO TO 400
WRITE(6,130)NOK
DO 350 I=1,KUN
READ(5,230)PLACE,CUN(1)
WRITE(6,135)PLACE,CUN(1)

350 CONTINUE

```
400 IF(SWCH.EQ.0)WRITE(6,165)
      RETURN
      END
```

```

SUBROUTINE F7

PARAMETER N=8 , NQ=4

PARAMETER NP=N-NQ

COMMON/DIRV/DEP(N,NQ),TPR(N+4,NQ)

DOUBLE PRECISION DEP,TPR

COMMON/INITAL/DEPO(N),IO,TF,EPS

DOUBLE PRECISION DEPO,IO,TF,EPS

COMMON/MODE/S(N,3),INN(5)

DOUBLE PRECISION S

COMMON/EVAL/VIND,DELT,INDI

DOUBLE PRECISION VIND,DELT

C
C      SET UP THE INITIAL CONDITIONS FOR INTEGRATION
C
VIND=10

1002      DO 25 J = 1,N

          DEP(J,1) = DEPO(J)
          S(J,1) = DEPO(J)

          DO 25 JAY=2,NQ

          DEP(J,JAY) = 0.000
CONTINUE

25      CONTINUE

DO 30 I=2,NQ

          J=NP+I

          DEP(J,1)=1.000
CONTINUE

30      CONTINUE

RETURN

END

```

SUBROUTINE INTEG
PARAMETER N=8
COMMON/MIDTR/J,JKK
COMMON/NUM/STEP,KEY,KK,PLT,IPLT,IPRO
DOUBLE PRECISION STEP
INTEGER PLT
COMMON/COUNT/METHOD,ILUC,I1B,I2B,I3B,K2B
COMMON/LOGIC/BOOL
LOGICAL BOOL
COMMON/EVAL/VIND,DELT,INDI
DOUBLE PRECISION VIND,DELT
COMMON/MODE/S(N,3),JAT,NU,IONE,ITWO,IERRCT
DOUBLE PRECISION S
COMMON/INITAL/DEPO(N),I1,T2,EPS
DOUBLE PRECISION DEPO,I1,T2,EPS
DOUBLE PRECISION XN,ERR
DATA ERR/3.0/,IERR/3/,JERR/9/
C
C MAIN INTEGRATION ROUTINE. THIS ROUTINE USES A RUNGE-KUTTA STARTER
C
C AND AN ADAMS PREDICTOR-CORRECTOR. THE LAST STEP IS ALSO DONE WITH
C
C A RUNGE-KUTTA.
C
DELT = STEP/ERR
BOOL = ,FALSE.
IKEY = 1

LL

INDI = 1
I48 = 1
IONE = 1
ITWO = 2
IERRCT = 1
ILOC = 1
I1B = 2
I2B = 3
I3B = 4
VIND = T1
LTHL = JERR

5 DO 35 JKK = I48,LTHL
K2B = ILOC
IF(JKK.NE.4 .AND. JKK.NE.7) GO TO 10
IF(JKK.NE.4 .AND. JKK.NE.7) GO TO 15
JKL = JKK/IERR
GO TO 11

10 CONTINUE
JKL = JKK - 1

11 CONTINUE
J = MOD(JKL,IPLT)

ITT=9
CALL FPLEW(ITT)
IF(IPRO0.LE.0) GO TO 15
J = MOD(JKL,IPRO0)
CALL FPRNT(ITT)

15 IF(IKEY.EQ.2) GO TO 20
CALL INTRK5
IF(BOOL) GO TO 40 A-52

```
IF(JKK.EQ.LTHL) GO TO 30  
IF(IERRCT.EQ.IERR) GO TO 25  
IERRCT = IERRCT + 1  
GO TO 35  
20 CALL ABAM  
XN = T2 - VIND  
IF(XN.LE.0.D0) GO TO 45  
IF(XN.GE.DELT) GO TO 35  
DELT = XN  
BOOL = .TRUE.  
IKEY = 1  
GO TO 35  
25 IERRCT = 1  
I2B = I2B - 1  
GO TO 35  
30 IKEY = 2  
DELT = STEP  
I48 = 4  
I2B = 3  
LTHL = (T2 - VIND)/DELT + 4.0D0*ERR  
GO TO 5  
35 CONTINUE  
40 CONTINUE  
K2B = I2B  
45 CONTINUE  
RETURN  
END
```

SUBROUTINE INTRK5

PARAMETER N=8, NQ=4

COMMON/EVAL/VIND,DELT,INDI

DOUBLE PRECISION VIND,DELT

COMMON/DIFEQ/P(N),DEPVAR(N),XN

DOUBLE PRECISION P,DEPVAR,XN

COMMON/DIRV/DEP(N,NQ),YPR(N+4,NQ)

DOUBLE PRECISION DEP,YPR

COMMON/MODE/S(N+3),JAY,NU,IONE,ITWO,IERRCT

DOUBLE PRECISION S

DOUBLE PRECISION XIP(N)

COMMON/LOGIC/BOOL

LOGICAL BOOL

COMMON/COUNT/METHOD,ILOC,I1B,I2B,I3B,K2B

DOUBLE PRECISION A(4), P1(4)

DATA (A(J),J=1,4)/0.16666666666667D0, 0.33333333333333D0,

+ 33333333333333D0, .16666666666667D0/

DATA (P1(J),J=1,4)/0.0D0,0.5D0,0.5D0,1.0D0/

DATA IERR/3/

C

C THIS ROUTINE PERFORMS A RUNGE-KUTTA INTEGRATION

C

NS=NQ

DO 90 JAY = 1,NS

XN = VIND

ISAV = INDI

DO 10 J = 1,N

XIP(J)=0.0D0

10 DEPVAR(J) = DEP(J,JAY)
NU = 1

15 CONTINUE

IF(JAY.NE.1) GO TO 17

CALL F1

GO TO 18

17 CONTINUE

CALL F2

18 CONTINUE

IF(ISAV.EQ.2) GO TO 30

ISAV = 2

DO 20 J = 1,N

20 YPR(J,4,JAY) = P(J)

30 DO 35 J = 1,N

35 XIP(J) = A(NU)*P(J) + XIP(J)

IF(NU = 4) 40, 55, 55

40 NU = NU + 1

XN = VIND + P1(NU)*DELT

DO 50 J = 1,N

50 DEPVAR(J) = DEP(J,JAY) + P1(NU)*DELT*P(J)

GO TO 15

55 DO 60 J = 1,N

DEP(J,JAY) = DEP(J,JAY) + DELT*XIP(J)

60 DEPVAR(J) = DEP(J,JAY)

IF(JAY.NE.1) GO TO 70

CALL F1

DO 65 J = 1,N

65 S(J,ITWO) = DEPVAR(J)
A-55

GO TO 75

70 CONTINUE

CALL F2

IF(BOOL) GO TO 76

75 IF(IERRCT,NE,IERR) GO TO 90

76 DO 80 J = 1,N

80 YPR(J,I2B,JAY) = P(J)

90 CONTINUE

INDI = 2

VIND = XN

ISAV = IONE

IONE = ITWO

ITWO = ISAV

RETURN

END

```
SUBROUTINE ITER

PARAMETER N=8 , NQ=4

PARAMETER NQ1=NQ-1, NP=N-NQ+1

COMMON/MATRIX/A(NQ,NQ),H(NQ),XF(NQ)

DOUBLE PRECISION A,H,XF

COMMON/INITAL/DEPO(N),T0,TF,EPS

DOUBLE PRECISION DEPO,T0,TF,EPS

C
C      THIS ROUTINE ADDS THE CORRECTION ON THE INITIAL DEPENDENT VARIABLE
C
C      AND FINAL TIME.

C
      TF = TF + H(NQ)

      DO 30 I=1,NQ1
          J = NP + I
          DEPO(J) = DEPO(J) + H(I)
30      CONTINUE
      RETURN
      END
```

SUBROUTINE MINVDP(A,N,K,X,J2,I2)

C* MATRIX INVERSION ROUTINE-FORMULATED BY E. G. CLAYTON

C** --- CALLING SEQUENCE ---

C*** CALL MINVDP(A,N,E,K)

C**** A--SQUARE ARRAY (DOUBLE PRECISION) CONTAINING ORIGINAL MATRIX

C**** N--ORDER OF ORIGINAL MATRIX

C**** E--TEST CRITERION FOR NEAR ZERO DIVISOR (DOUBLE PRECISION)

C*** K--LOCATION FOR SINGULARITY OR ILL-CONDITION INDICATOR

C** K=0 => MATRIX NONSINGULAR.

C* K=1 => MATRIX SINGULAR (OR ILL-CONDITIONED)

DOUBLE PRECISION A,X,BIGA,DIV ,E

DIMENSION A(N,N),X(N),J2(N),I2(N)

DATA E/.1D-17/

C* INITIALIZATION

M=N

K=0

I2(1)=0

J2(1)=0

C* BEGIN COMPUTATION OF THE INVERSE

DO 15 L=1,M

L1=L-1

BIGA=0.0D0

C* LOOK FOR THE ELEMENT OF GREATEST ABSOLUTE VALUE,CHOOSING

C* ONE FROM A ROW AND COLUMN NOT PREVIOUSLY USED.

DO 5 I=1,M

DO 1 I3=1,L1

IF(I-I2(I3))1,5,1

```

1 CONTINUE
DO 4 J=1,M
DO 2 I3=1,L1
IF(J-J2(I3))2,4,2
2 CONTINUE
IF(BIGA=DABS(A(I,J)))3,3,4
3 BIGA=DABS(A(I,J))
J1=J
I1=I
4 CONTINUE
5 CONTINUE
C*      TAG THE ROW AND COLUMN FROM WHICH THE ELEMENT IS CHOSEN.
J2(L)=J1
I2(L)=I1
DIV=A(I1,J1)
C*      TEST ELEMENT AGAINST ZERO CRITERION.
IF(DABS(DIV)-E)221,221,6
C*      PERFORM THE COMPUTATIONS
6 DO 7 J=1,M
A(I1,J)=A(I1,J)/DIV
7 CONTINUE
A(I1,J1)=1.0D0/DIV
DO 11 I=1,M
IF(I1-I)8,11,8
8 DO 10 J=1,M
IF(J1-J)9,10,9
9 A(I,J)=A(I,J)-A(I1,J)*A(I,J1)
10 CONTINUE
11 CONTINUE

```

```
DO 14 I=1,M  
IF(I1-I) 13,14,13  
13 A(I,J1)=-A(I,J1)*A(I1,J1)  
14 CONTINUE  
15 CONTINUE  
C* COMPUTATION COMPLETE AT THIS POINT  
C* UNSCRAMBLE THE INVERSE  
DO 18 J=1,M  
DO 16 I=1,M  
I1=I2(I)  
J1=J2(I)  
X(J1)=A(I1,J)  
16 CONTINUE  
DO 17 I=1,M  
A(I,J)=X(I)  
17 CONTINUE  
18 CONTINUE  
DO 21 I=1,M  
DO 19 J=1,M  
I1=I2(J)  
J1=J2(J)  
X(I1)=A(I,J1)  
19 CONTINUE  
DO 20 J=1,M  
A(I,J)=X(J)  
20 CONTINUE  
21 CONTINUE  
RETURN  
221 K=1
```

RETURN

END

SUBROUTINE MLTPLY

PARAMETER NQ=4

COMMON/FLAG/FL1,FL2,IFL

DOUBLE PRECISION FL1,FL2

COMMON/LEWALN/HNN,HMAX,NTHRU

DOUBLE PRECISION HNN,HMAX

COMMON/MATRIX/A(NQ,NQ),C(NQ),XF(NQ)

DOUBLE PRECISION A,C,XF

COMMON/TRL/D(NQ),ISET

DOUBLE PRECISION D

DOUBLE PRECISION B(NQ,NQ),BETA,BMAX,ALPHA

DATA BETA/0.1D-03/

DO 10 I=1,NQ

DO 10 J=1,NQ

B(I,J)=0.0D0

DO 10 K=1,NQ

10 B(I,J) = B(I,J) + A(K,I)*A(K,J)

DO 35 I=1,NQ

IF (B(I,I) .GT. BETA) GO TO 35

BMAX=0.0D0

DO 30 L=1,NQ

IF (B(L,L) .GT. BMAX) BMAX = B(L,L)

30 CONTINUE

B(1,1) = BMAX

35 CONTINUE

GO TO (50,80,150,150),IFL

```

50 DO 60 I=1,NQ
60 B(I,I) = B(I,I) + FL1*B(I,I)
GO TO 300

C STEPPED-BETA TECHNIQUE

80 DO 85 I=1,NQ
85 B(I,I) = B(I,I) + FL1
GO TO 300

C VARIABLE-ALPHA TECHNIQUE

150 ALPHA = FL1 * (HNN/HMAX)**FL2
IF (ALPHA .LT. 0.000) ALPHA = 0.000
IF (ALPHA .GT. FL1) ALPHA = FL1
IF (IFL .EQ. 4) GO TO 180
DO 160 I=1,NQ
160 B(I,I) = B(I,I) + ALPHA*B(I,I)
GO TO 300

C VARIABLE-BETA TECHNIQUE

180 DO 185 I=1,NQ
185 B(I,I) = B(I,I) + ALPHA

C STORAGE MATRICES

300 DO 310 I=1,NQ
D(I) = 0.0
DO 310 J=1,NQ
310 D(I) = D(I) - A(J,I)*C(J)
DO 330 I=1,NQ
DO 330 J=1,NQ
330 A(I,J) = B(I,J)

C

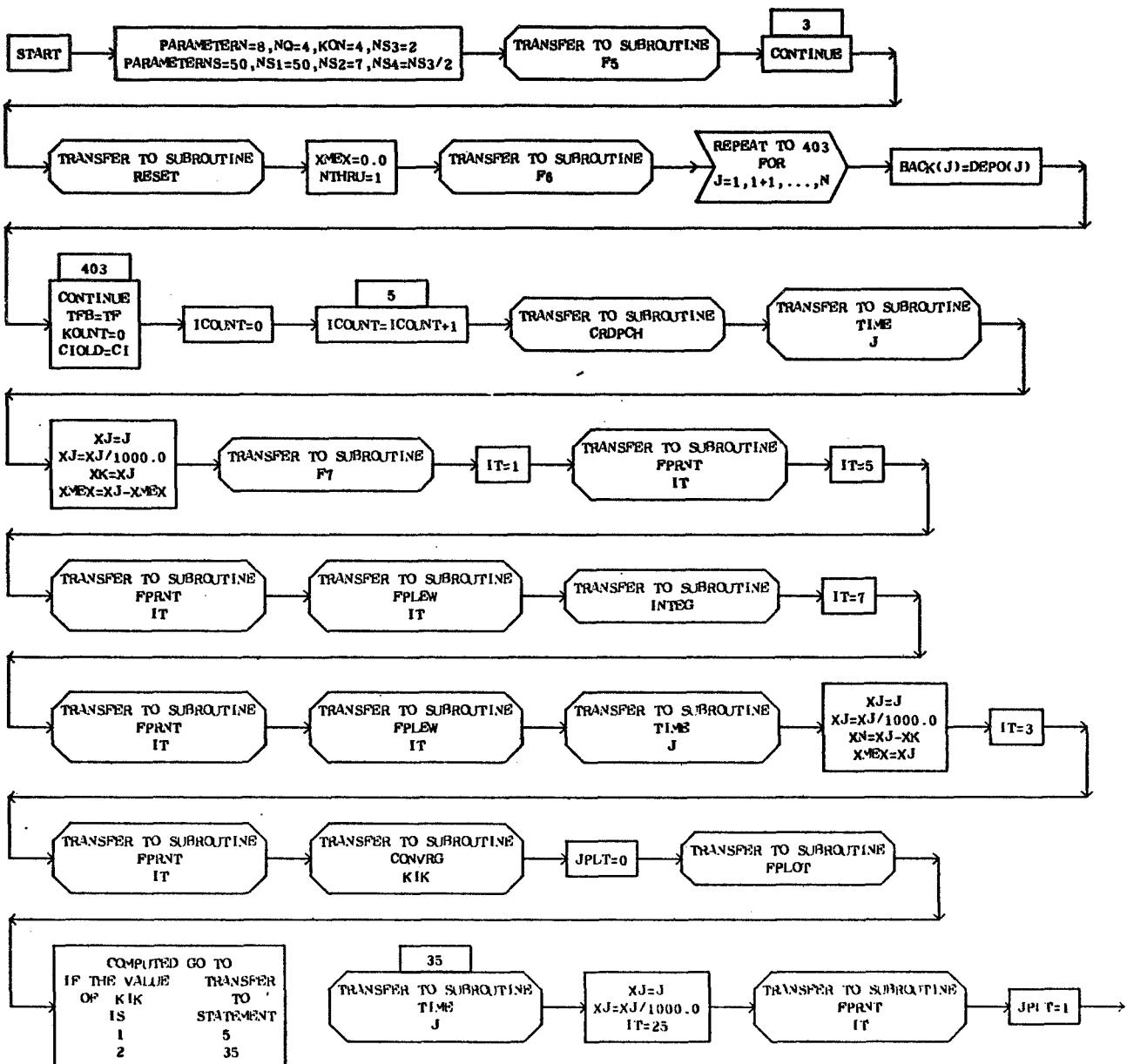
RETURN
END

```

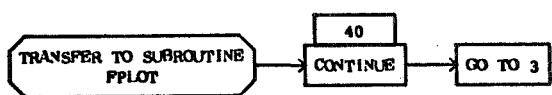
3. FLOW CHARTS

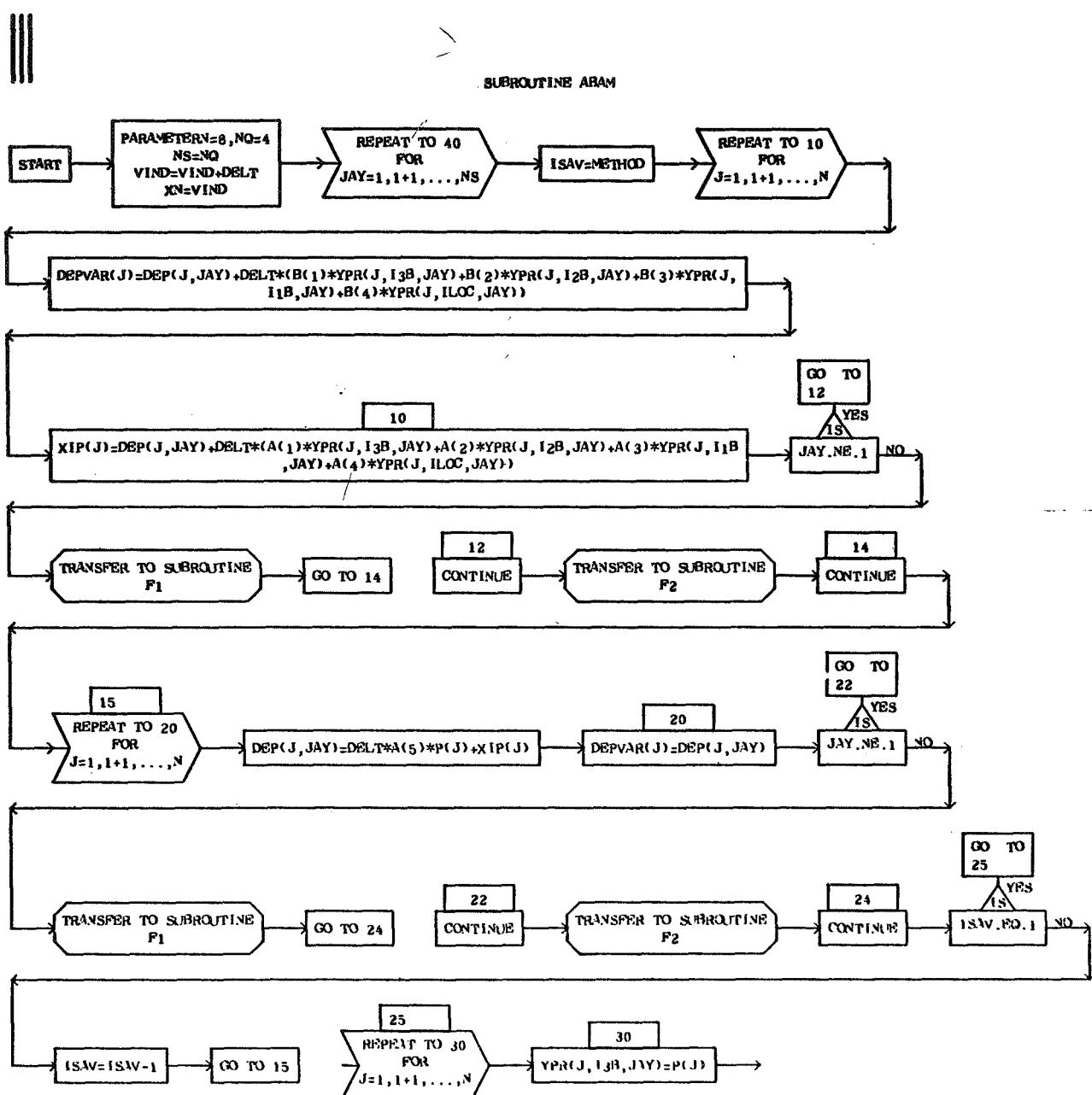


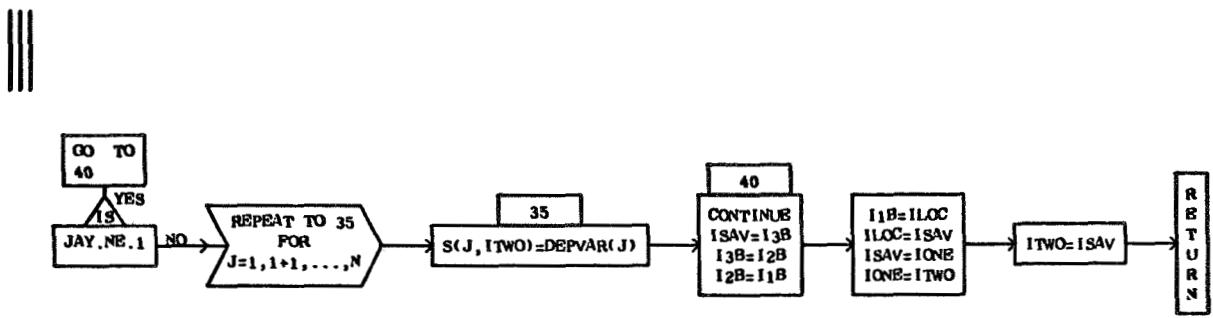
MAIN PROGRAM FOR MPP

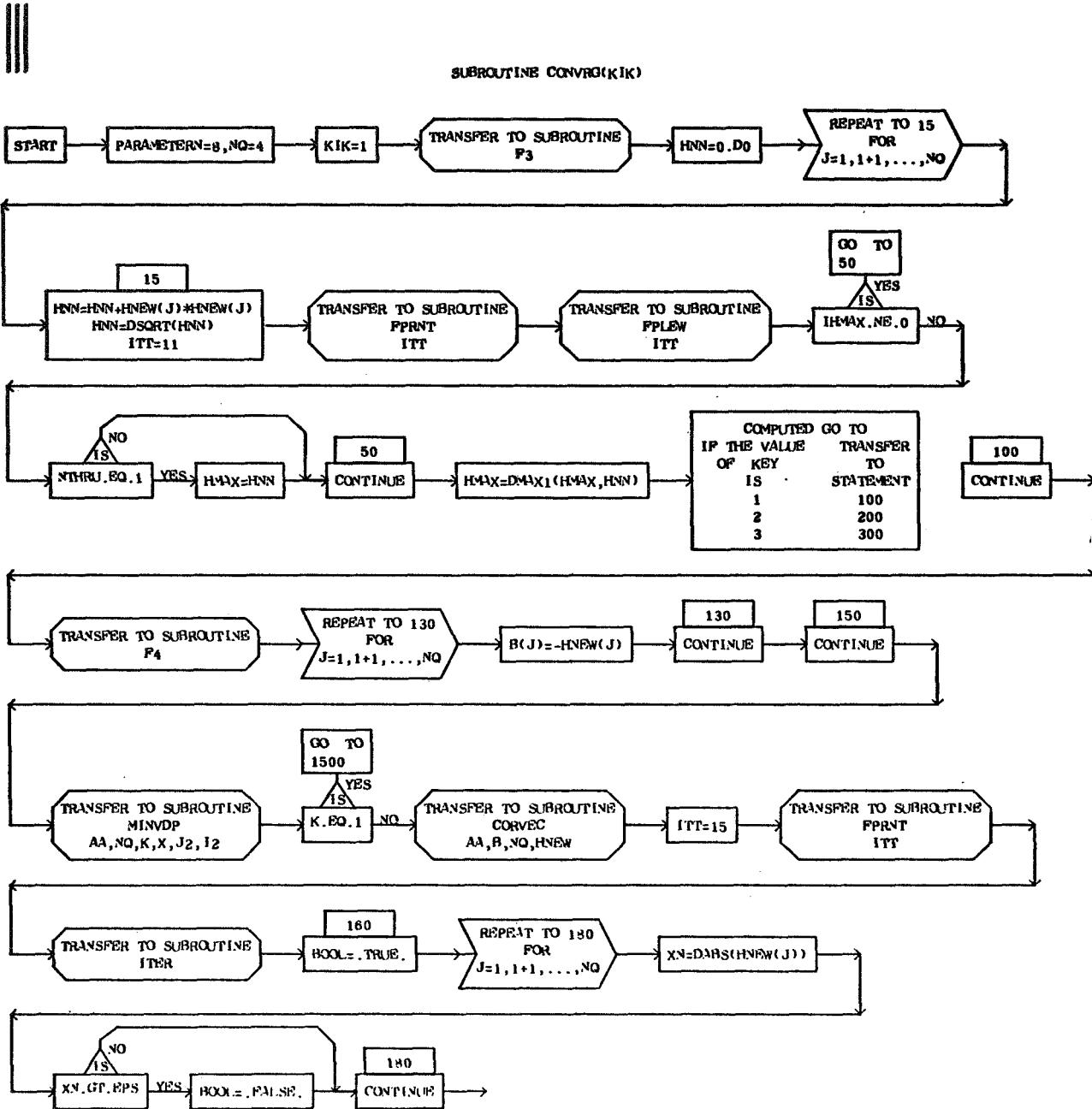


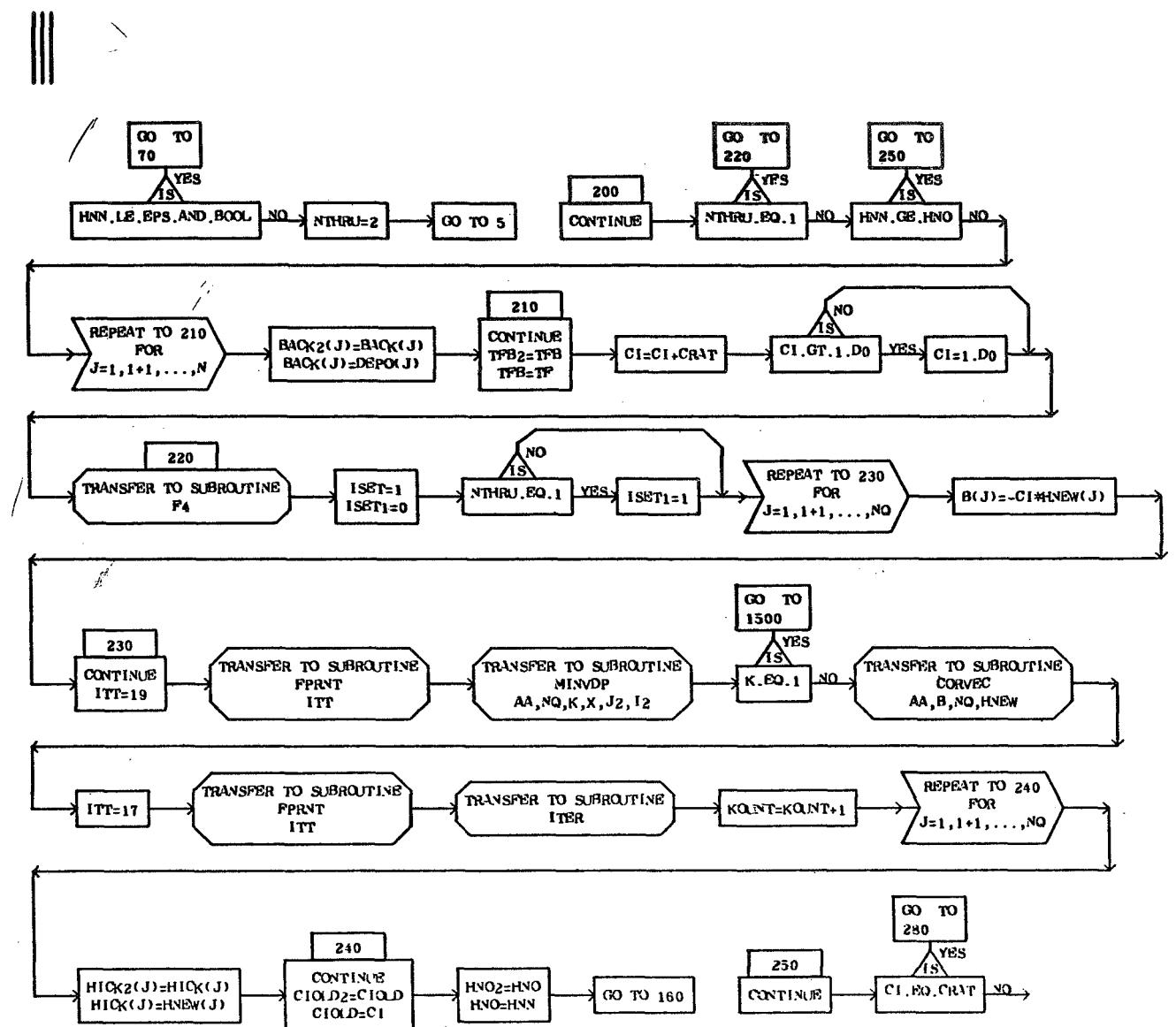
|||

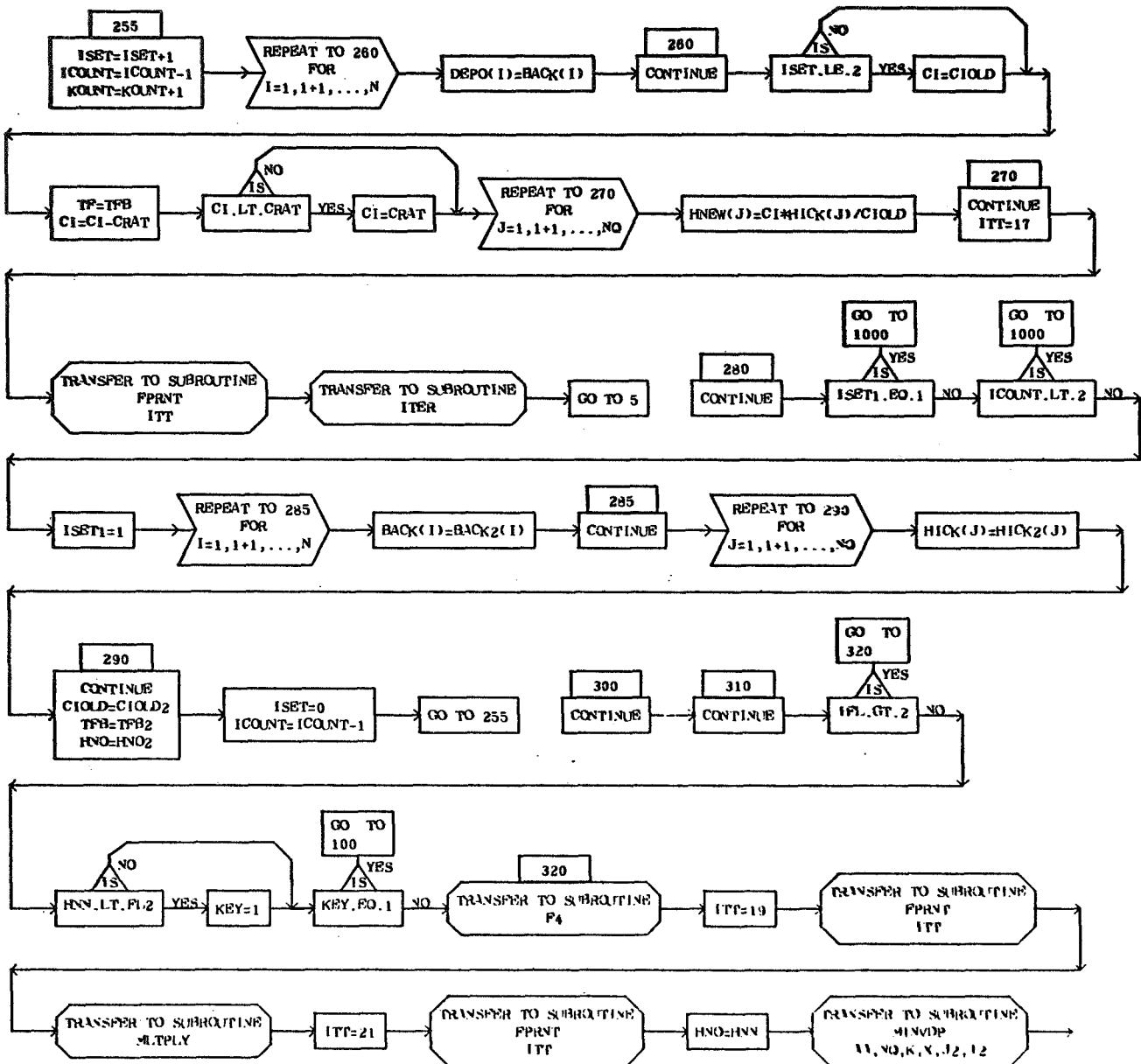




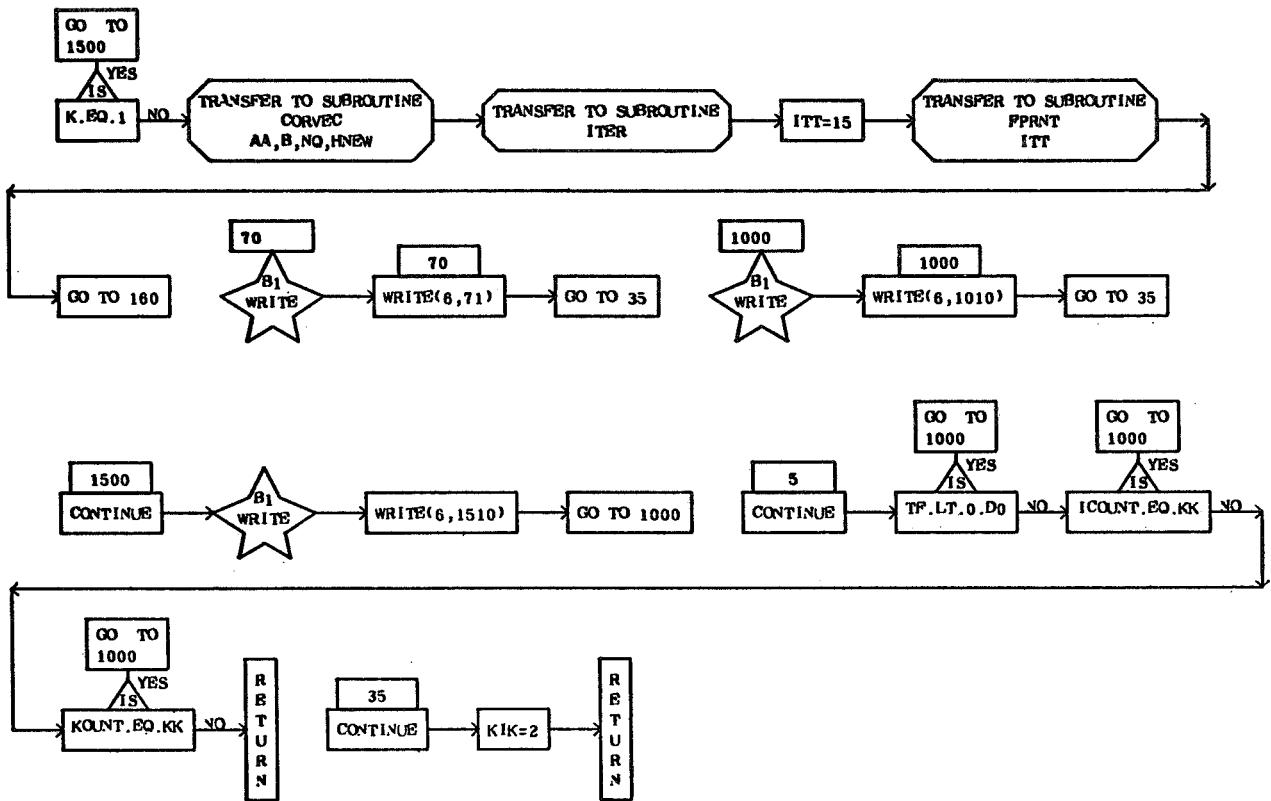






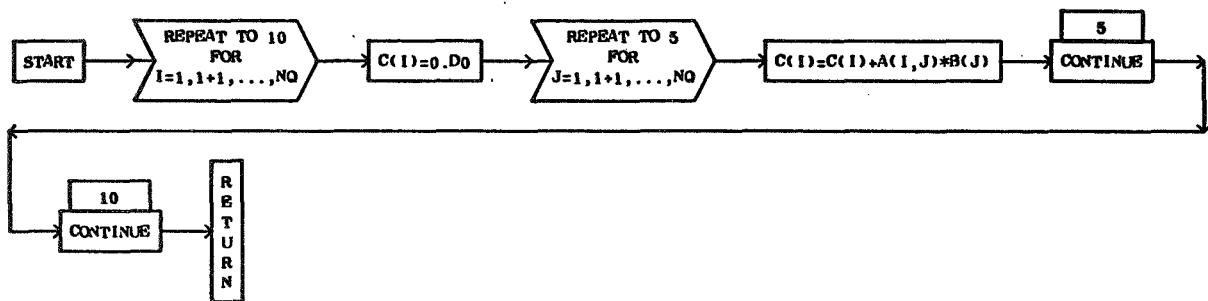


|||

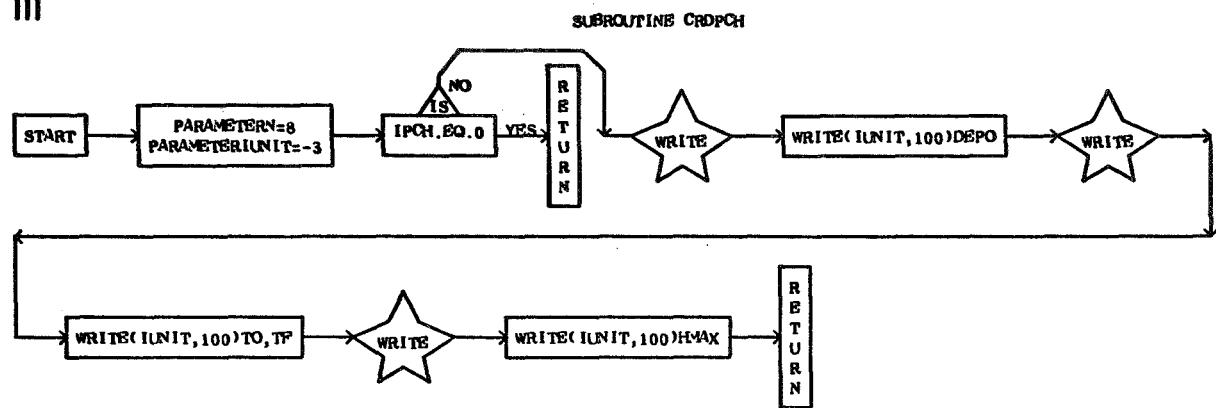


|||

SUBROUTINE CORVEC(A,B,NQ,C)

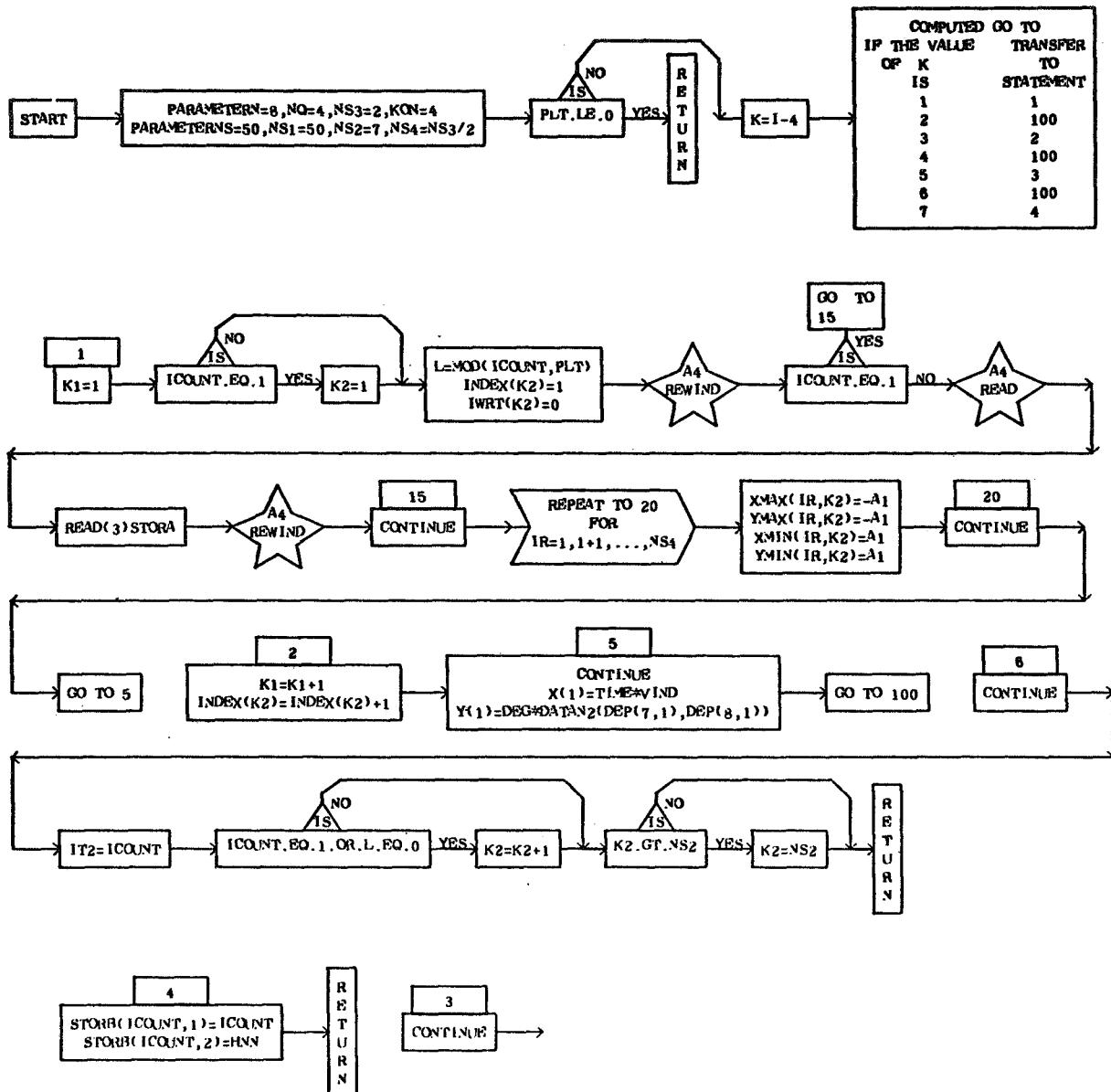


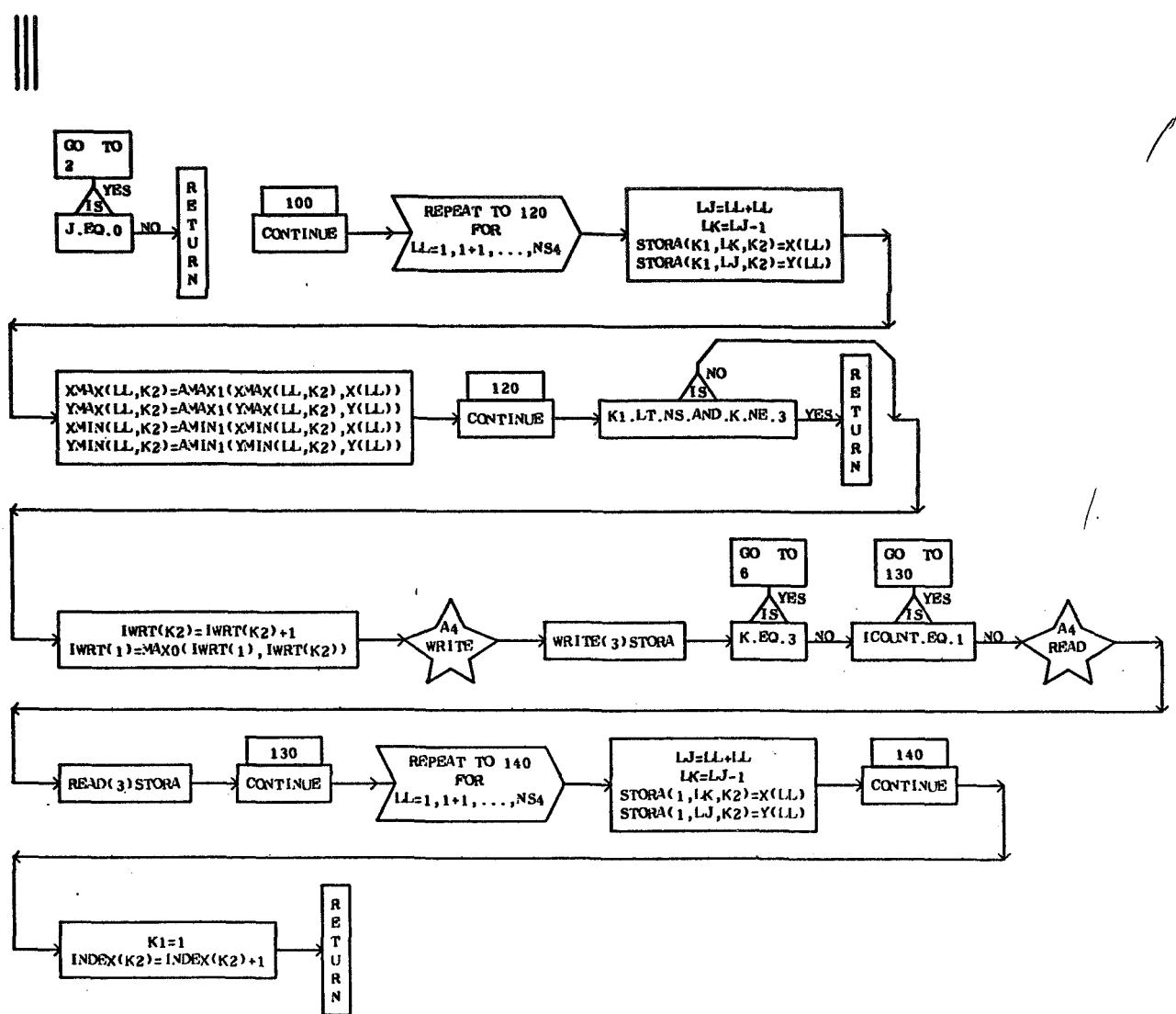
|||

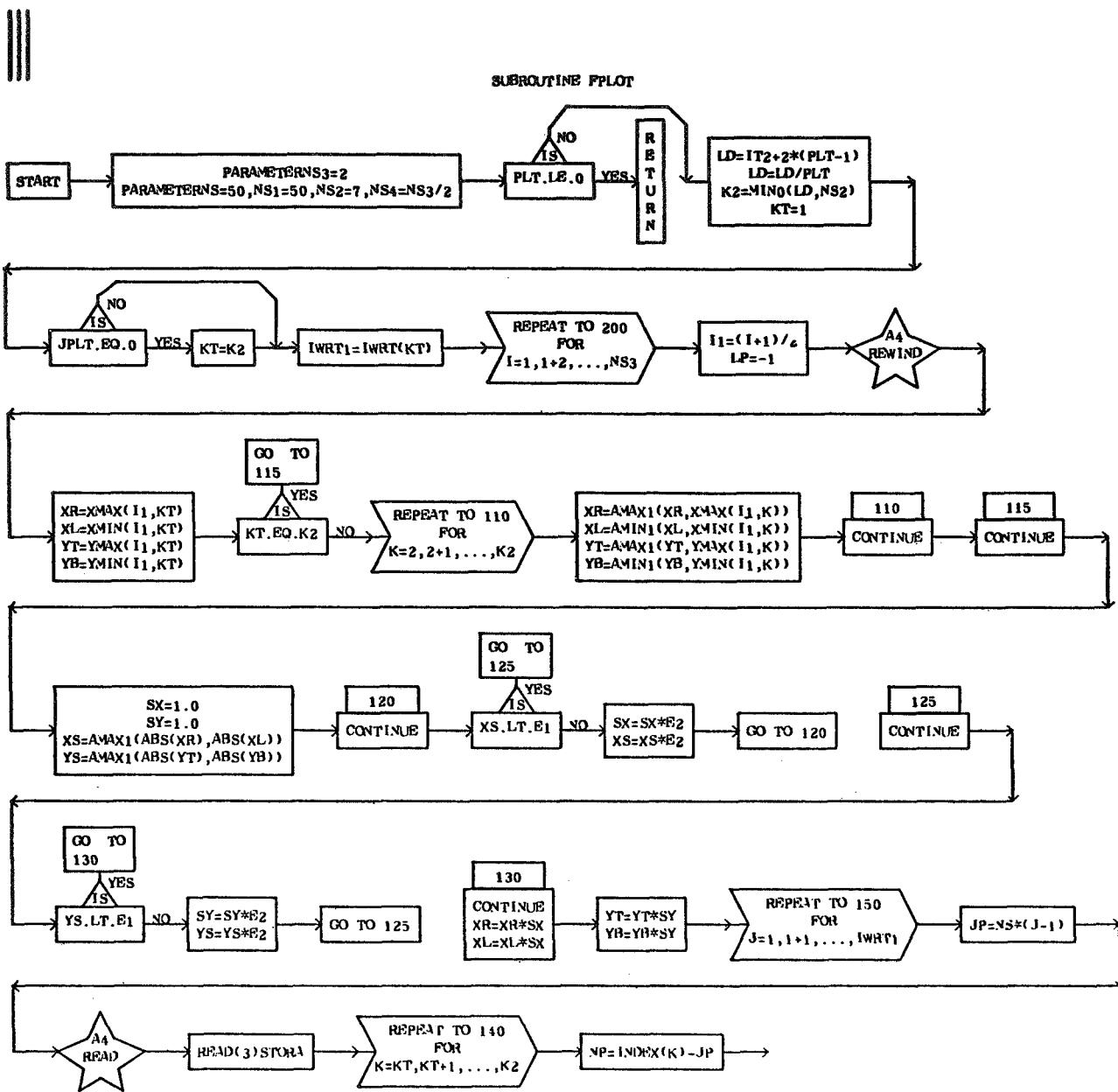


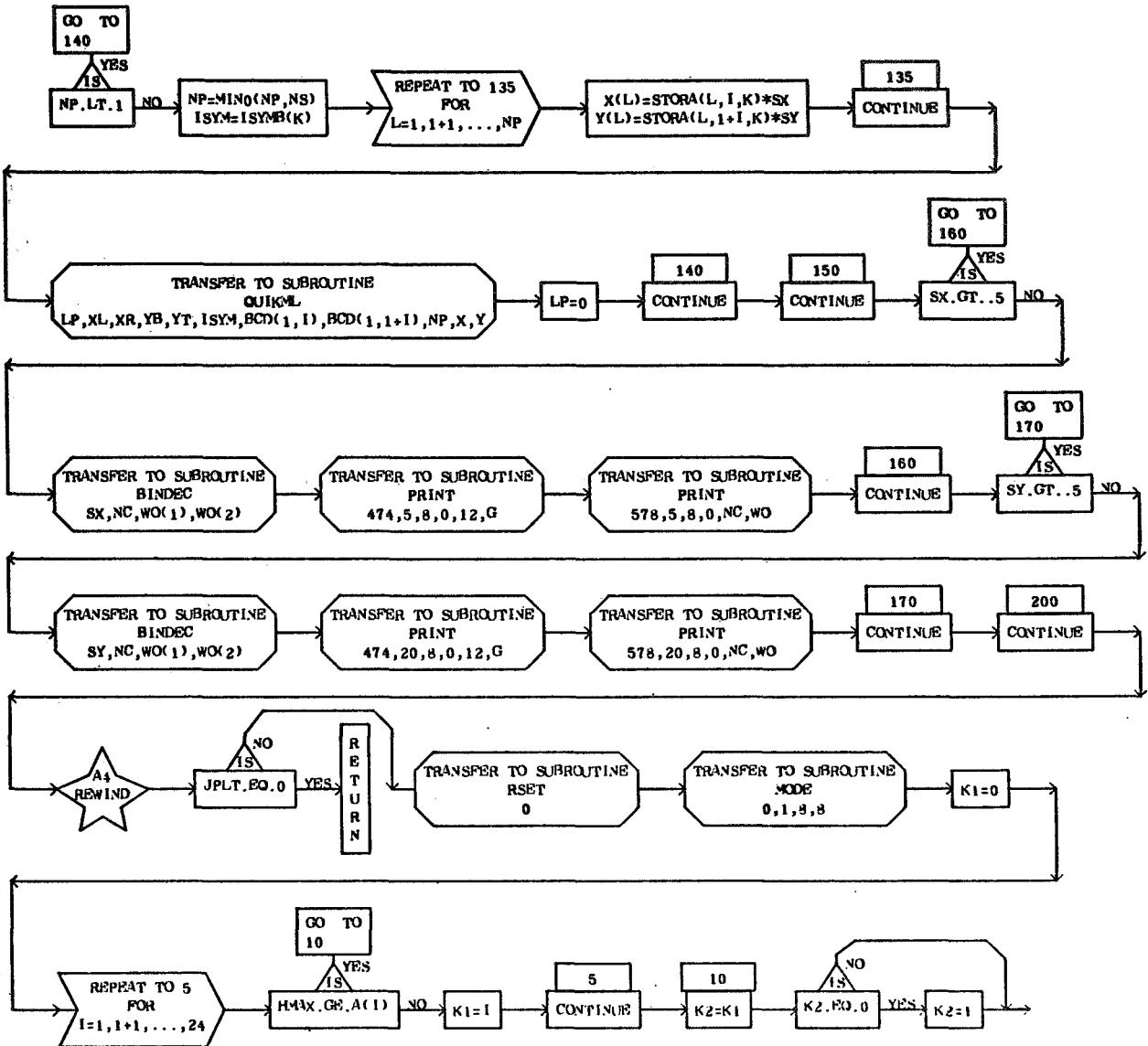
|||

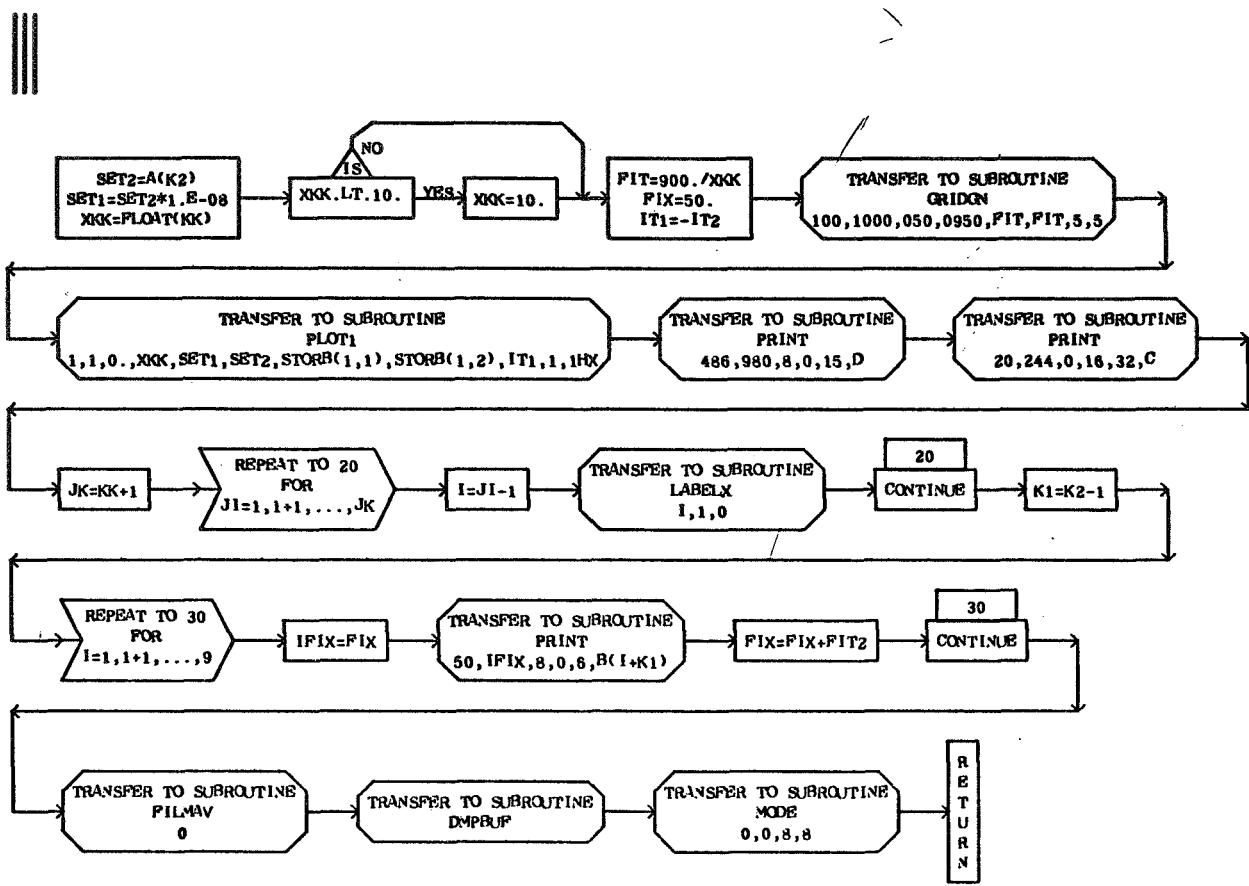
SUBROUTINE PPLEW(1)





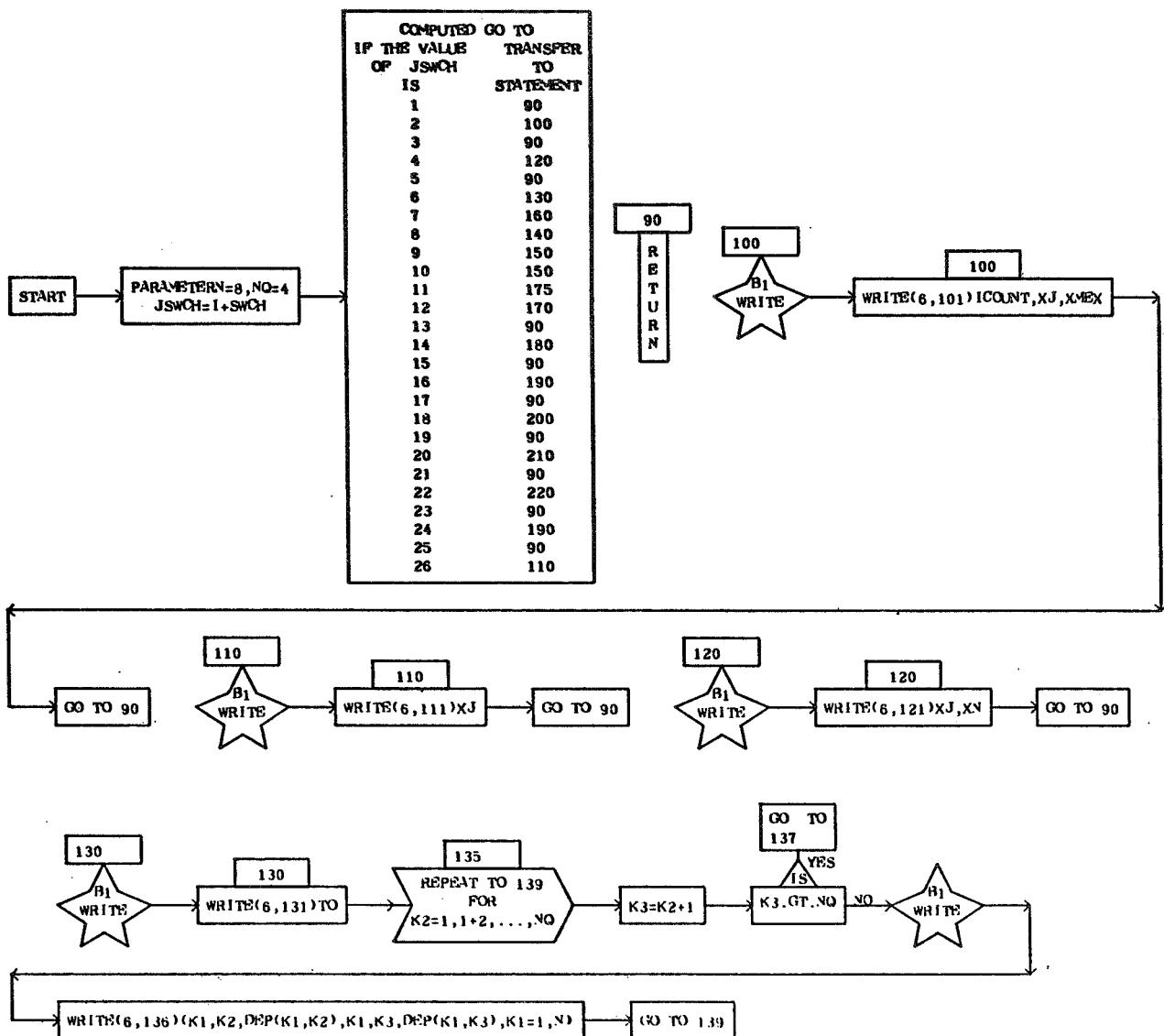


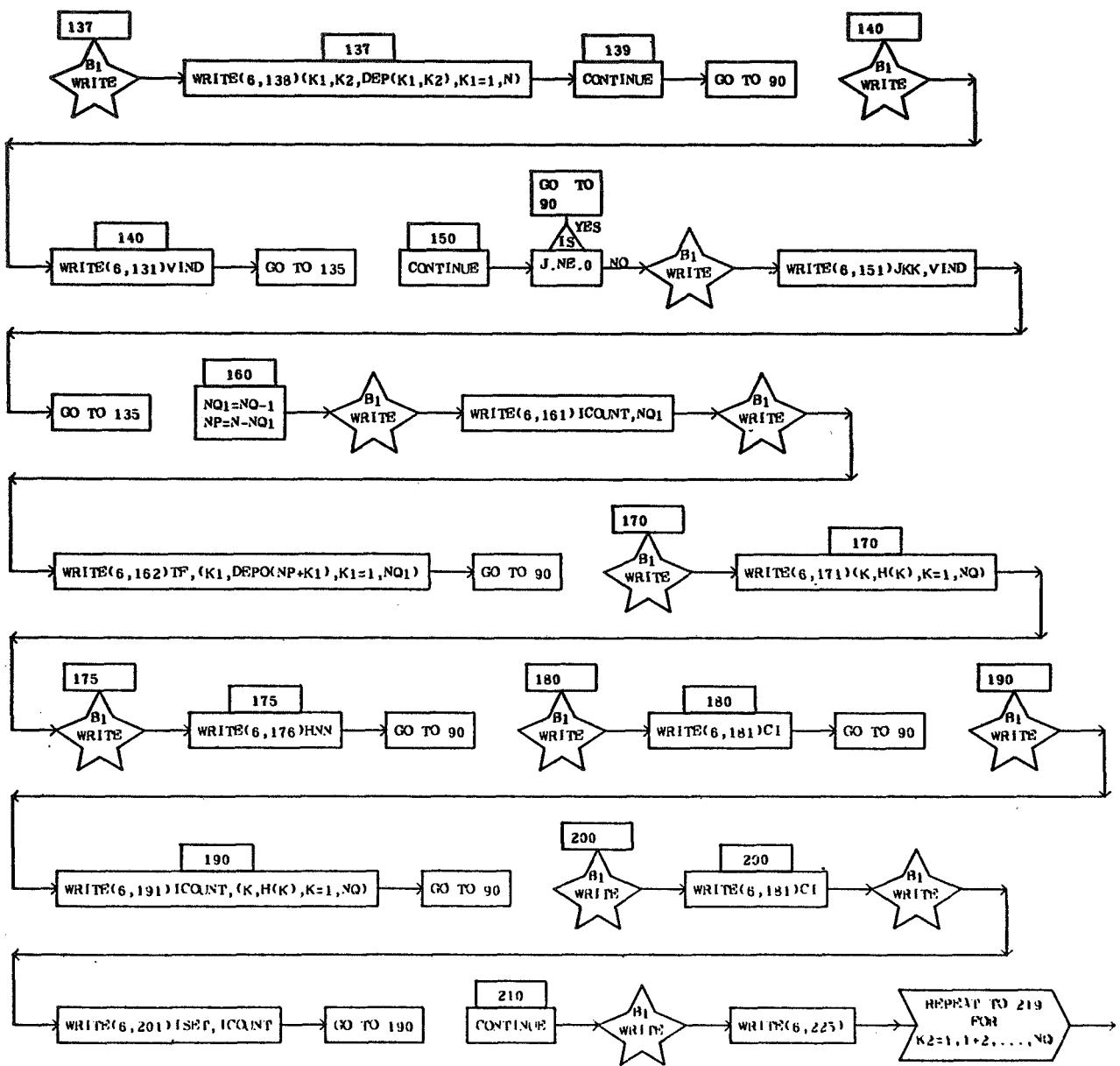


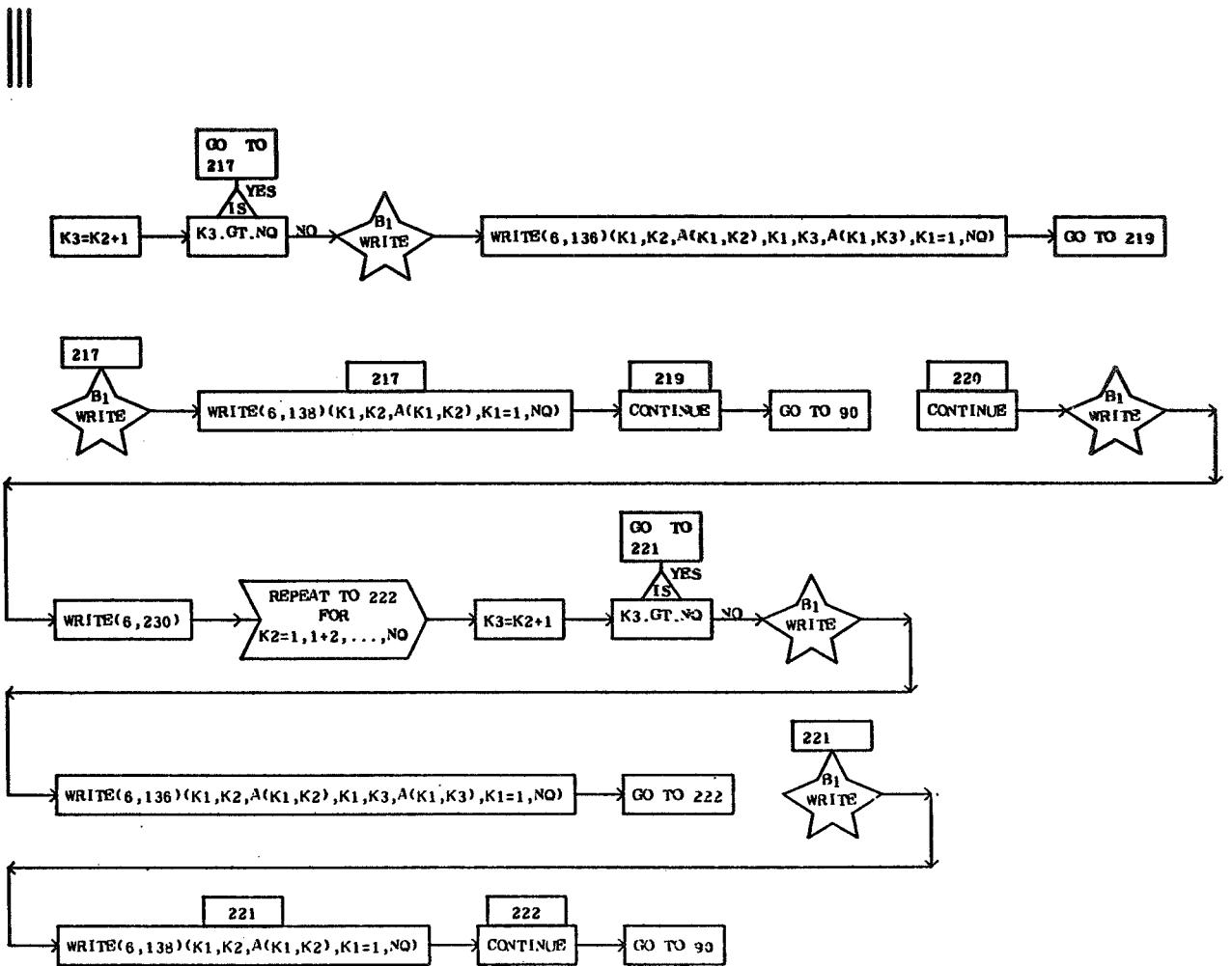


|||

SUBROUTINE FPRNT(1)

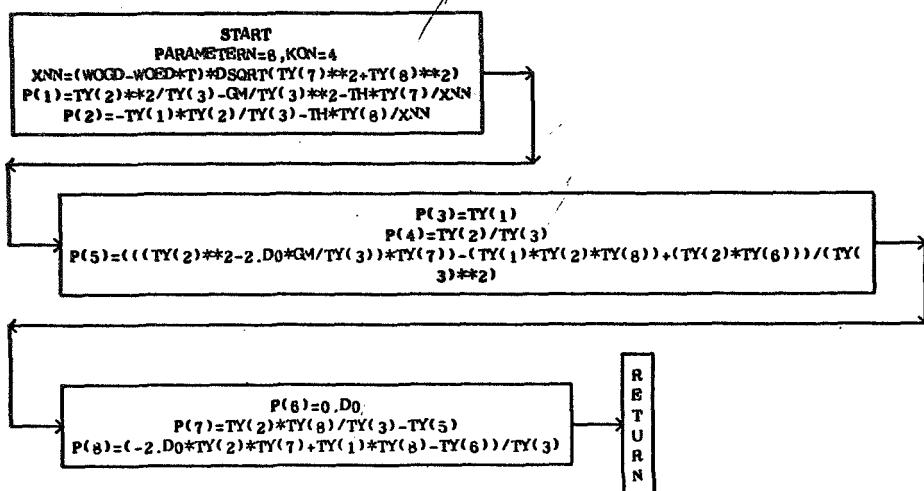






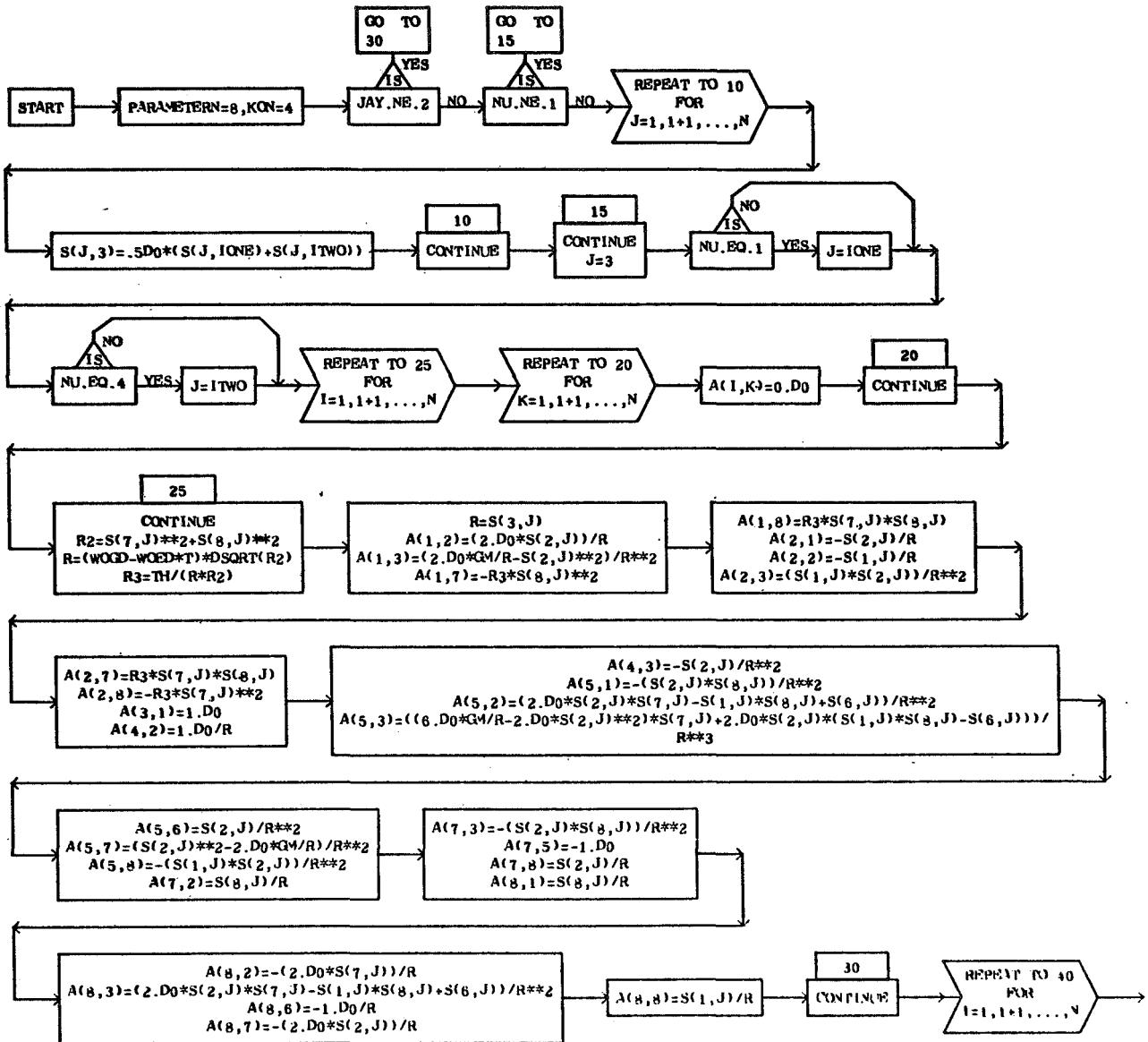
|||

SUBROUTINE F1

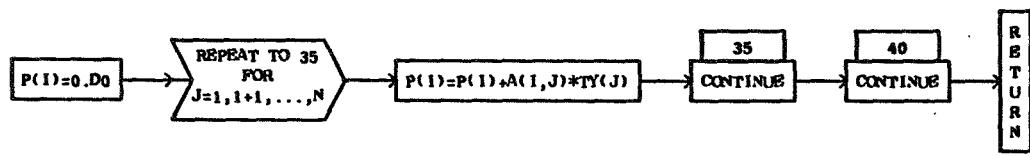




SUBROUTINE F2

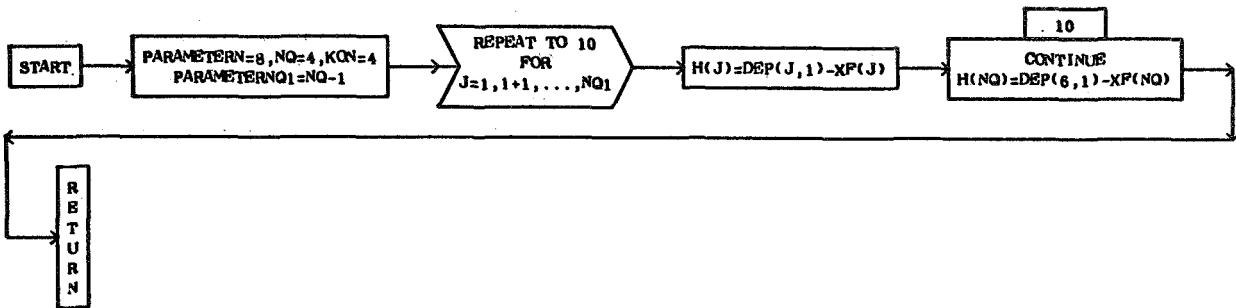


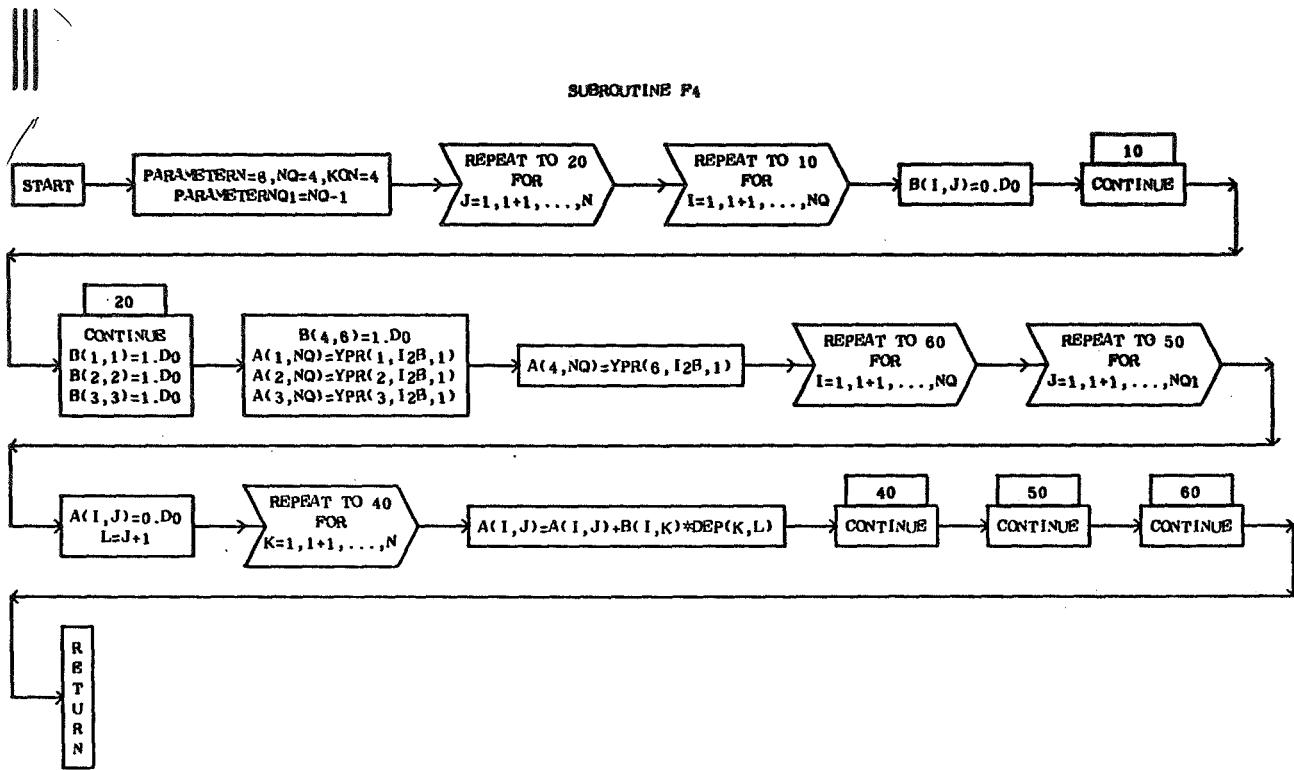
|||



|||

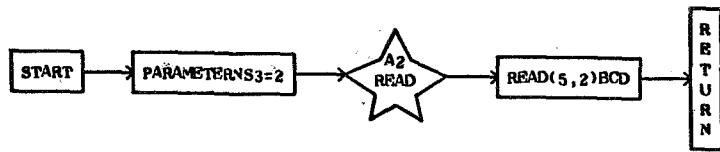
SUBROUTINE P3





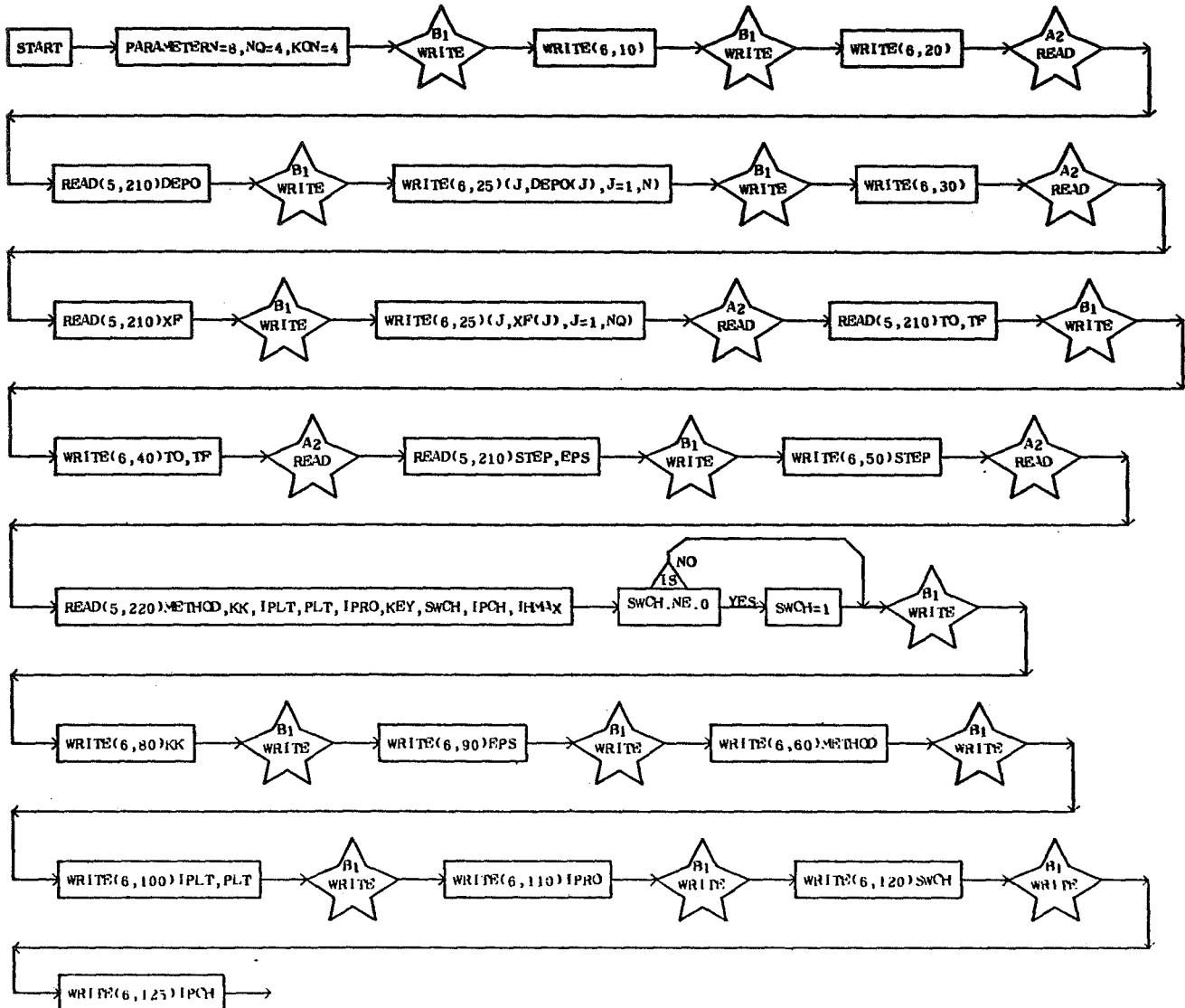
|||

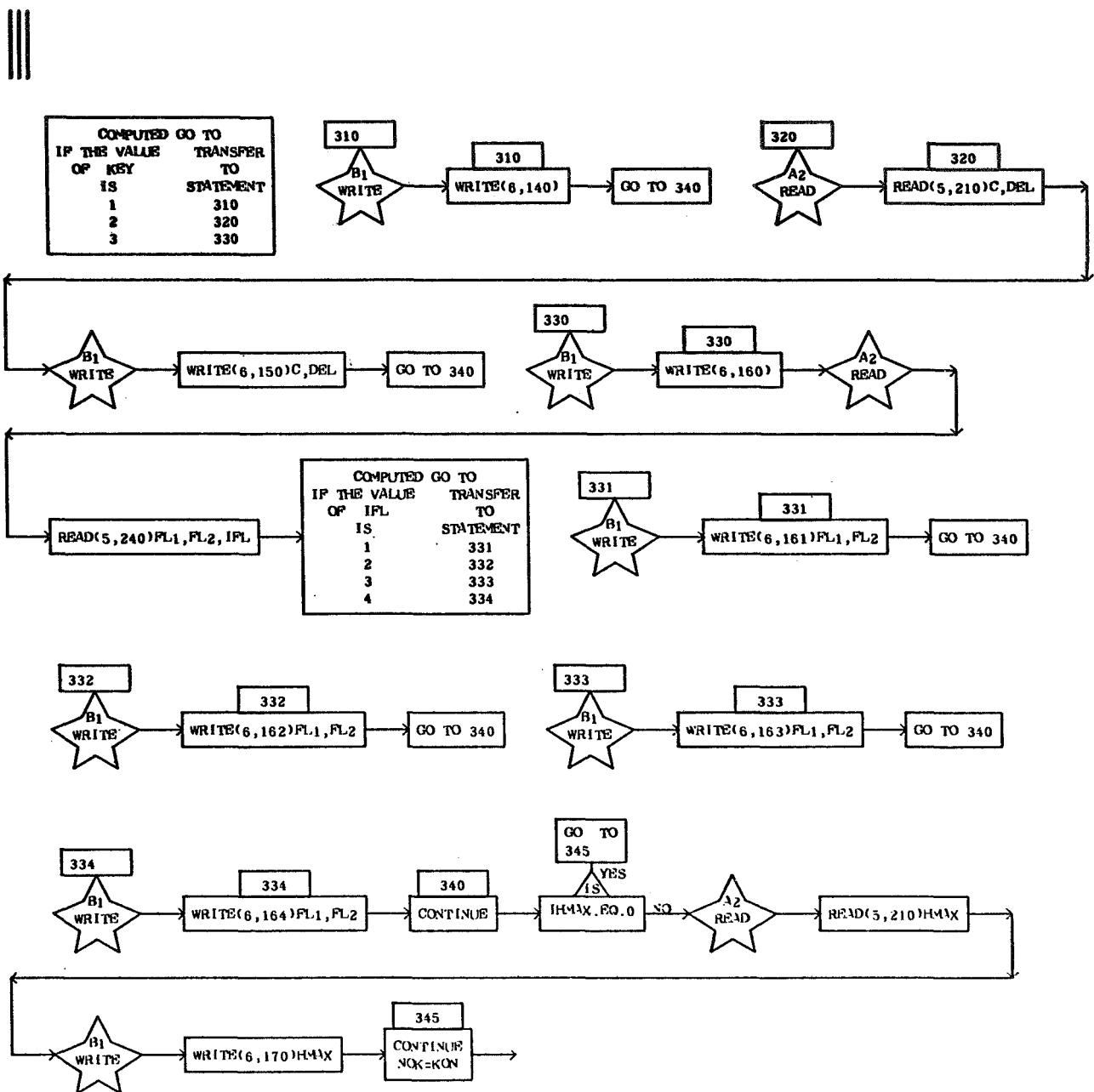
SUBROUTINE P5

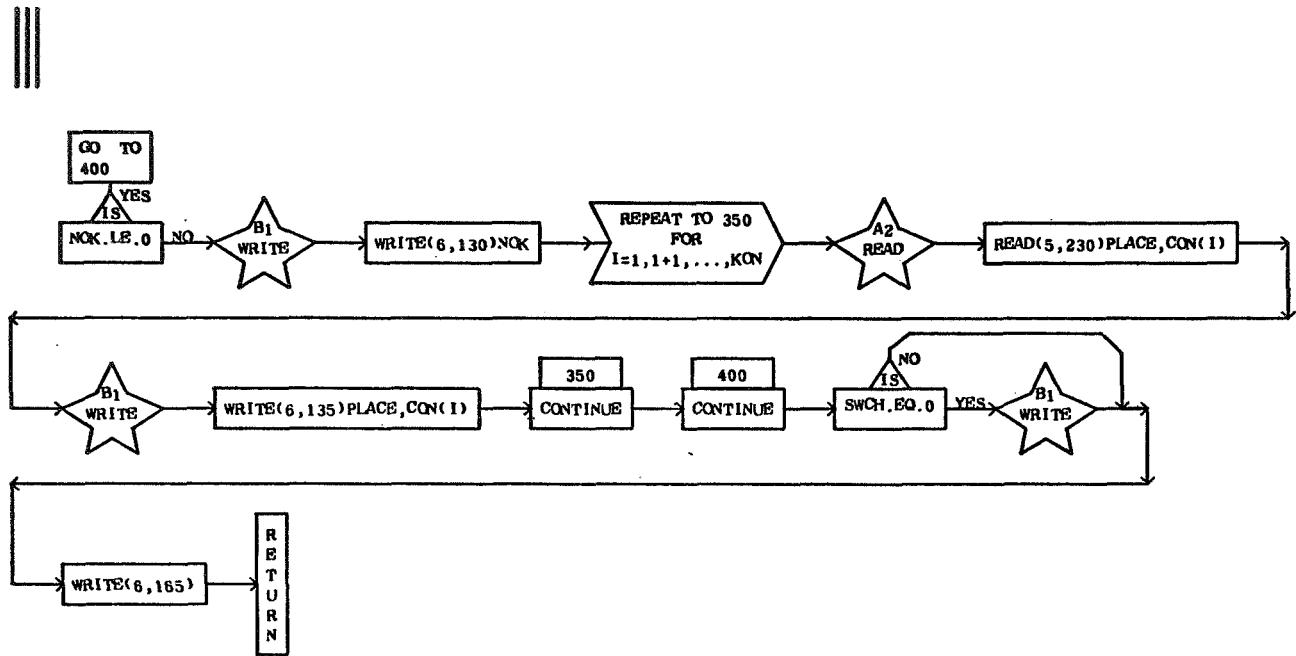


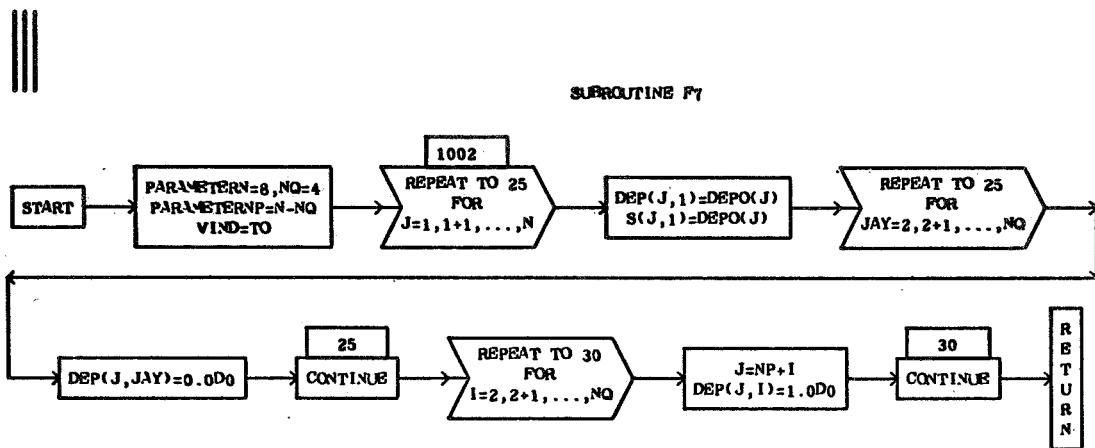
|||

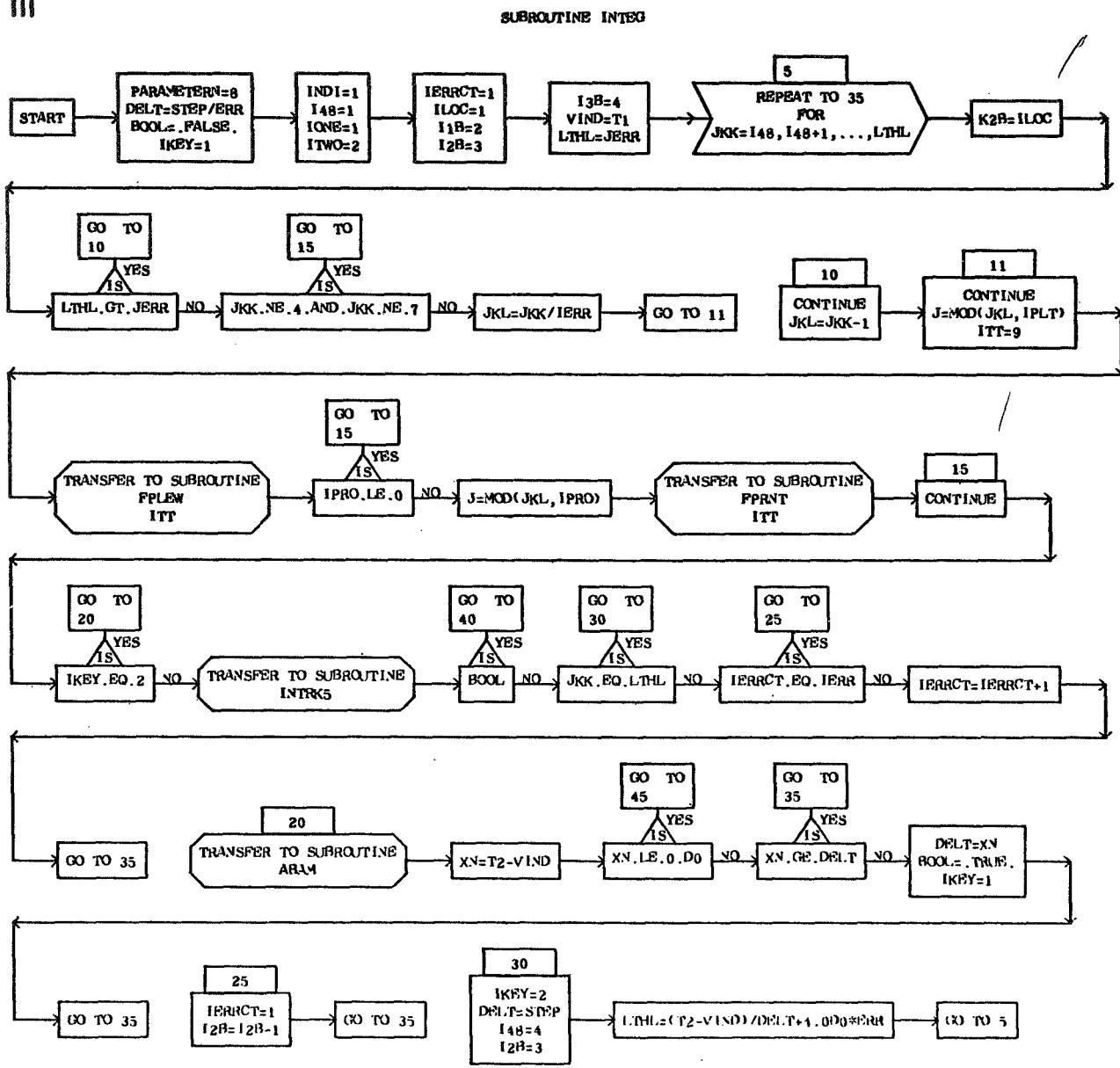
SUBROUTINE P6



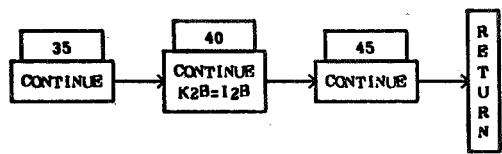


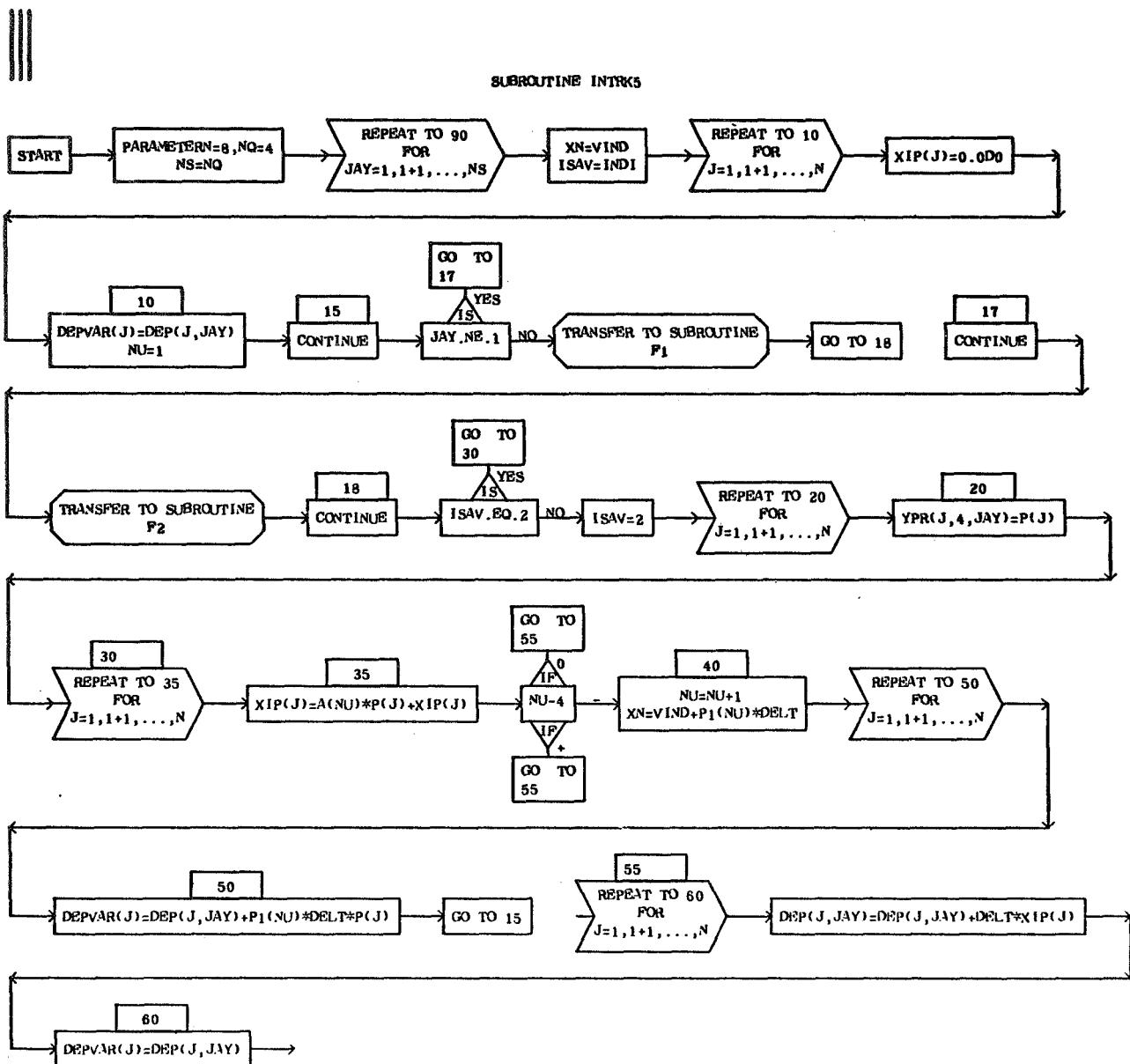


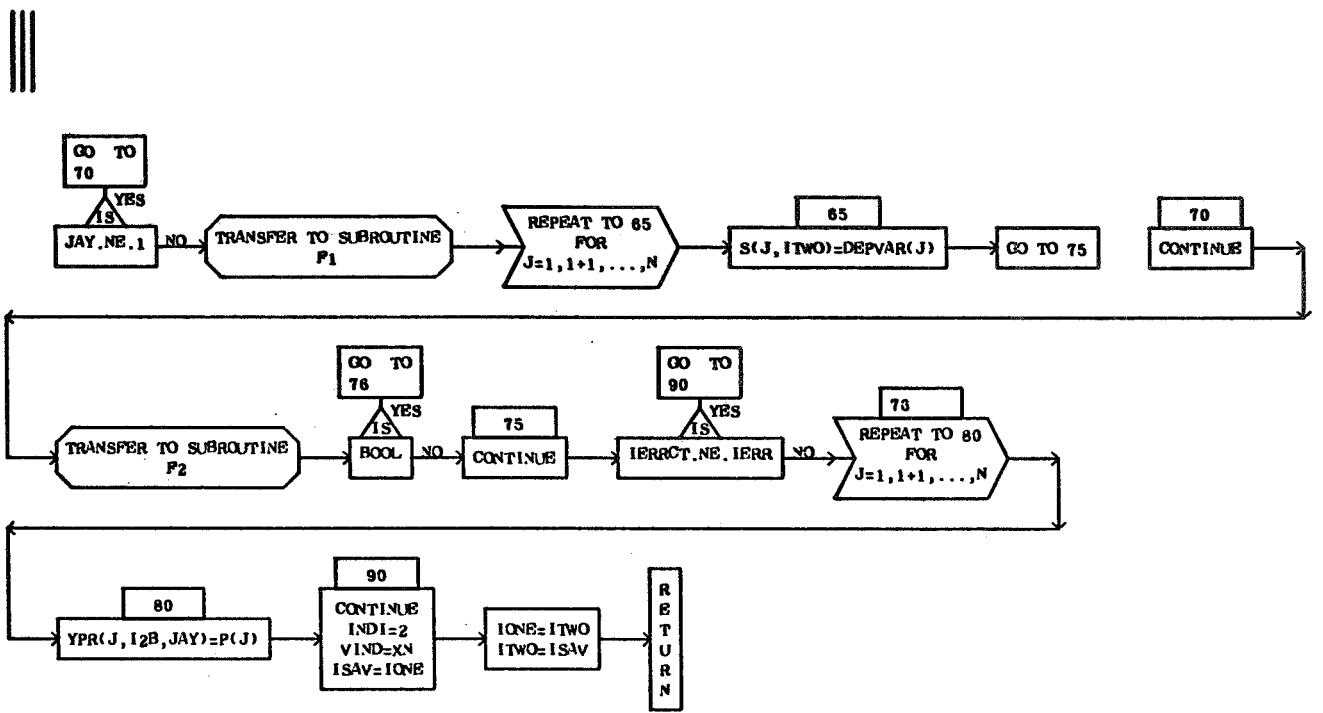


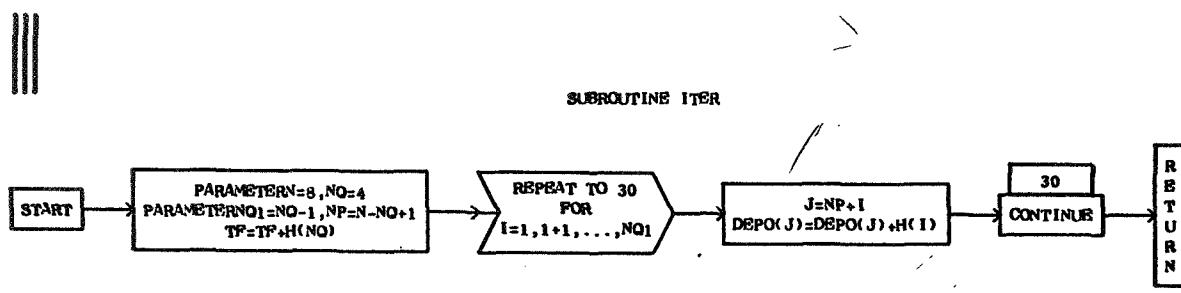


|||



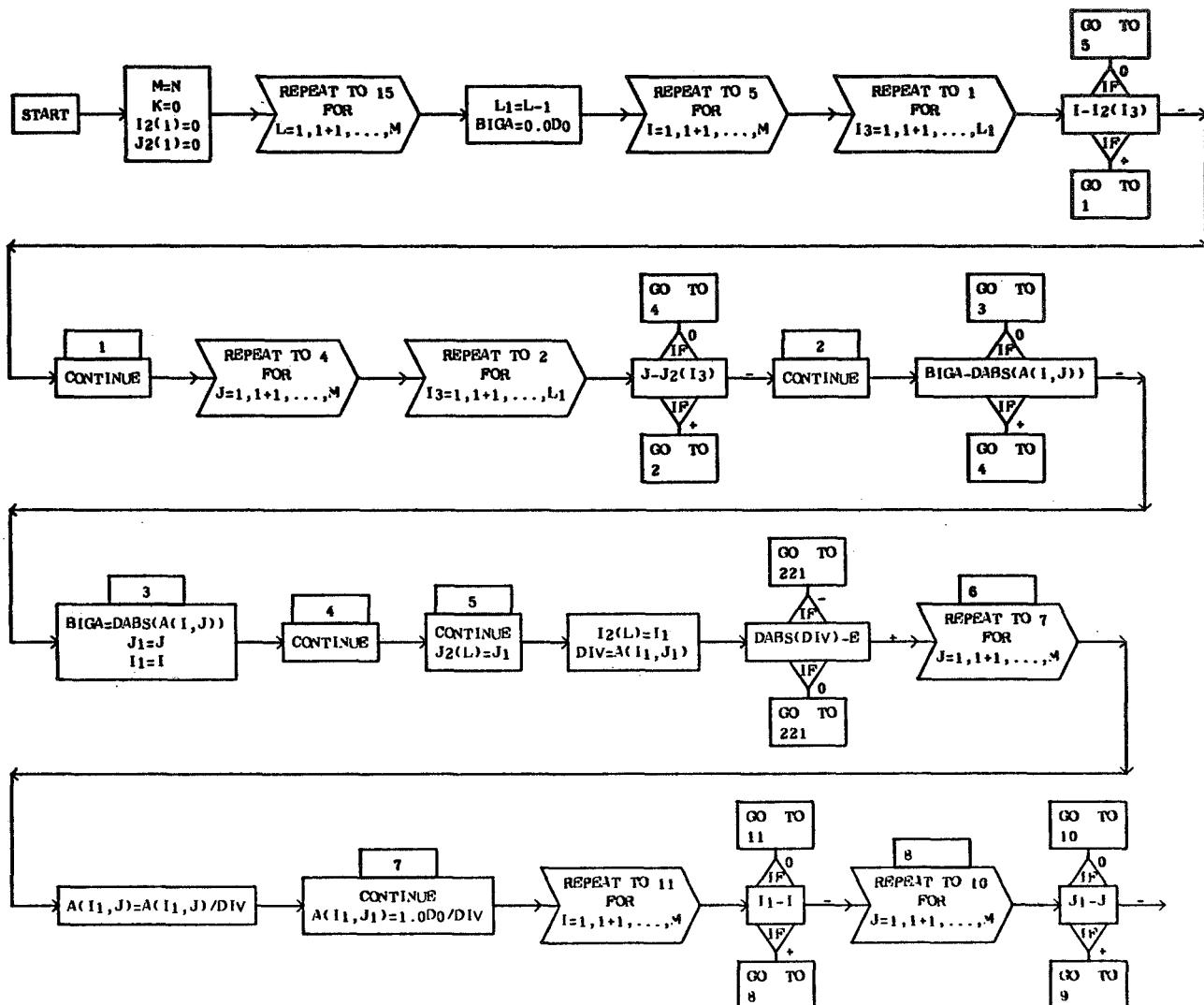




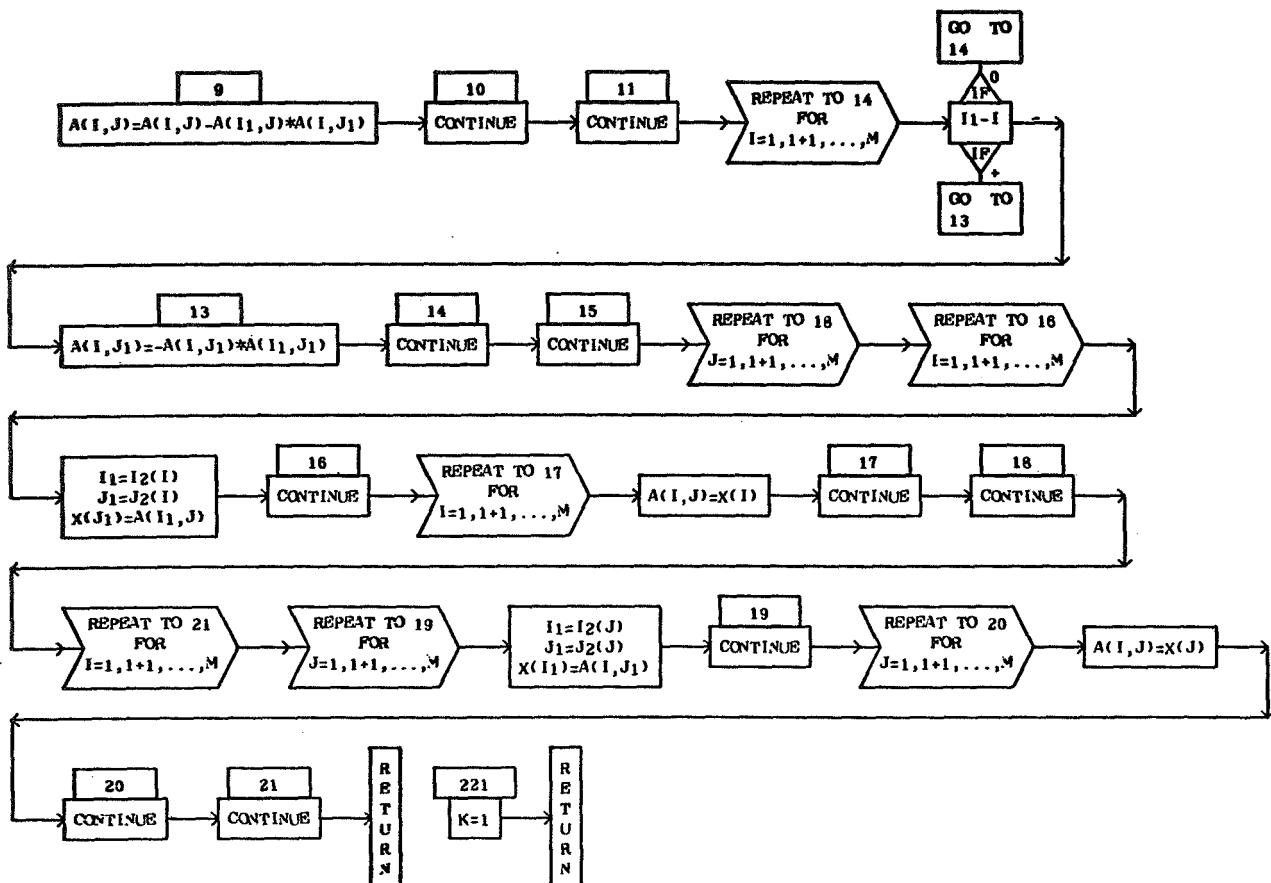


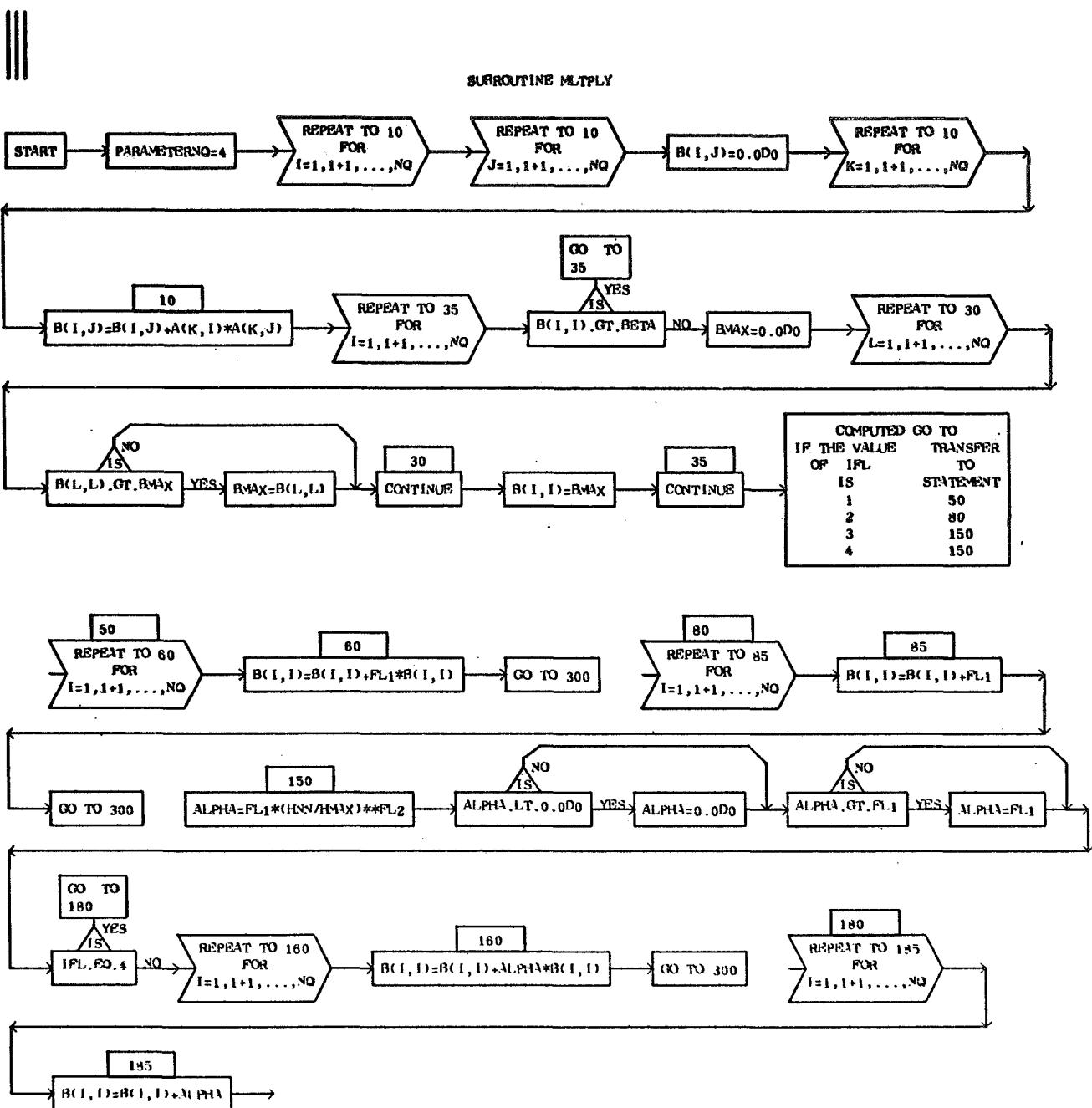
|||

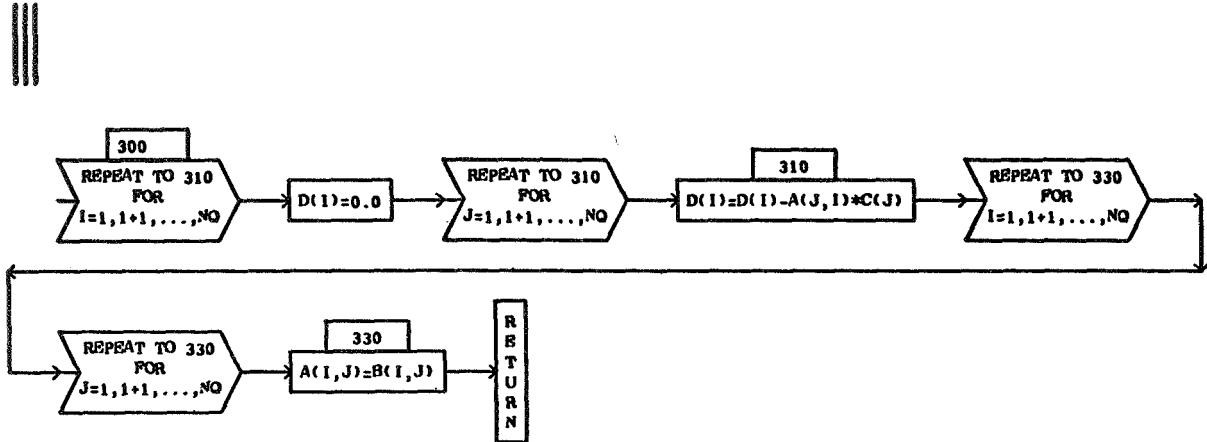
SUBROUTINE MINVDP(A,N,K,X,J2,I2)



|||







4. SAMPLE INPUT

TIME IN DAYS					
CONTROL ANGLE IN DEGREES					
0.0					1.0
1.0					0.0
-1.0					0.0
-0.49484461					-1.0785625
0.0					0.81012728
1.523679					0.0
0.0					3.3194865
0.3				-001	0.1
2	25	5	5		1 1
GRAVITATIONAL CONSTANT					1.0
INITIAL MASS					1.0
MASS FLOW RATE					0.74800391
THRUST					0.14012969
0.0					1.0
1.0					0.0
-1.0					0.0
-0.49484461					-1.0785625
0.0					0.81012728
1.523679					0.0
0.0					3.3194865
0.3				-001	0.1
2	25	5	5		2 1
0.5					0.1
GRAVITATIONAL CONSTANT					1.0
INITIAL MASS					1.0
MASS FLOW RATE					0.74800391
THRUST					0.14012969
0.0					1.0
1.0					0.0
-1.0					0.0
-0.49484461					-1.0785625
0.0					0.81012728
1.523679					0.0
0.0					3.3194865
0.3				-001	0.1
2	25	5	5		3 1
2.0					2.0
GRAVITATIONAL CONSTANT					1.0
INITIAL MASS					1.0
MASS FLOW RATE					0.74800391
THRUST					0.14012969
0.0					1.0
1.0					0.0
-1.0					0.0
-0.395874528					-0.86285152
0.0					0.81012728
1.523679					0.0
0.0					3.3194865
0.3				-001	0.1
2	25	5	5		3 1
2.0					2.0
GRAVITATIONAL CONSTANT					1.0
INITIAL MASS					1.0
MASS FLOW RATE					0.74800391
THRUST					0.14012969

0.0		0.81012728	
1.523679		0.0	
0.0		3.3194865	
0.3	-001	0.1	-005
2	25	5	5
			3
2.0		2.0	3
GRAVITATIONAL CONSTANT		1.0	
INITIAL MASS		1.0	
MASS FLOW RATE		0.74800391	-001
THRUST		0.14012060	
' EOF			

5. SAMPLE OUTPUT

METHOD OF PERTURBATION FUNCTIONS

TWO-POINT BOUNDARY VALUE PROBLEM

INITIAL VALUE OF THE DEPENDENT VARIABLES

1	.000000000000000	2	.100000000000000+001
3	.100000000000000+001	4	.000000000000000
5	-.100000000000000+001	6	.000000000000000
7	-.494844609999999+000	8	-.107856250000000+001

DESIRED VALUES OF THE TERMINAL VARIABLES

1	.000000000000000	2	.810127279999999+000
3	.152367900000000+001	4	.000000000000000

THE INITIAL TIME INTERVAL
FROM .000000000000000 / TO .331948649999999+001

THE INTEGRATION STEP SIZE IS .300000000000000-001

THE MAXIMUM NUMBER OF ITERATIONS ALLOWED IS 25

THE ACCURACY REQUIRED FOR TERMINATION IS .100000000000000-005

THE NUMBER OF ITERATIONS WITH THE ADAMS-MOULTON CORRECTOR IS 2

EVERY 5TH POINT WILL BE PLOTTED FOR EACH 5TH ITERATION

EVERY 0TH POINT WILL BE PRINTED ON EACH ITERATION

PRINT CONTROL SWITCH = 1

PUNCH CONTROL SWITCH = 0

NORMAL CORRECTION PROCEDURE

4 SPECIAL INPUT CONSTANTS AND THEIR IDENTIFICATIONS

GRAVITATIONAL CONSTANT	.100000000000000+001
INITIAL MASS	.100000000000000+001
MASS FLOW RATE	.748003910000000-001
THRUST	.140129690000000+000

BEGINNING THE 1 TH ITERATION
 TIME AT THE COMMENCEMENT OF FORWARD INTEGRATION IS .1170
 ELAPSED TIME WAS .1170

THE DEPENDENT VARIABLES AT TIME = .0000000000000000

1 1	.0000000000000000	1 2	.0000000000000000
2 1	.1000000000000000+001	2 2	.0000000000000000
3 1	.1000000000000000+001	3 2	.0000000000000000
4 1	.0000000000000000	4 2	.0000000000000000
5 1	-.1000000000000000+001	5 2	.0000000000000000
6 1	.0000000000000000	6 2	.1000000000000000+001
7 1	-.4948446099999999+000	7 2	.0000000000000000
8 1	-.1078562500000000+001	8 2	.0000000000000000

1 3	.0000000000000000	1 4	.0000000000000000
2 3	.0000000000000000	2 4	.0000000000000000
3 3	.0000000000000000	3 4	.0000000000000000
4 3	.0000000000000000	4 4	.0000000000000000
5 3	.0000000000000000	5 4	.0000000000000000
6 3	.0000000000000000	6 4	.0000000000000000
7 3	.1000000000000000+001	7 4	.0000000000000000
8 3	.0000000000000000	8 4	.1000000000000000+001

THE DEPENDENT VARIABLES AT TIME = .3319486499999999+001

1 1	.1340051529213354-006	1 2	-.1394670731000021+001
2 1	.8101273411074560+000	2 2	-.1863971775431036+001
3 1	.1523679074695318+001	3 2	.2795203649501503+000
4 1	.2490447285702952+001	4 2	-.1981448782912016+001
5 1	-.7118565185737861+000	5 2	.4169080799384614+001
6 1	.0000000000000000	6 2	.1000000000000000+001
7 1	.7584227019400643+000	7 2	-.3860097934044046+001
8 1	-.6742203751794457+000	8 2	.7973314319544971+001

1 3	.5272719904312362+001	1 4	.2390304718702410+001
2 3	.6285599236809059+000	2 4	.1344724990351260+001
3 3	.4832814651829563+001	3 4	.1272919107827973+001
4 3	.1493460241952366+001	4 4	.1455845270113698+001
5 3	.3005228841106784+001	5 4	.2039032539623578+000
6 3	.0000000000000000	6 4	.0000000000000000
7 3	-.3009761305147651+001	7 4	-.5954886874401262+000
8 3	-.1533775869407431+001	8 4	-.3954961594593700+001

TIME AT COMPLETION OF FORWARD INTEGRATION IS 2.5220
 ELAPSED TIME WAS 2.4050

THE TERMINAL CONSTRAINT VECTOR

1	.1340051529213354-006	2	.6110745605792528-007
3	.7469531817377861-007	4	.0000000000000000

THE NORM OF THE TERMINAL CONSTRAINTS .1651390103895012-006

THE CORRECTIONS AT THE 1 TH ITERATION

1	.4263213842357138-025	2	-.3051449300367758-008
3	-.4709508566341746-007	4	.3835757786522355-007

***** CONVERGENCE HAS BEEN ACHEIVED *****

THE FINAL TIME IS 2.6470

METHOD OF PERTURBATION FUNCTIONS
TWO-POINT BOUNDARY VALUE PROBLEM

INITIAL VALUE OF THE DEPENDENT VARIABLES

1	.000000000000000	2	.100000000000000+001
3	.100000000000000+001	4	.000000000000000
5	-.100000000000000+001	6	.000000000000000
7	-.494844609999999+000	8	-.1078562500000000+001

DESIRED VALUES OF THE TERMINAL VARIABLES

1	.000000000000000	2	.810127279999999+000
3	.152367900000000+001	4	.000000000000000

THE INITIAL TIME INTERVAL
FROM .000000000000000 / TO .331948649999999+001

THE INTEGRATION STEP SIZE IS .300000000000000-001

THE MAXIMUM NUMBER OF ITERATIONS ALLOWED IS 25

THE ACCURACY REQUIRED FOR TERMINATION IS .100000000000000-005

THE NUMBER OF ITERATIONS WITH THE ADAMS-MOULTON CORRECTOR IS 2

EVERY 5TH POINT WILL BE PLOTTED FOR EACH 5TH ITERATION

EVERY 0TH POINT WILL BE PRINTED ON EACH ITERATION

PRINT CONTROL SWITCH = 1

PUNCH CONTROL SWITCH = 0

FRACTIONAL CORRECTION PROCEDURE

C = .500000000000000+000 DEL = .100000000000000+000

4 SPECIAL INPUT CONSTANTS AND THEIR IDENTIFICATIONS

GRAVITATIONAL CONSTANT	.100000000000000+001
INITIAL MASS	.100000000000000+001
MASS FLOW RATE	.748003910000000-001
THRUST	.140129690000000+000

BEGINNING THE 1 TH ITERATION
 TIME AT THE COMMENCEMENT OF FORWARD INTEGRATION IS .1170
 ELAPSED TIME WAS .1170

THE DEPENDENT VARIABLES AT TIME = .000000000000000

1 1	.000000000000000	1 2	.000000000000000
2 1	.100000000000000+001	2 2	.000000000000000
3 1	.100000000000000+001	3 2	.000000000000000
4 1	.000000000000000	4 2	.000000000000000
5 1	-.100000000000000+001	5 2	.000000000000000
6 1	.000000000000000	6 2	.100000000000000+001
7 1	-.494844609999999+000	7 2	.000000000000000
8 1	-.107856250000000+001	8 2	.000000000000000

1 3	.000000000000000	1 4	.000000000000000
2 3	.000000000000000	2 4	.000000000000000
3 3	.000000000000000	3 4	.000000000000000
4 3	.000000000000000	4 4	.000000000000000
5 3	.000000000000000	5 4	.000000000000000
6 3	.000000000000000	6 4	.000000000000000
7 3	.100000000000000+001	7 4	.000000000000000
8 3	.000000000000000	8 4	.100000000000000+001

THE DEPENDENT VARIABLES AT TIME = .3319486499999999+001

1 1	.1340051529213354-006	1 2	-.1394670731000021+001
2 1	.8101273411074560+000	2 2	-.1863971775431036+001
3 1	.1523679074695318+001	3 2	.2795203649501503+000
4 1	.2490447285702952+001	4 2	-.1981448782912016+001
5 1	-.7118565185737861+000	5 2	.4169080799384614+001
6 1	.000000000000000	6 2	.100000000000000+001
7 1	.7584227019400643+000	7 2	-.3860097934044046+001
8 1	-.6742203751794457+000	8 2	.7973314319544971+001

1 3	.5272719904312362+001	1 4	.2390304718702410+001
2 3	.8285599236809059+000	2 4	.1344724990351260+001
3 3	.4832814651829563+001	3 4	.1272919107827973+001
4 3	.1493460241952366+001	4 4	.1455845270113698+001
5 3	.3005228841106784+001	5 4	.2039032539623578+000
6 3	.000000000000000	6 4	.000000000000000
7 3	-.3009761305147651+001	7 4	-.5954886874401262+000
8 3	-.1533775869407431+001	8 4	-.3954961594593700+001

TIME AT COMPLETION OF FORWARD INTEGRATION IS 2.5160
 ELAPSED TIME WAS 2.3990

THE TERMINAL CONSTRAINT VECTOR

1	.1340051529213354-006	2	.6110745605792528-007
3	.7469531817377861-007	4	.000000000000000

THE NORM OF THE TERMINAL CONSTRAINTS .1651390103895012-006

THE A MATRIX

1	1	-.1394670731000021+001	1	2	.5272719904312362+001
2	1	-.1863971775431036+001	2	2	.8285599236809059+000
3	1	.2795203649501503+000	3	2	.4832814651829563+001
4	1	.1000000000000000+001	4	2	.0000000000000000

1	3	.2390304718702410+001	1	4	-.1393234470670404+000
2	3	.1344724990351260+001	2	4	.1238553480234650+000
3	3	.1272919107827973+001	3	4	.1340051529213354-006
4	3	.0000000000000000	4	4	.0000000000000000

THE FRACTIONAL CORRECTION CONSTANT .5000000000000000+000

THIS IS THE 1 TH ATTEMPT TO CORRECT FROM THE 1 TH ITERATION

THE CORRECTIONS AT THE 1 TH ITERATION

1	.2131606921178569-025	2	-.1525724650183879-008
3	-.2354754283170873-007	4	.1917878893261177-007

***** CONVERGENCE HAS BEEN ACHEIVED *****

THE FINAL TIME IS 2.6840

METHOD OF PERTURBATION FUNCTIONS
TWO-POINT BOUNDARY VALUE PROBLEM

INITIAL VALUE OF THE DEPENDENT VARIABLES

1	.000000000000000	2	.100000000000000+001
3	.100000000000000+001	4	.000000000000000
5	-.100000000000000+001	6	.000000000000000
7	-.494844609999999+000	8	-.1078562500000000+001

DESIRED VALUES OF THE TERMINAL VARIABLES

1	.000000000000000	2	.810127279999999+000
3	.152367900000000+001	4	.000000000000000

THE INITIAL TIME INTERVAL
FROM .000000000000000 / TO .331948649999999+001

THE INTEGRATION STEP SIZE IS .300000000000000-001

THE MAXIMUM NUMBER OF ITERATIONS ALLOWED IS 25

THE ACCURACY REQUIRED FOR TERMINATION IS .100000000000000-005

THE NUMBER OF ITERATIONS WITH THE ADAMS-MOULTON CORRECTOR IS 2

EVERY 5TH POINT WILL BE PLOTTED FOR EACH 5TH ITERATION

EVERY 0TH POINT WILL BE PRINTED ON EACH ITERATION

PRINT CONTROL SWITCH = 1

PUNCH CONTROL SWITCH = 0

MINIMUM NORM CORRECTION PROCEDURE

VARIABLE ALPHA PROCEDURE
ALPHA = .200000000000000+001 P = .200000000000000+001

4 SPECIAL INPUT CONSTANTS AND THEIR IDENTIFICATIONS

GRAVITATIONAL CONSTANT	.100000000000000+001
INITIAL MASS	.100000000000000+001
MASS FLOW RATE	.748003910000000-001
THRUST	.1401296900000000+000

BEGINNING THE 1 TH ITERATION
TIME AT THE COMMENCEMENT OF FORWARD INTEGRATION IS .1230
ELAPSED TIME WAS .1230

THE DEPENDENT VARIABLES AT TIME = .0000000000000000

1 1	.0000000000000000	1 2	.0000000000000000
2 1	.1000000000000000+001	2 2	.0000000000000000
3 1	.1000000000000000+001	3 2	.0000000000000000
4 1	.0000000000000000	4 2	.0000000000000000
5 1	-.1000000000000000+001	5 2	.0000000000000000
6 1	.0000000000000000	6 2	.1000000000000000+001
7 1	-.4948446099999999+000	7 2	.0000000000000000
8 1	-.1078562500000000+001	8 2	.0000000000000000

1 3	.0000000000000000	1 4	.0000000000000000
2 3	.0000000000000000	2 4	.0000000000000000
3 3	.0000000000000000	3 4	.0000000000000000
4 3	.0000000000000000	4 4	.0000000000000000
5 3	.0000000000000000	5 4	.0000000000000000
6 3	.0000000000000000	6 4	.0000000000000000
7 3	.1000000000000000+001	7 4	.0000000000000000
8 3	.0000000000000000	8 4	.1000000000000000+001

THE DEPENDENT VARIABLES AT TIME = .3319486499999999+001

1 1	.1340051529213354-006	1 2	-.1394670731000021+001
2 1	.8101273411074560+000	2 2	-.1863971775431036+001
3 1	.1523679074695318+001	3 2	.2795203649501503+000
4 1	.2490447285702952+001	4 2	-.1981448782912016+001
5 1	-.7118565185737861+000	5 2	.4169080799384614+001
6 1	.0000000000000000	6 2	.1000000000000000+001
7 1	.7584227019400643+000	7 2	-.3860097934044046+001
8 1	-.6742203751794457+000	8 2	.7973314319544971+001

1 3	.5272719904312362+001	1 4	.2390304718702410+001
2 3	.8285599236809059+000	2 4	.1344724990351260+001
3 3	.4832814651829563+001	3 4	.1272919107827973+001
4 3	.1493460241952366+001	4 4	.1455845270113698+001
5 3	.3005228841106784+001	5 4	.2039032539623578+000
6 3	.0000000000000000	6 4	.0000000000000000
7 3	-.3009761305147651+001	7 4	-.5954886874401262+000
8 3	-.1533775869407431+001	8 4	-.3954961594593700+001

TIME AT COMPLETION OF FORWARD INTEGRATION IS 2.5240
ELAPSED TIME WAS 2.4010

THE TERMINAL CONSTRAINT VECTOR

1	.1340051529213354-006	2	.6110745605792528-007
5	.7469531817377861-007	4	.0000000000000000

THE NORM OF THE TERMINAL CONSTRAINTS .1651390103895012-006

THE A MATRIX

1 1	-.1394670731000021+001	1 2	.5272719904312362+001
2 1	-.1863971775431036+001	2 2	.8285599236809059+000
3 1	.2795203649501503+000	3 2	.4832814651829563+001
4 1	.1000000000000000+001	4 2	.0000000000000000
1 3	.2390304718702410+001	1 4	-.1393234470670404+000
2 3	.1344724990351260+001	2 4	.1238553480234650+000
3 3	.1272919107827973+001	3 4	.1340051529213354-006
4 3	.0000000000000000	4 4	.0000000000000000

THE A*A MATRIX

1 1	.1949288658580058+002	1 2	-.7547250320084354+001
2 1	-.7547250320084354+001	2 2	.1555325525862011+003
3 1	-.5484410643504918+001	3 2	.1986937461795605+002
4 1	-.3655250172832569-001	4 2	-.6319912871599335+000
1 3	-.5484410643504918+001	1 4	-.3655250172832569-001
2 3	.1986937461795605+002	2 4	-.6319912871599335+000
3 3	.2742649500899931+002	3 4	-.1664739406967066+000
4 3	-.1664739406967066+000	4 4	.1042535104100221+000

THE CORRECTIONS AT THE 1 TH ITERATION

1	.9079297568059467-008	2	-.4921214232811461-008
3	-.1239660938720785-007	4	.6004170665702625-007

***** CONVERGENCE HAS BEEN ACHEIVED *****

THE FINAL TIME IS 2.7360

METHOD OF PERTURBATION FUNCTIONS
TWO-POINT BOUNDARY VALUE PROBLEM

INITIAL VALUE OF THE DEPENDENT VARIABLES

1	.000000000000000	2	.100000000000000+001
3	.100000000000000+001	4	.000000000000000
5	-.100000000000000+001	6	.000000000000000
7	-.3958745279999999+000	8	-.8628515199999999+000

DESIRED VALUES OF THE TERMINAL VARIABLES

1	.000000000000000	2	.810127279999999+000
3	.1523679000000000+001	4	.000000000000000

THE INITIAL TIME INTERVAL
FROM .000000000000000 / TO .331948649999999+001

THE INTEGRATION STEP SIZE IS .300000000000000-001

THE MAXIMUM NUMBER OF ITERATIONS ALLOWED IS 25

THE ACCURACY REQUIRED FOR TERMINATION IS .100000000000000-005

THE NUMBER OF ITERATIONS WITH THE ADAMS-MOULTON CORRECTOR IS 2

EVERY 5TH POINT WILL BE PLOTTED FOR EACH 5TH ITERATION

EVERY 0TH POINT WILL BE PRINTED ON EACH ITERATION

PRINT CONTROL SWITCH = 1

PUNCH CONTROL SWITCH = 0

MINIMUM NORM CORRECTION PROCEDURE

VARIABLE ALPHA PROCEDURE
ALPHA = .200000000000000+001 P = .200000000000000+001

4 SPECIAL INPUT CONSTANTS AND THEIR IDENTIFICATIONS

GRAVITATIONAL CONSTANT	.100000000000000+001
INITIAL MASS	.100000000000000+001
MASS FLOW RATE	.748003910000000-001
THRUST	.140129690000000+000

BEGINNING THE 1 TH ITERATION
 TIME AT THE COMMENCEMENT OF FORWARD INTEGRATION IS .1240
 ELAPSED TIME WAS .1240

THE DEPENDENT VARIABLES AT TIME = .0000000000000000

1 1	.0000000000000000	1 2	.0000000000000000
2 1	.1000000000000000+001	2 2	.0000000000000000
3 1	.1000000000000000+001	3 2	.0000000000000000
4 1	.0000000000000000	4 2	.0000000000000000
5 1	-.1000000000000000+001	5 2	.0000000000000000
6 1	.0000000000000000	6 2	.1000000000000000+001
7 1	-.3958745279999999+000	7 2	.0000000000000000
8 1	-.8628515199999999+000	8 2	.0000000000000000

1 3	.0000000000000000	1 4	.0000000000000000
2 3	.0000000000000000	2 4	.0000000000000000
3 3	.0000000000000000	3 4	.0000000000000000
4 3	.0000000000000000	4 4	.0000000000000000
5 3	.0000000000000000	5 4	.0000000000000000
6 3	.0000000000000000	6 4	.0000000000000000
7 3	.1000000000000000+001	7 4	.0000000000000000
8 3	.0000000000000000	8 4	.1000000000000000+001

THE DEPENDENT VARIABLES AT TIME = .3319486499999999+001

1 1	.4393133799683166+000	1 2	.5699911572726967+000
2 1	.9030159840681569+000	2 2	-.2769863623328869+000
3 1	.1826985034993032+001	3 2	.1035177230850690+001
4 1	.2695180469031795+001	4 2	-.6300583362142748+000
5 1	-.5862242216022068+000	5 2	.3273930891986523+001
6 1	.0000000000000000	6 2	.1000000000000000+001
7 1	.3672794036554746+000	7 2	-.2814215594421436+001
8 1	-.1435088556299252+001	8 2	.7048535636381724+001

1 3	.9208285722259291+000	1 4	.1337858902226654+000
2 3	-.1292596787507494+000	2 4	.134125799600308+000
3 3	.8933933781390381+000	3 4	-.2445348688635943+000
4 3	.2245413927907733+000	4 4	.377537511605853+000
5 3	.7301358683896568+000	5 4	-.2421948794618864+000
6 3	.0000000000000000	6 4	.0000000000000000
7 3	-.2067285202548250+001	7 4	-.6984550113388558+000
8 3	-.2502866970053362+000	8 4	-.2868997681268256+001

TIME AT COMPLETION OF FORWARD INTEGRATION IS 2.5240
 ELAPSED TIME WAS 2.4000

THE TERMINAL CONSTRAINT VECTOR

1	.4393133799683166+000	2	.9288870406815695-001
3	.3033060349930329+000	4	.0000000000000000

THE NORM OF THE TERMINAL CONSTRAINTS .5418663193314779+000

THE A MATRIX

1 1	.5699911572726967+000	1 2	.9208285722259291+000
2 1	-.2769863623328369+000	2 2	-.1292596787507494+000
3 1	.1035177230850690+001	3 2	.8933933781390381+000
4 1	.1000000000000000+001	4 2	.0000000000000000

1 3	.1337858902226654+000	1 4	.1005183228839424+000
2 3	.1341257999600308+000	2 4	-.3654140383864541-001
3 3	-.2445348688635943+000	3 4	.4393133799683166+000
4 3	.0000000000000000	4 4	.0000000000000000

THE A*A MATRIX

1 1	.7419609790677531+001	1 2	.1485487794988623+001
2 1	.1485487794988623+001	2 2	.4988355156243217+001
3 1	-.2140311714277947+000	3 2	-.1126090201041422+000
4 1	.5221832338627916+000	4 2	.4897631384566236+000

1 3	-.2140311714277947+000	1 4	.5221832338627916+000
2 3	-.1126090201041422+000	2 4	.4897631384566236+000
3 3	.2870570901831736+000	3 4	-.9888065147137800-001
4 3	-.9888065147137300-001	4 4	.6133063597472582+000

THE CORRECTIONS AT THE 1 TH ITERATION

1	-.4221201814654489-001	2	-.1052090398174694+000
3	-.1259232058014515+000	4	-.1840721347124976+000

BEGINNING THE 2 TH ITERATION
 TIME AT THE COMMENCEMENT OF FORWARD INTEGRATION IS 2.7350
 ELAPSED TIME WAS .2110

THE DEPENDENT VARIABLES AT TIME = .0000000000000000

1 1	.0000000000000000	1 2	.0000000000000000
2 1	.1000000000000000+001	2 2	.0000000000000000
3 1	.1000000000000000+001	3 2	.0000000000000000
4 1	.0000000000000000	4 2	.0000000000000000
5 1	-.1000000000000000+001	5 2	.0000000000000000
6 1	-.4221201814654489-001	6 2	.1000000000000000+001
7 1	-.5010835678174694+000	7 2	.0000000000000000
8 1	-.9887747258014515+000	8 2	.0000000000000000

1 3	.0000000000000000	1 4	.0000000000000000
2 3	.0000000000000000	2 4	.0000000000000000
3 3	.0000000000000000	3 4	.0000000000000000
4 3	.0000000000000000	4 4	.0000000000000000
5 3	.0000000000000000	5 4	.0000000000000000
6 3	.0000000000000000	6 4	.0000000000000000
7 3	.1000000000000000+001	7 4	.0000000000000000
8 3	.0000000000000000	8 4	.1000000000000000+001

THE DEPENDENT VARIABLES AT TIME = .3135414365287502+001

1 1	.1994873188021790+000	1 2	.1238421621978655+000
2 1	.9020494976662636+000	2 2	-.5080784428607843+000
3 1	.1574107834433610+001	3 2	.8475395073996991+000
4 1	.2535181753024635+001	4 2	-.8309029900726967+000
5 1	-.8326715153627686+000	5 2	.4002854537931061+001
6 1	-.4221201814654489-001	6 2	.1000000000000000+001
7 1	.7990421600745388+000	7 2	-.3324284777156963+001
8 1	-.1125205657676754+001	8 2	.6745501161216616+001

1 3	.3032372083904383+001	1 4	.8555863350487180+000
2 3	-.1148433650501800-001	2 4	.3660847515889093+000
3 3	.2893751399111466+001	3 4	.2953189656545785+000
4 3	.6261444967761286+000	4 4	.6114530187334158+000
5 3	.2249893120068476+001	5 4	-.1808538684731704+000
6 3	.0000000000000000	6 4	.0000000000000000
7 3	-.2719017398540971+001	7 4	-.5020126380482614+000
8 3	-.9472558769528791+000	8 4	-.3216695622133141+001

TIME AT COMPLETION OF FORWARD INTEGRATION IS 5.0620
 ELAPSED TIME WAS 2.3270

THE TERMINAL CONSTRAINT VECTOR

1	.1994873188021790+000	2	.9192221766626364-001
3	.5042883443361056-001	4	-.4221201814654489-001

THE NORM OF THE TERMINAL CONSTRAINTS .2292810639409694+000

THE A MATRIX

1 1	.1238421621978655+000	1 2	.3032372083904383+001
2 1	-.5080784428607843+000	2 2	-.1148433650501800-001
3 1	.8475395073996991+000	3 2	.2893751399111466+001
4 1	.1000000000000000+001	4 2	.0000000000000000
1 3	.8555863350487180+000	1 4	.7349727471776571-002
2 3	.3660847515889093+000	2 4	.3494073635183525-001
3 3	.2953189656545785+000	3 4	.1994873188021790+000
4 3	.0000000000000000	4 4	.0000000000000000

THE A*A MATRIX

1 1	.2705032179369647+001	1 2	.2833939094608047+001
2 1	.2833939094608047+001	2 2	.2386042090763186+002
3 1	.1702523818136807+000	3 2	.3444831547342279+001
4 1	.1522309551337057+000	4 2	.5991525451245235+000
1 3	.1702523818136807+000	1 4	.1522309551337057+000
2 3	.3444831547342279+001	2 4	.5991525451245235+000
3 3	.1294603974565704+001	3 4	.7799198582874405-001
4 3	.7799198582874405-001	4 4	.5577649987030418-001

THE CORRECTIONS AT THE 2 TH ITERATION

1	.3340495839818038-001	2	-.1692635058238253-001
3	-.1291077692269256+000	4	.6950028160102317-002

BEGINNING THE 3 TH ITERATION
 TIME AT THE COMMENCEMENT OF FORWARD INTEGRATION IS 5.2720
 ELAPSED TIME WAS .2100

THE DEPENDENT VARIABLES AT TIME = .000000000000000

1	1	.000000000000000	1	2	.000000000000000
2	1	.100000000000000+001	2	2	.000000000000000
3	1	.100000000000000+001	3	2	.000000000000000
4	1	.000000000000000	4	2	.000000000000000
5	1	-.100000000000000+001	5	2	.000000000000000
6	1	-.8807059748364503-002	6	2	.100000000000000+001
7	1	-.5180099183998519+000	7	2	.000000000000000
8	1	-.1117882495028377+001	8	2	.000000000000000
1	3	.000000000000000	1	4	.000000000000000
2	3	.000000000000000	2	4	.000000000000000
3	3	.000000000000000	3	4	.000000000000000
4	3	.000000000000000	4	4	.000000000000000
5	3	.000000000000000	5	4	.000000000000000
6	3	.000000000000000	6	4	.000000000000000
7	3	.100000000000000+001	7	4	.000000000000000
8	3	.000000000000000	8	4	.100000000000000+001

THE DEPENDENT VARIABLES AT TIME = .3142364393447604+001

1	1	-.9231252372457258-001	1	2	-.1315202793544773-002
2	1	.7314536006146396+000	2	2	-.2156658372477752+001
3	1	.1438023206640970+001	3	2	.1248192371447132+001
4	1	.2342897200439062+001	4	2	-.1493078999043401+001
5	1	-.7278964344283806+000	5	2	.5136398369717176+001
6	1	-.8807059748364503-002	6	2	.100000000000000+001
7	1	.7665687371058279+000	7	2	-.4824380985158326+001
8	1	-.4156486732513994+000	8	2	.6792843686655758+001

1	3	.2506134205968545+001	1	4	.6042705845667370+000
2	3	.7048112314969139+000	2	4	.1377479392436861+001
3	3	.2229557687920515+001	3	4	-.1614446324458156+000
4	3	.7633571644328466+000	4	4	.9159246240451239+000
5	3	.1183326975107175+001	5	4	-.9616730132444799+000
6	3	.000000000000000	6	4	.000000000000000
7	3	-.1576450945521766+001	7	4	.5337697539689575+000
8	3	-.1588649034731688+001	8	4	-.3584841440136115+001

TIME AT COMPLETION OF FORWARD INTEGRATION IS 7.6000
 ELAPSED TIME WAS 2.3280

THE TERMINAL CONSTRAINT VECTOR

1	-.9231252372457258-001	2	-.7867367938536031-001
3	-.8565579335902925-001	4	-.8807059748364503-002

THE NORM OF THE TERMINAL CONSTRAINTS .1487468624939908+000

THE A MATRIX

1 1	-.1315202793544773-002	1 2	.2508134205968545+001
2 1	-.2156658372477752+001	2 2	.7048112314969139+000
3 1	.1248192371447132+001	3 2	.2229557687920515+001
4 1	.1000000000000000+001	4 2	.0000000000000000
1 3	.6042705845667370+000	1 4	-.2725628931284872+000
2 3	.1377479392436861+001	2 4	.1342731987234617+000
3 3	-.1614446324458156+000	3 4	-.9231252372457258-001
4 3	.0000000000000000	4 4	.0000000000000000

THE A*A MATRIX

1 1	.8295652050709999+001	1 2	.1259581149225245+001
2 1	.1259581149225245+001	2 2	.1353053245803726+002
3 1	-.3173061161605356+001	3 2	.2126504548314416+001
4 1	-.4044467306499074+000	4 2	-.7948031539454608+000
1 3	-.3173061161605356+001	1 4	-.4044467306499074+000
2 3	.2126504548314416+001	2 4	-.7948031539454608+000
3 3	.2633579644812991+001	3 4	.3516018689905522-001
4 3	.3516018689905522-001	4 4	.1160392091104279+000

THE CORRECTIONS AT THE 3 TH ITERATION

1	.3977544980233116-002	2	.2975532391312685-001
3	.3768163143842494-001	4	.1231594694984688-001

BEGINNING THE 4 TH ITERATION
 TIME AT THE COMMENCEMENT OF FORWARD INTEGRATION IS 7.8080
 ELAPSED TIME WAS .2080

THE DEPENDENT VARIABLES AT TIME = .000000000000000

1	1	.000000000000000	1	2	.000000000000000
2	1	.100000000000000+001	2	2	.000000000000000
3	1	.100000000000000+001	3	2	.000000000000000
4	1	.000000000000000	4	2	.000000000000000
5	1	-.100000000000000+001	5	2	.000000000000000
6	1	-.4829514768131386-002	6	2	.100000000000000+001
7	1	-.4882545944867250+000	7	2	.000000000000000
8	1	-.1080200863589952+001	8	2	.000000000000000
1	3	.000000000000000	1	4	.000000000000000
2	3	.000000000000000	2	4	.000000000000000
3	3	.000000000000000	3	4	.000000000000000
4	3	.000000000000000	4	4	.000000000000000
5	3	.000000000000000	5	4	.000000000000000
6	3	.000000000000000	6	4	.000000000000000
7	3	.100000000000000+001	7	4	.000000000000000
8	3	.000000000000000	8	4	.100000000000000+001

THE DEPENDENT VARIABLES AT TIME = .3154680340397451+001

1	1	.6284025235808489-001	1	2	-.1522378532924491+001
2	1	.8065984615459050+000	2	2	-.1859759822469675+001
3	1	.1547885248260215+001	3	2	.5999994284889531-001
4	1	.2422309454630739+001	4	2	-.193868204806252+001
5	1	-.6799926099886249+000	5	2	.3670345759321230+001
6	1	-.4829514768131386-002	6	2	.100000000000000+001
7	1	.6976232172234834+000	7	2	-.3598155401678205+001
8	1	-.5879696425191652+000	8	2	.7106255401576260+001

1	3	.6126851133969627+001	1	4	.2765968068571844+001
2	3	.1422977616154302+001	2	4	.1522782491203944+001
3	3	.5040184459718744+001	3	4	.1509379884454352+001
4	3	.2133002943092935+001	4	4	.1630356379138410+001
5	3	.2850514465236823+001	5	4	.3152149329089573+000
6	3	.000000000000000	6	4	.000000000000000
7	3	-.3047106528487672+001	7	4	-.5566577814574003+000
8	3	-.2091187503760970+001	8	4	-.3856906853806662+001

TIME AT COMPLETION OF FORWARD INTEGRATION IS 10.1470
 ELAPSED TIME WAS 2.3390

THE TERMINAL CONSTRAINT VECTOR

1	.6284025235808489-001	2	-.3528818454094940-002
3	.2420624826021514-001	4	-.4829514768131386-002

THE NORM OF THE TERMINAL CONSTRAINTS .6760633508659172-001

THE A MATRIX

1	1	-.1522378532924491+001	1	2	.6126851133969627+001
2	1	-.1859759822469675+001	2	2	.1422977616154302+001
3	1	.5999994284889531-001	3	2	.5040184459718744+001
4	1	.1000000000000000+001	4	2	.0000000000000000
1	3	.2765968068571844+001	1	4	-.1372973751262878+000
2	3	.1522782491203944+001	2	4	.8545288415985267-001
3	3	.1509379884454352+001	3	4	.6284025235808489-001
4	3	.0000000000000000	4	4	.0000000000000000

THE A*A MATRIX

1	1	.6991022708145193+001	1	2	-.1167137246004586+002
2	1	-.1167137246004586+002	2	2	.6698923323225061+002
3	1	-.6952297399045237+001	3	2	.2672111303414698+002
4	1	.5386714739458759-001	4	2	-.4028765737082296+000
1	3	-.6952297399045237+001	1	4	.5386714739458759-001
2	3	.2672111303414698+002	2	4	-.4028765737082296+000
3	3	.1262897988452051+002	3	4	-.1547841868332113+000
4	3	-.1547841868332113+000	4	4	.3103881589858251-001

THE CORRECTIONS AT THE 4 TH ITERATION

1	.4164955715104852-003	2	-.5544137570013910-002
3	-.2381941291416356-002	4	.1541133933505249+000

BEGINNING THE 5 TH ITERATION
 TIME AT THE COMMENCEMENT OF FORWARD INTEGRATION IS 10.3560
 ELAPSED TIME WAS .2090

THE DEPENDENT VARIABLES AT TIME = .0000000000000000

1 1	.0000000000000000	1 2	.0000000000000000
2 1	.1000000000000000+001	2 2	.0000000000000000
3 1	.1000000000000000+001	3 2	.0000000000000000
4 1	.0000000000000000	4 2	.0000000000000000
5 1	-.1000000000000000+001	5 2	.0000000000000000
6 1	-.4413019196620901-002	6 2	.1000000000000000+001
7 1	-.4937987320567389+000	7 2	.0000000000000000
8 1	-.1082582804881368+001	8 2	.0000000000000000

1 3	.0000000000000000	1 4	.0000000000000000
2 3	.0000000000000000	2 4	.0000000000000000
3 3	.0000000000000000	3 4	.0000000000000000
4 3	.0000000000000000	4 4	.0000000000000000
5 3	.0000000000000000	5 4	.0000000000000000
6 3	.0000000000000000	6 4	.0000000000000000
7 3	.1000000000000000+001	7 4	.0000000000000000
8 3	.0000000000000000	8 4	.1000000000000000+001

THE DEPENDENT VARIABLES AT TIME = .3308793733747976+001

1 1	.3543537076885874-002	1 2	-.1391361569232885+001
2 1	.8124370545538605+000	2 2	-.1810006433295714+001
3 1	.1522414831387861+001	3 2	.2458130897395349+000
4 1	.2489156105214893+001	4 2	-.1943279055614225+001
5 1	-.7255672257196072+000	5 2	.4136001262217634+001
6 1	-.4413019196620901-002	6 2	.1000000000000000+001
7 1	.7708253439118592+000	7 2	-.3822817950986701+001
8 1	-.6863372913144886+000	8 2	.7928616315944276+001

1 3	.5294724581476790+001	1 4	.2390195774391711+001
2 3	.8260307928105425+000	2 4	.1315786340494376+001
3 3	.4834442566675688+001	3 4	.1285830362747110+001
4 3	.1511415215861273+001	4 4	.1443059629132175+001
5 3	.3069594460230455+001	5 4	.2317318704508106+000
6 3	.0000000000000000	6 4	.0000000000000000
7 3	-.3053186685582263+001	7 4	-.6148470479626041+000
8 3	-.1554060932460036+001	8 4	-.3947065906590448+001

TIME AT COMPLETION OF FORWARD INTEGRATION IS 12.7950
 ELAPSED TIME WAS 2.4390

THE TERMINAL CONSTRAINT VECTOR

1	.3543537076885874-002	2	.2309774553860570-002
3	-.1264168612138140-002	4	-.4413019196620901-002

THE NORM OF THE TERMINAL CONSTRAINTS .6242161021199833-002

THE A MATRIX

1	1	-.1391361569232885+001	1	2	.5294724581476790+001
2	1	-.1810006433295714+001	2	2	.8260307928105425+000
3	1	.2458130897395349+000	3	2	.4834442566675688+001
4	1	.1000000000000000+001	4	2	.0000000000000000
1	3	.2390195774391711+001	1	4	-.1369738866755262+000
2	3	.1315786340494376+001	2	4	.1219427048335951+000
3	3	.1285830362747110+001	3	4	.3543537076885874-002
4	3	.0000000000000000	4	4	.0000000000000000

THE A*A MATRIX

1	1	.6274099140418095+001	1	2	-.7673628086944041+001
2	1	-.7673628086944041+001	2	2	.5210209488959550+002
3	1	-.5391134350221134+001	3	2	.1995858139441027+002
4	1	-.2926583053602419-001	4	2	-.6073795491690776+000
1	3	-.5391134350221134+001	1	4	-.2926583053602419-001
2	3	.1995858139441027+002	2	4	-.6073795491690776+000
3	3	.9100103863953840+001	3	4	-.1623874722258897+000
4	3	-.1623874722258897+000	4	4	.3365335508086855-001

THE CORRECTIONS AT THE 5 TH ITERATION

1	.4382799879546237-002	2	-.1006996231750214-002
3	.3909715075967291-002	4	.1068890832527561-001

BEGINNING THE 6 TH ITERATION
 TIME AT THE COMMENCEMENT OF FORWARD INTEGRATION IS 13.0050
 ELAPSED TIME WAS .2100

THE DEPENDENT VARIABLES AT TIME = .000000000000000

1	1	.000000000000000	1	2	.000000000000000
2	1	.100000000000000+001	2	2	.000000000000000
3	1	.100000000000000+001	3	2	.000000000000000
4	1	.000000000000000	4	2	.000000000000000
5	1	-.100000000000000+001	5	2	.000000000000000
6	1	-.3021931707466411-004	6	2	.100000000000000+001
7	1	-.4948057282884892+000	7	2	.000000000000000
8	1	-.1078673089805401+001	8	2	.000000000000000

1	3	.000000000000000	1	4	.000000000000000
2	3	.000000000000000	2	4	.000000000000000
3	3	.000000000000000	3	4	.000000000000000
4	3	.000000000000000	4	4	.000000000000000
5	3	.000000000000000	5	4	.000000000000000
6	3	.000000000000000	6	4	.000000000000000
7	3	.100000000000000+001	7	4	.000000000000000
8	3	.000000000000000	8	4	.100000000000000+001

THE DEPENDENT VARIABLES AT TIME = .3319482642073252+001

1	1	-.1643661890425153-004	1	2	-.1394502866172372+001
2	1	.8100667134705156+000	2	2	-.1864483536570655+001
3	1	.1523717810411174+001	3	2	.2797642048738974+000
4	1	.2490402217852878+001	4	2	-.1981509606264765+001
5	1	-.7118873438209103+000	5	2	.4169192442773725+001
6	1	-.3021931707466411-004	6	2	.100000000000000+001
7	1	.7584867497044485+000	7	2	-.3860524782276137+001
8	1	-.6740805432377951+000	8	2	.7973318897854506+001

1	3	.5270784715397122+001	1	4	.2389616139440934+001
2	3	.8289105287211714+000	2	4	.1345030337553331+001
3	3	.4830785318752025+001	3	4	.1272258642729933+001
4	3	.1493350177371745+001	4	4	.1455784106726813+001
5	3	.3003787193476435+001	5	4	.2034101626281530+000
6	3	.000000000000000	6	4	.000000000000000
7	3	-.3008948286699502+001	7	4	-.5950162772538181+000
8	3	-.1534050720025032+001	8	4	-.3955084250825209+001

TIME AT COMPLETION OF FORWARD INTEGRATION IS 15.4430
 ELAPSED TIME WAS 2.4380

THE TERMINAL CONSTRAINT VECTOR

1	1	-.1643661890425153-004	2	2	-.6056652948429989-004
3	1	.3881041117436215-004	4	4	-.3021931707466411-004

THE NORM OF THE TERMINAL CONSTRAINTS .7973657927675546-004

THE A MATRIX

1	1	-.1394502866172372+001	1	2	.5270784715397122+001
2	1	-.1864483536570655+001	2	2	.8289105287211714+000
3	1	.2797042048738974+000	3	2	.4830785318752025+001
4	1	.1000000000000000+001	4	2	.0000000000000000
1	3	.2389616139440934+001	1	4	-.1393948583534474+000
2	3	.1345030337553331+001	2	4	.1238439172982659+000
3	3	.1272258642729933+001	3	4	-.1643661890425153-004
4	3	.0000000000000000	4	4	.0000000000000000

THE A*A MATRIX

1	1	.6499205393697581+001	1	2	-.7544133613072361+001
2	1	-.7544133613072361+001	2	2	.5180475322008156+002
3	1	-.5484181048481116+001	3	2	.1985607040463404+002
4	1	-.3652301378107893-001	4	2	-.6321441636249838+000
1	3	-.5484181048481116+001	1	4	-.3652301378107893-001
2	3	.1985607040463404+002	2	4	-.6321441636249838+000
3	3	.9138014352558944+001	3	4	-.1665472890193257+000
4	3	-.1665472890193257+000	4	4	.3476824416303780-001

THE CORRECTIONS AT THE 6 TH ITERATION

1	•3021919026750394-004	2	•3889340044863507-004
3	•1105287276712826-003	4	•3910861120734608-005

BEGINNING THE 7 TH ITERATION
 TIME AT THE COMMENCEMENT OF FORWARD INTEGRATION IS 15.6530
 ELAPSED TIME WAS .2100

THE DEPENDENT VARIABLES AT TIME = .000000000000000

1 1	.000000000000000	1 2	.000000000000000
2 1	.100000000000000+001	2 2	.000000000000000
3 1	.100000000000000+001	3 2	.000000000000000
4 1	.000000000000000	4 2	.000000000000000
5 1	-.100000000000000+001	5 2	.000000000000000
6 1	-.1268071601735065-009	6 2	.100000000000000+001
7 1	-.4948446216889378+000	7 2	.000000000000000
8 1	-.1078562561077730+001	8 2	.000000000000000

1 3	.000000000000000	1 4	.000000000000000
2 3	.000000000000000	2 4	.000000000000000
3 3	.000000000000000	3 4	.000000000000000
4 3	.000000000000000	4 4	.000000000000000
5 3	.000000000000000	5 4	.000000000000000
6 3	.000000000000000	6 4	.000000000000000
7 3	.100000000000000+001	7 4	.000000000000000
8 3	.000000000000000	8 4	.100000000000000+001

THE DEPENDENT VARIABLES AT TIME = .3319486552934372+001

1 1	-.8080414465471319-007	1 2	-.1394668860772473+001
2 1	.8101272560754422+000	2 2	-.1863972791396285+001
3 1	.1523678940437577+001	3 2	.2795225540555539+000
4 1	.2490447207725215+001	4 2	-.1981448756683368+001
5 1	-.7118565780011420+000	5 2	.4169083082093652+001
6 1	-.1268071601735065-009	6 2	.100000000000000+001
7 1	.7584227926670698+000	7 2	-.3860099736361985+001
8 1	-.6742201594092333+000	8 2	.7973314682733444+001

1 3	.5272713802641023+001	1 4	.2390301516658150+001
2 3	.8285595237660352+000	2 4	.1344725334681354+001
3 3	.4832809048485204+001	3 4	.1272915975232330+001
4 3	.1493458489199005+001	4 4	.1455844626959524+001
5 3	.3005225572023532+001	5 4	.2039010833442484+000
6 3	.000000000000000	6 4	.000000000000000
7 3	-.3009758737085179+001	7 4	-.5954870203627850+000
8 3	-.1533775374965061+001	8 4	-.3954961637140963+001

TIME AT COMPLETION OF FORWARD INTEGRATION IS 18.0910
 ELAPSED TIME WAS 2.4380

THE TERMINAL CONSTRAINT VECTOR

1	-.8080414465471319-007	2	-.2392455772642215-007
3	-.5956242241783593-007	4	-.1268071601735065-009

THE NORM OF THE TERMINAL CONSTRAINTS .1031958938142468-006

THE A MATRIX

1 1	-.1394668860772473+001	1 2	.5272713802641023+001
2 1	-.1863972791396285+001	2 2	.8285595237660352+000
3 1	.2795225540555539+000	3 2	.4832809048485204+001
4 1	.1000000000000000+001	4 2	.0000000000000000

1 3	.2390301516658150+001	1 4	-.1393236032160158+000
2 3	.1344725334681354+001	2 4	.1238554324718680+000
3 3	.1272915975232330+001	3 4	-.8080414465471319-007
4 3	.0000000000000000	4 4	.0000000000000000

THE A*A MATRIX

1 1	.6497628656500260+001	1 2	-.7547223032165429+001
2 1	-.7547223032165429+001	2 2	.5184406502810879+002
3 1	-.5484401804392589+001	3 2	.1986932062532108+002
4 1	-.3655288780476607-001	4 2	-.6319922780770420+000

1 3	-.5484401804392589+001	1 4	-.3655288780476607-001
2 3	.1986932062532108+002	2 4	-.6319922780770420+000
3 3	.9142142646274473+001	3 4	-.1664739850475708+000
4 3	-.1664739850475708+000	4 4	.3475123456589633-001

THE CORRECTIONS AT THE 7 TH ITERATION

1	.1268071601740951-009	2	.8633038132783541-008
3	.1398768757996405-007	4	-.1454648348999221-007

***** CONVERGENCE HAS BEEN ACHEIVED *****

THE FINAL TIME IS 18.3010

METHOD OF PERTURBATION FUNCTIONS

TWO-POINT BOUNDARY VALUE PROBLEM

INITIAL VALUE OF THE DEPENDENT VARIABLES

1	.0000000000000000	2	.1000000000000000+001
3	.1000000000000000+001	4	.0000000000000000
5	-.1000000000000000+001	6	.0000000000000000
7	-.3958745279999999+000	8	-.8628515199999999+000

DESIRED VALUES OF THE TERMINAL VARIABLES

1	.0000000000000000	2	.8101272799999999+000
3	.1523679000000000+001	4	.0000000000000000

THE INITIAL TIME INTERVAL
FROM .0000000000000000 / TO .3319486499999999+001

THE INTEGRATION STEP SIZE IS .3000000000000000-001

THE MAXIMUM NUMBER OF ITERATIONS ALLOWED IS 25

THE ACCURACY REQUIRED FOR TERMINATION IS .1000000000000000-005

THE NUMBER OF ITERATIONS WITH THE ADAMS-MOULTON CORRECTOR IS 2

EVERY 5TH POINT WILL BE PLOTTED FOR EACH 5TH ITERATION

EVERY 0TH POINT WILL BE PRINTED ON EACH ITERATION

PRINT CONTROL SWITCH = 0

PUNCH CONTROL SWITCH = 0

MINIMUM NORM CORRECTION PROCEDURE

VARIABLE ALPHA PROCEDURE
ALPHA = .2000000000000000+001 P = .2000000000000000+001

4 SPECIAL INPUT CONSTANTS AND THEIR IDENTIFICATIONS

GRAVITATIONAL CONSTANT	.1000000000000000+001
INITIAL MASS	.1000000000000000+001
MASS FLOW RATE	.7480039100000000-001
THRUST	.1401296900000000+000

1 TH ITERATION
DISPLAYING FINAL TIME AND THE LAST 3 DEPENDENT VARIABLES

TF = .3319486499999999+001
1 .0000000000000000 2 -.3958745279999999+000
3 -.8628515199999999+000

THE NORM OF THE TERMINAL CONSTRAINTS .5418663193314779+000

2 TH ITERATION
DISPLAYING FINAL TIME AND THE LAST 3 DEPENDENT VARIABLES

TF = .3135414365287502+001
1 -.4221201814654489-001 2 -.5010835678174694+000
3 -.9887747258014515+000

THE NORM OF THE TERMINAL CONSTRAINTS .2292810639409694+000

3 TH ITERATION
DISPLAYING FINAL TIME AND THE LAST 3 DEPENDENT VARIABLES

TF = .3142364393447604+001
1 -.8807059748364503-002 2 -.5180099183998519+000
3 -.1117882495028377+001

THE NORM OF THE TERMINAL CONSTRAINTS .1487468624939908+000

4 TH ITERATION
DISPLAYING FINAL TIME AND THE LAST 3 DEPENDENT VARIABLES

TF = .3154680340397451+001
1 -.4629514768131386-002 2 -.4882545944867250+000
3 -.1080200863589952+001

THE NORM OF THE TERMINAL CONSTRAINTS .6760633508659172-001

5 TH ITERATION
DISPLAYING FINAL TIME AND THE LAST 3 DEPENDENT VARIABLES

TF = .3308793733747976+001
1 -.4413019196620901-002 2 -.4937987320567389+000
3 -.1082582804881368+001

THE NORM OF THE TERMINAL CONSTRAINTS .6242161021199833-002

6 TH ITERATION
DISPLAYING FINAL TIME AND THE LAST 3 DEPENDENT VARIABLES

TF = .3319482642073252+001
1 -.3021931707466411-004 2 -.4948057282884892+000
3 -.1078673089805401+001

THE NORM OF THE TERMINAL CONSTRAINTS .7973657927675546-004

7 TH ITERATION
DISPLAYING FINAL TIME AND THE LAST 3 DEPENDENT VARIABLES
A-128.

TF = .3319486552934372+001
1 -.1268071601735065-009 2 -.4948446216889378+000
3 -.1078562561077730+001

THE NORM OF THE TERMINAL CONSTRAINTS .1031958938142468-006

***** CONVERGENCE HAS BEEN ACHEIVED *****

Appendix B
BRACHISTOCHRONE
EXAMPLE

1. FORMULATION

The brachistochrone problem is stated as follows:
 Determine the control history $\theta(t)$ such that a particle falling from x_0, y_0 to x_f, y_f under the influence of a constant gravitational force g will do so in a minimum time.

The differential equations of motion are

$$\dot{x} = \sqrt{2g(y-a)} \cos \theta$$

$$\dot{y} = \sqrt{2g(y-a)} \sin \theta$$

where a is a constant $y_0 - \frac{v_0^2}{2g}$ and $v = (\dot{x}^2 + \dot{y}^2)^{1/2}$

When the optimization process is applied, the control angle θ is eliminated from the above equations and two additional equations are required to be satisfied (Euler-Lagrange equations). Hence, the nonlinear two-point boundary value problem may be summarized as:

The differential equations for F1 are

$$\dot{z}_1 = \dot{x} = - \frac{\lambda_1 \sqrt{2g(y-a)}}{\sqrt{\lambda_1^2 + \lambda_2^2}}$$

$$\dot{z}_2 = \dot{y} = - \frac{\lambda_2 \sqrt{2g(y-a)}}{\sqrt{\lambda_1^2 + \lambda_2^2}}$$

$$\dot{z}_3 = \dot{\lambda}_1 = 0$$

$$\dot{z}_4 = \dot{\lambda}_2 = \frac{g \sqrt{\lambda_1^2 + \lambda_2^2}}{\sqrt{2g(y-a)}}$$

where $t = t_o$ $t = t_f$ (unspecified)

$$x(t_o) = 0 \quad x(t_f) = 5.0$$

$$y(t_o) = 1.0 \quad y(t_f) = 8.0$$

$$1 - \sqrt{2g(y-a)} \left. \sqrt{\lambda_1^2 + \lambda_2^2} \right|_{t_f} = 0$$

The perturbation equations for F2 are

$$\begin{aligned} \delta \dot{z}_1 &= \delta \dot{x} = - \left[\frac{\lambda_1 g}{\sqrt{\lambda_1^2 + \lambda_2^2} \sqrt{2g(y-a)}} \right] \delta z_2 - \left[\frac{\lambda_2^2 \sqrt{2g(y-a)}}{(\lambda_1^2 + \lambda_2^2)^{3/2}} \right] \delta z_3 \\ &\quad + \left[\frac{\lambda_1 \lambda_2 \sqrt{2g(y-a)}}{(\lambda_1^2 + \lambda_2^2)^{3/2}} \right] \delta z_4 \end{aligned}$$

$$\delta \dot{z}_2 = \delta \dot{y} = - \left[\frac{\lambda_2 g}{\sqrt{\lambda_1^2 + \lambda_2^2} \sqrt{2g(y-a)}} \right] \delta z_2 + \left[\frac{\lambda_1 \lambda_2 \sqrt{2g(y-a)}}{(\lambda_1^2 + \lambda_2^2)^{3/2}} \right] \delta z_3$$

$$- \left[\frac{\lambda_1^2 \sqrt{2g(y-a)}}{(\lambda_1^2 + \lambda_2^2)^{3/2}} \right] \delta z_4$$

$$\delta \dot{z}_3 = \delta \dot{\lambda}_1 = 0$$

$$\delta \dot{z}_4 = \delta \dot{\lambda}_2 = - \left[\frac{g^2 \sqrt{\lambda_1^2 + \lambda_2^2}}{2g(y-a)^{3/2}} \right] \delta z_2 + \left[\frac{\lambda_1 g}{\sqrt{\lambda_1^2 + \lambda_2^2} \sqrt{2g(y-a)}} \right] \delta z_3$$

$$+ \left[\frac{\lambda_2 g}{\sqrt{\lambda_1^2 + \lambda_2^2} \sqrt{2g(y-a)}} \right] \delta z_4$$

The terminal constraints for F3 are

$$h_1 = x(t_f) - 5.0$$

$$h_2 = y(t_f) - 8.0$$

$$h_3 = 1 - \sqrt{\lambda_1^2 + \lambda_2^2} \sqrt{2g(y-a)} \Big|_{t_f}$$

The partial derivative of the terminal constraints with respect to the dependent variables and the time rates of change of the terminal constraints for F4 are

$$\left[\frac{\partial h}{\partial z} \right] = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ D_1 & D_2 & D_3 & D_4 \end{bmatrix}_{t_f}$$

where

$$D_1 = 0 = -\dot{z}_3(t_f)$$

$$D_2 = -g \frac{\sqrt{\lambda_1^2 + \lambda_2^2}}{\sqrt{2g(y-a)}} = -\dot{z}_4(t_f)$$

$$D_3 = -\frac{\lambda_1 \sqrt{2g(y-a)}}{\sqrt{\lambda_1^2 + \lambda_2^2}} = \dot{z}_1(t_f)$$

$$D_4 = -\frac{\lambda_2 \sqrt{2g(y-a)}}{\sqrt{\lambda_1^2 + \lambda_2^2}} = \dot{z}_2(t_f)$$

and

$$\dot{h} = \begin{bmatrix} \dot{x}(t_f) \\ \dot{y}(t_f) \\ 0 \end{bmatrix}$$

2. MODIFICATIONS FOR CUR SYSTEM

\$JOB LEC1SW S WILLIAMS L09856 ED02 E131 7006 C 3 2 4020 4 B
 'N HDG INPUT 10615.
 ' ASG P=10615
 ' XQT CUR
 TRW P
 IN P
 TRI P
 TOC
 'T FOR,* MAIN,MAIN,MAIN/B
 -1,1 PARAMETER N=4,NQ=3,KON=2,NS3=2
 'T FOR,* ABAM,ABAM,ABAM/B
 -2,2 PARAMETER N=4,NQ=3
 'T FOR,* CONVRG,CONVRG,CONVRG/B
 -2,2 PARAMETER N=4,NQ=3
 'T FOR,* CRDPCH,CRDPCH,CRDPCH/B
 -2,2 PARAMETER N=4
 'T FOR,* FPLEW,FPLEW,FPLEW/B
 -2,4 PARAMETER N=4,NQ=3,NS3=2, KON=2
 DOUBLE PRECISION DEG
 DATA DEG/57.29577951308232D0/
 -61,62 X(1)=DEP(1,1)
 Y(2)=-DEP(2,1)
 'T FOR,* FPLOT,FPLOT,FPLOT/B
 -2,2 PARAMETER NS3=2
 'T FOR,* FPRNT,FPRNT,FPRNT/B
 -2,2 PARAMETER N=4, NQ=3
 'T FOR,* F1,F1,F1/B
 -2,2 PARAMETER N=4,KON=2
 -5,6 DOUBLE PRECISION WOED,WOGD,GM,AI,DSQRT
 EQUIVALENCE (CON(1),GM),(CON(2),AI)
 -9,19 WOED=DSQRT(2.0*GM*(TY(2)-AI))
 WOGD=DSQRT(TY(3)**2+TY(4)**2)
 P(1)=-TY(3)*WOED/WOGD
 P(2)=-TY(4)*WOED/WOGD
 P(3)=0.0D0
 P(4)= GM*WOGD/WOED
 'T FOR,* F2,F2,F2/B
 -2,2 PARAMETER N=4,KON=2
 -5,7 DOUBLE PRECISION WOED,WOGD,GM,AI,DSQRT
 EQUIVALENCE (CON(1),GM),(CON(2),AI)
 DOUBLE PRECISION R,R2,R3,R4
 -27,59 WOED=DSQRT(2.0*GM*(S(2,J)-AI))
 WOGD=SQRT(S(3,J)**2+S(4,J)**2)
 R=WOED**3
 R2=WOGD**3
 R3=S(3,J)**2
 R4=S(4,J)**2
 A(1,2)= -S(3,J)*GM/(WOED*WOGD)

```

A(1,3)=-R4*WOED/R2
A(1,4)= S(4,J)*S(3,J)*WOED/R2
A(2,2)= -S(4,J)*GM/(WOED*WOGD)
A(2,3)= S(3,J)*S(4,J)*WOED/R2
A(2,4)= -R3*WOED/R2
A(4,2)= -GM*GM*WOGD/R
A(4,3)= S(3,J)*GM/(WOGD*WOED)
A(4,4)= S(4,J)*GM/(WOGD*WOED)
'T FOR,* F3,F3,F3/B
-2,2
      PARAMETER N=4,NQ=3, KON=2
-15,18
      H(1)=DEP(1,1)-XF(1)
      H(2)=DEP(2,1)-XF(2)
      H(3)=0.D0
      DO 20 I=1,NQ1
      H(3)=H(3)+YPR(I,I2B,1)*DEP(NQ1+I,1)
20 CONTINUE
      H(3)=1.D0 +H(3)
'T FOR,* F4,F4,F4/B
-2,2
      PARAMETER N=4,NQ=3, KON=2
-23,26
      B(1,1)=1.D0
      B(2,2)=1.D0
      B(3,2)=-YPR(4,I2B,1)
      B(3,3)= YPR(1,I2B,1)
      B(3,4)= YPR(2,I2B,1)
-30,33
      A(1,NQ)=YPR(1,I2B,1)
      A(2,NQ)=YPR(2,I2B,1)
      A(3,NQ)=0.D0
'T FOR,* F5,F5,F5/B
-2,2
      PARAMETER NS3=2
'T FOR,* F6,F6,F6/B
-2,2
      PARAMETER N=4,NQ=3,KON=2
'T FOR,* F7,F7,F7/B
-2,2
      PARAMETER N=4,NQ=3
'T FOR,* INTEG,INTEG,INTEG/B
-2,2
      PARAMETER N=4
'T FOR,* INTRK5,INTRK5,INTRK5/B
-2,2
      PARAMETER N=4,NQ=3
'T FOR,* ITER,ITER,ITER/B
-2,2
      PARAMETER N=4, NQ=3
'T FOR,* MLTPLY,MLTPLY,MLTPLY/B
-2,2
      PARAMETER NQ=3
' XQT CUR
      TOC
' XQT MAIN/B

```

X-DISPLACEMENT
Y-DISPLACEMENT

0.0	1.0
-.0357353733314653	-.1726379971977703
5.0	8.0
0.0	
0.0	.6076633560035577
0.006	0.0000000001
2 50 1 1 200	1 1 0 0
GRAVITATIONAL CONSTANT	32.1741
INITIAL CONSTANT	0.5
' EOF	

METHOD OF PERTURBATION FUNCTIONS
TWO-POINT BOUNDARY VALUE PROBLEM

INITIAL VALUE OF THE DEPENDENT VARIABLES

1	.000000000000000	2	.100000000000000+001
3	-.3573537333146529-001	4	-.1726379971977703+000

DESIRED VALUES OF THE TERMINAL VARIABLES

1	.500000000000000+001	2	.800000000000000+001
3	.000000000000000		

THE INITIAL TIME INTERVAL
FROM .000000000000000 / TO .6076633560035576+000

THE INTEGRATION STEP SIZE IS .599999999999999-002

THE MAXIMUM NUMBER OF ITERATIONS ALLOWED IS 50

THE ACCURACY REQUIRED FOR TERMINATION IS .100000000000000-009

THE NUMBER OF ITERATIONS WITH THE ADAMS-MOULTON CORRECTOR IS 2

EVERY 1TH POINT WILL BE PLOTTED FOR EACH 1TH ITERATION

EVERY 200TH POINT WILL BE PRINTED ON EACH ITERATION

PRINT CONTROL SWITCH = 1

PUNCH CONTROL SWITCH = 0

NORMAL CORRECTION PROCEDURE

2 SPECIAL INPUT CONSTANTS AND THEIR IDENTIFICATIONS

GRAVITATIONAL CONSTANT	.321741000000000+002
INITIAL CONSTANT	.500000000000000+000

BEGINNING THE 1 TH ITERATION
TIME AT THE COMMENCEMENT OF FORWARD INTEGRATION IS .0810
ELAPSED TIME WAS .0810

THE DEPENDENT VARIABLES AT TIME = .0000000000000000

1 1	.0000000000000000	1 2	.0000000000000000
2 1	.1000000000000000+001	2 2	.0000000000000000
3 1	-.3573537333146529-001	3 2	.1000000000000000+001
4 1	-.1726379971977703+000	4 2	.0000000000000000

1 3	.0000000000000000
2 3	.0000000000000000
3 3	.0000000000000000
4 3	.1000000000000000+001

THE DEPENDENT VARIABLES AT TIME = .6076633560035576+000

1 1	.4999999999999774+001	1 2	-.9064133566788512+002
2 1	.8000000000000303+001	2 2	.1148759769856110+003
3 1	-.3573537333146529-001	3 2	.1000000000000000+001
4 1	-.2819650908506611-001	4 2	-.6200888295657205+000

1 3	.1877417351888520+002
2 3	-.2379487288617123+002
3 3	.0000000000000000
4 3	.2918527621568634+000

TIME AT COMPLETION OF FORWARD INTEGRATION IS .9530
ELAPSED TIME WAS .8720

THE TERMINAL CONSTRAINT VECTOR

1	-.2251185349244622-012	2	.3036321194471724-012
3	-.6320638457069094-013		

THE NORM OF THE TERMINAL CONSTRAINTS .3832308257134548-012

THE CORRECTIONS AT THE 1 TH ITERATION

1	-.2170286656173745-015	2	.1141735878249916-013
3	-.5163256476650848-015		

***** CONVERGENCE HAS BEEN ACHEIVED *****

THE FINAL TIME IS 1.1900

NOTES

NOTES (Concluded)

f